

UNDER PRESSURE

Version 4.0

USER MANUAL

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GENERAL INFORMATION

UNDER PRESSURE DESIGN SOFTWARE

Under Pressure is a user-friendly software program running in a PC environment that uses theoretical elastic stress and strain formulas to calculate the stresses, strains and deflections of simple pressure vessel geometries. Under Pressure can provide quick and reliable results for pressure vessel geometries that closely approximate the formulas. Under Pressure can also help approximate more complex designs, prior to the much more time consuming application of the finite element method of stress analysis.

Under Pressure contains a database of commonly used pressure vessel materials that can be easily edited and supplemented for the user's specific needs. Under Pressure evaluates structural capabilities, deflections, and weights of common pressure vessel geometries such as cylindrical tubes, spheres, as well as hemispherical, conical, flat circular, and flat annular end closures.

This program was developed primarily for the oceanographic instrument designer, but it lends itself to all types of pressure vessel calculations. Under Pressure assumes that the designer understands the application of these formulas to his or her specific vessel design problem. The designer must manually iterate on wall thickness to achieve the safety factor and material sizing appropriate to the particular problem.

Under Pressure is NOT an automated pressure vessel design program. Under Pressure will NOT calculate exact stresses for real world geometry. The elastic formulas apply only to idealized pressure vessel configurations and the designer must interpret and apply these results as appropriate. See Appendix E for comparisons to Finite Element Method (FEA) analysis. Under Pressure cannot evaluate the impact of tolerances (out of round conditions), material variations (tempering variations, surface damage in brittle materials under tension), O-ring sealing grooves, stress concentrations, off center holes in round end caps etc. A sophisticated application of the finite element method is required to model such situations.

Most importantly, Under Pressure is no substitute for good engineering practices. No pressure vessel design should ever be considered complete until a qualified engineer has checked the calculations performed by the software and verified the appropriateness of their application to the specific problem and confirmed that the results are reasonable. This is particularly important when a design is to be fabricated and human safety and/or significant costs are involved. It is the engineer's responsibility to check the results of Under Pressure, not the other way around.

It is our hope that Under Pressure will aid pressure housing designers in efficiently creating more reliable, higher quality designs.

DEEPSEA POWER AND LIGHT

DeepSea Power and Light was founded in 1983 with the goal of providing high quality, innovative products to the oceanographic community. Initially manufacturing deep water power systems, the company's expertise has grown along with its product line to include underwater video and lighting systems, as well as video pipe inspection systems. All of DeepSea's standard products are rigorously

designed to perform in harsh marine environments, from wet/dry applications to full ocean depth deployments. Under Pressure software is one of the important tools used in designing our standard products.

DeepSea Power & Light is headquartered in over 62,000 square feet of high tech manufacturing space. Included in the plant are environmental and pressure testing facilities, complete machine shop with CNC lathes and mills, CAD and 3D drafting stations, electronics workshops, mold making and ultrasonic welding stations, and assembly and repair facilities. Our staff includes mechanical, electrical, and software engineers, machinists, and sales and service personnel.

Equipment manufactured by DeepSea has been used by various titanic expeditions (including lighting for the IMAX film *Titanica*), National Geographic Society, Woods Hole Oceanographic Institute, Monterey Bay Aquarium Research Institute, NASA, Lockheed, Oceaneering Technologies, and on dozens of deep diving submersibles including Alvin, Mirs I&II, Sea Cliff, Turtle, Nautille, Shinkai 6500, and Kaiko, an 11,000m ROV that has explored the Mariana trench.

CUSTOMER SUPPORT

Customer support is available during normal business hours at (858) 576-1261. For emergencies after hours or on the weekend, customer support can be reached by using the emergency paging service. Just follow the instructions in the after hours recording. Faxes can be sent to (858) 576-0219. Email can be sent to mail to: UPWIN@deepsea.com, or via our web site at <http://www.deepsea.com/>.

We encourage you to send us suggestions for future releases of the Under Pressure program to the above email address. Also please send us any material information you would like to see included in the standard material database in future release of the program.

INSTALLATION

COMPUTER REQUIREMENTS

- 486-66 or better processor
- Minimum 16MB RAM
- VGA Graphics
- 20MB hard disk space
- CD ROM drive
- Parallel port (for software key “Dongle,” or hardware lock)

SOFTWARE REQUIREMENTS

- Windows 95 or later, or Windows NT 4.0 or later.

INSTALLING UNDER PRESSURE

With power to your system turned off, plug the included software key into the printer port with the side that reads “COMPUTER” connected to the printer port. If you also have a printer, you can connect the printer cable to the other end of the key. This key must be installed in order for the software to run.

Turn on power to your computer, and then once Windows is loaded, insert the CD install disk. The installation program should automatically start. If not, double click on the “Setup” program.

Follow the on screen instructions that guide you through the installation. Once this installation is completed, a new group and items in the Start menu will be added.

After this installation, you will be required to install the drivers for the software key. Follow the on-screen directions to install. Carefully review the readme file during or after installation.

INSTALLING THE HARDWARE LOCK

Under Pressure is copy-protected using a hardware device called a hardware lock or dongle. The hardware lock only permits one copy of Under Pressure to run at a time. Attempting to run Under Pressure on more than one computer at a time, or attempting to defeat the intention of the hardware lock is a violation of the owner’s copyright, and is punishable by criminal and civil penalties.

Attach the hardware lock supplied with the program on the parallel printer port at the rear of the computer. If you do not have a printer, place the hardware lock in the port by itself. If you have a printer, disconnect the printer cable, then attach the hardware lock to the port, then attach the printer cable to the hardware lock. Your printer, and all other computer operations, will be unaffected by the presence of the hardware lock.

POLICY ON THE HARDWARE LOCK

Under Pressure is multiple-use protected by a device called a hardware lock. The hardware lock must be attached to the computer on which Under Pressure is running. The hardware lock prevents more than one copy of Under Pressure from running at one time. The registered user of the Under Pressure software is allowed to install the Under Pressure software on more than one computer, as long as there is no possibility that more than one copy of Under Pressure will be running at one time. For example, the user is allowed to install the Under Pressure software on a computer at work and another computer at home, as long as the two installations of Under Pressure are never used at the same time. The hardware lock must be physically moved from one computer to another, so that only one copy of Under Pressure can be run at one time.

The hardware lock supplied with the Under Pressure software is worth the full purchase price of the software. If the hardware lock is lost or stolen, it will not be replaced without payment of the full purchase price. Insure the hardware lock as you would any other business or personal asset of comparable value.

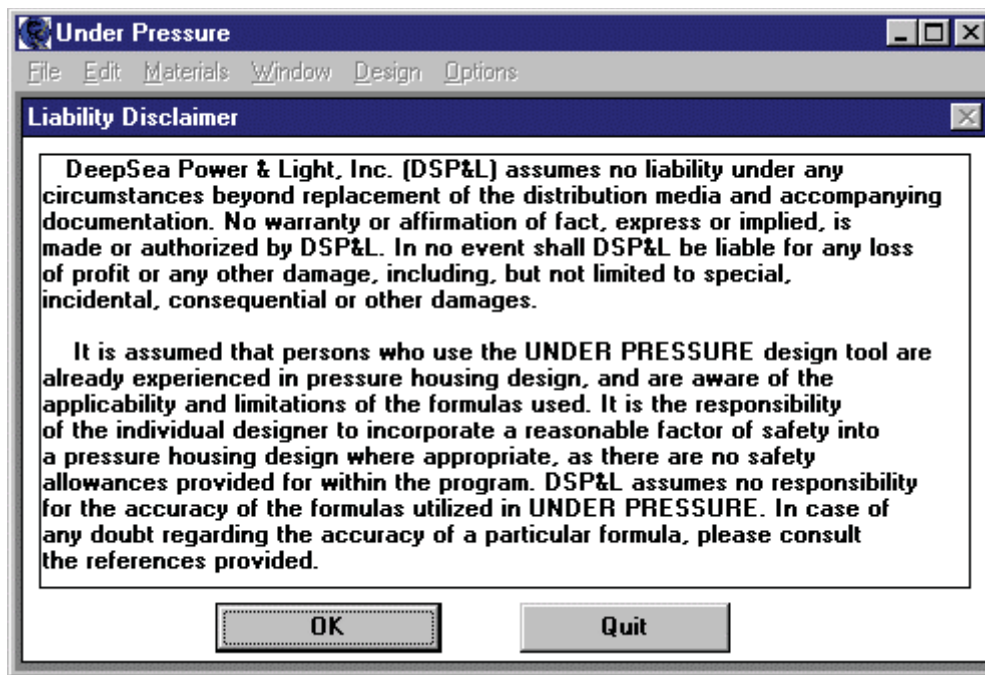
GETTING STARTED

NOTE: Most of the information in this manual is also available by clicking **H**elp on the menu bar of the Under Pressure Application Window or by pressing the **F1** key on the keyboard

STARTING THE UNDER PRESSURE APPLICATION WINDOW

-From Windows Start menu, choose the **Under Pressure** program by clicking on it.

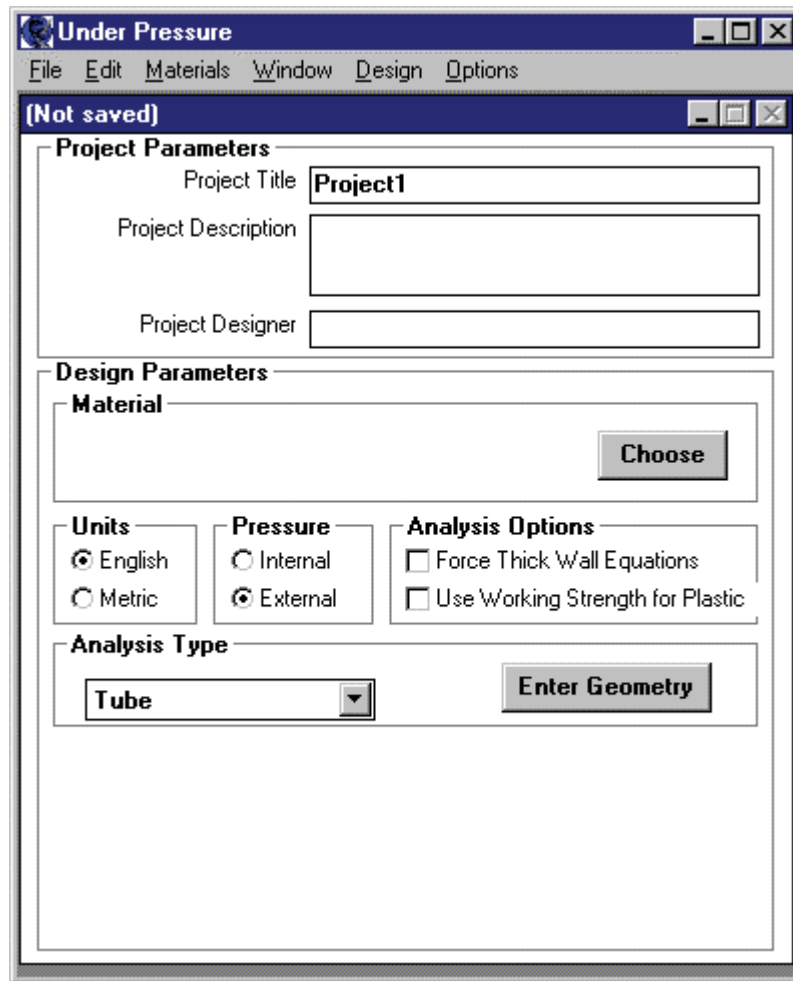
-Select **OK** if you accept the conditions of the Liability Disclaimer Box by single clicking on it to open the Under Pressure Application Window.



Under Pressure Liability Disclaimer Box

STARTING A NEW PROJECT

-Click on **F**ile on the menu bar of the Under Pressure Application Window then click on **N**ew Design or enter **Ctrl+N** from the keyboard.



Under Pressure Application Window

SETTING PROJECT PARAMETERS

-Use the cursor and keyboard to enter user defined **Project Parameters**:

- Project Title
- Project Description
- Project Designer

SETTING DESIGN PARAMETERS

Select pressure vessel **Material**

-Click on **CHOOSE** in the Under Pressure Application Window to open the Material Database Dialog Box (alternatively the Material Database Dialog Box can be accessed by clicking on **M**aterials on the menu bar in the Application Window and clicking on **V**iew **M**aterial or by entering Alt+M+V from the keyboard).

-Click on the appropriate **Main Category** of the desired material by clicking on the scroll arrow and clicking on choice.

-Click on the appropriate **Sub-Category** of the desired material by clicking on the scroll arrow and clicking on choice.

-Click on the **Name** of the desired material by clicking on the scroll arrow and clicking on choice.

-Click on **Done**.

Material Database Dialog Box

Select Analysis Units

-Click on either the **English** or **Metric** Option Button depending on the user's preference. (Note: you must switch to another analysis type or exit and reenter the Under Pressure program for this change to take effect.)

Select Pressure Orientation

-Click on either the **Internal** or **External** Option Button

-**Internal** option should be used when the magnitude of the applied pressure is greatest on the interior walls of the enclosed pressure vessel. **External** option should be used when the magnitude of the applied pressure is greatest on the exterior walls of the enclosed pressure vessel.

Examples of **Internal** Pressure Vessels:

- Boilers
- Reactors
- Hyperbaric Chambers
- Compressors
- Gas Storage, Scuba Tanks
- Steam Generators
- Pumps, Piping, Valves and other equipment used in energy systems, chemical processing plants etc.

Examples of **External** Pressure Vessels:

- Submerged Housings
- Vacuum Chambers
- Hyperbaric Chambers
- High Altitude Chambers

Select Analysis Options

-Click on the **Force Thick Wall Equations** check box if the user desires to force the use of thick wall equations for stress of analysis of tubes, spheres, and hemispheres in lieu of thin wall equations. Thick wall equations can be used for all ratios of mean shell wall radius to shell wall thickness. Thin wall equations are only recommended for ratios of mean shell wall radius to shell wall thickness > 10 . This check box has no relevance to the analysis of conical, flat circular, or flat annular endcaps.

-Click on the **Working Strength for Plastic** check box if the user desires to evaluate calculated stresses for plastic pressure vessel geometry's using the Working Strength of the selected plastic in lieu of the Ultimate Strength of the selected plastic. This check box has no relevance to pressure vessel materials other than plastics (such as metals, ceramics, and glass).

Select Analysis Type (Pressure Vessel Geometry)

-Click on the scroll arrow and select the user's choice of:

- **Tube**
- **Sphere**
- **Endcap Only**

by clicking on choice.

-If **Endcap Only Analysis Type** is selected, Click on the user's choice of **Endcap Configuration** by clicking on the scroll arrow and selecting:

- **Flat Annular**
- **Conical**
- **Hemispherical**
- **Flat Circular**

by clicking on choice.

-If **Flat Annular Endcap Configuration** is selected, click on the user's choice of **Edge Restraint Options-Outer/Inner** by clicking on the scroll arrow and selecting the Appropriate Boundary Condition:

- **Fixed/Free**
- **Fixed/Guided**
- **Fixed/Simply Supported**
- **Fixed/Fixed**
- **Simply Supported/Free**
- **Simply Supported/Guided**
- **Simply Supported/Simply Supported**
- **Simply Supported/Fixed**
- **Guided/Simply Supported**
- **Guided/Fixed**
- **Free/Simply Supported**
- **Free/Fixed**

by clicking on choice. Outer refers to the boundary condition that exists at the outside diameter (outer edge) of the plate. Inner refers to the boundary condition that exists at the edge of the center through hole in the plate.

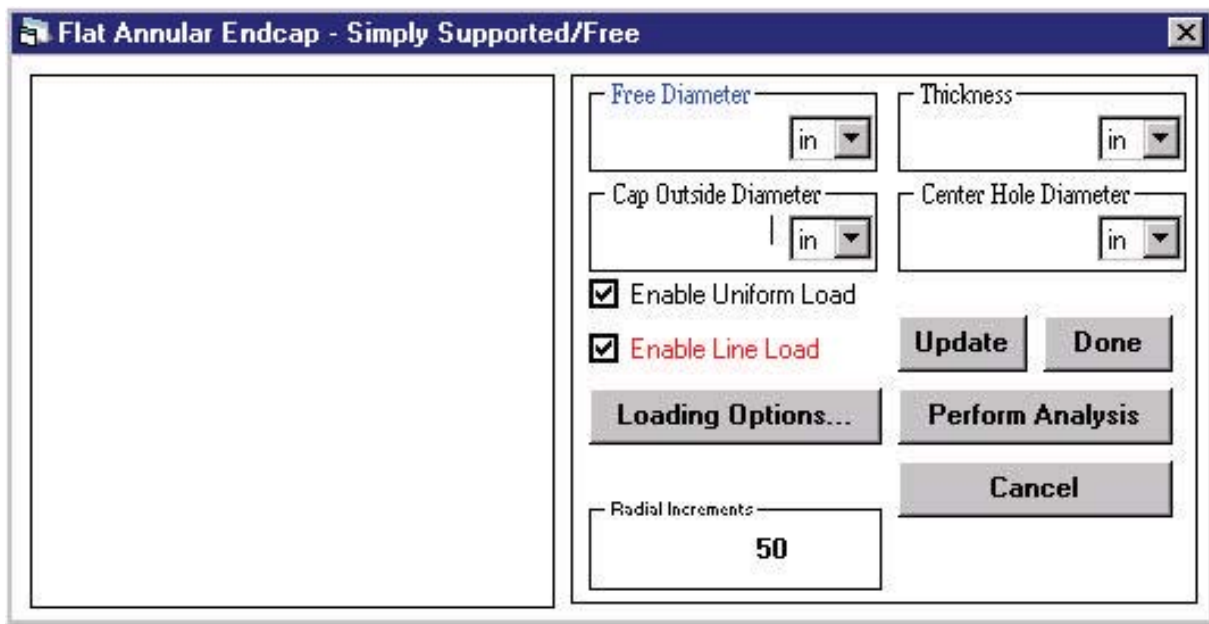
-If **Flat Circular Endcap Configuration** is selected, click on the user's choice of **Edge Restraint Options** by clicking on the scroll arrow and selecting the Appropriate Boundary Condition:

- **Simply Supported**
- **Fixed**

by clicking on choice.

ENTERING PRESSURE VESSEL GEOMETRY AND ANALYZE RESULTS

-Click on **Enter Geometry** to open the Geometry Dialog Box for the selected shape (Tube, Sphere, Flat Annular Endcap, Conical Endcap, Hemispherical Endcap, or Flat Circular Endcap). The title bar of this dialog box is based on the selected geometry and boundary conditions, for example: Flat Annular Endcap - Simply Supported/Free.

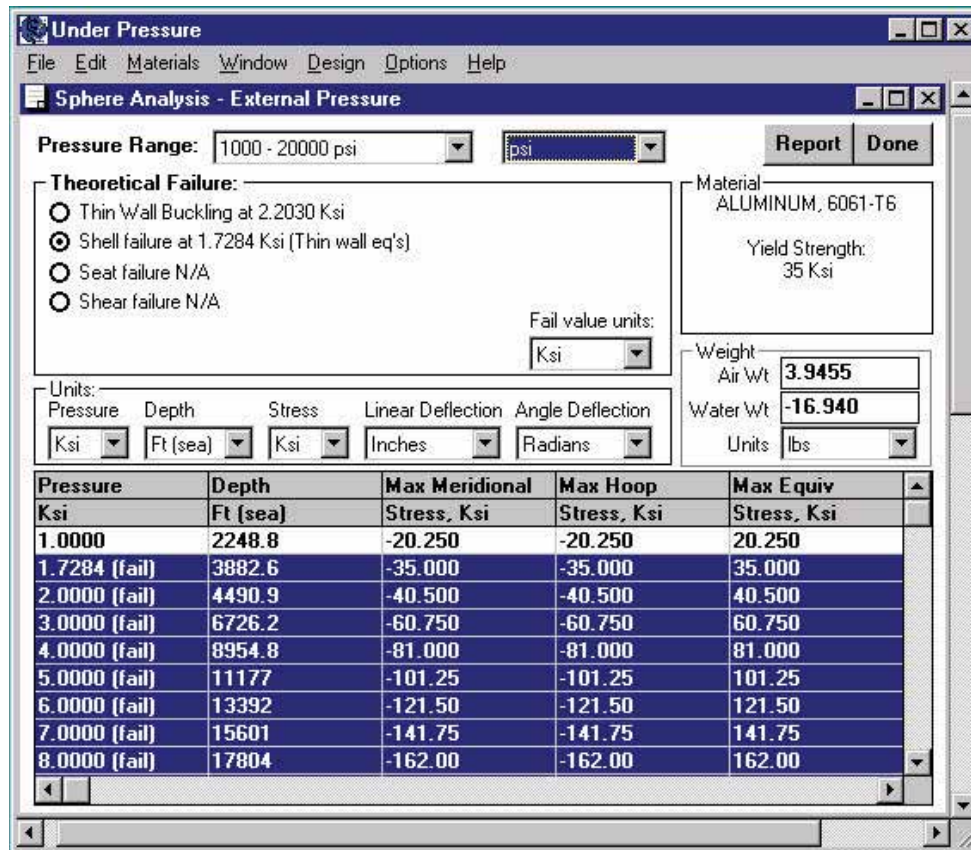


Geometry Dialog Box (shown for Flat Annular Endcap)

-Use the cursor and keyboard to enter dimensions and units of pressure vessel shape in the Geometry Dialog Box.

-Click on **Perform Analysis** to generate analysis results.

-Analysis results are displayed in an Analysis Dialog Box. The title bar of this dialog box is based on selected geometry and loading, for example: Sphere Analysis - External Pressure.



Analysis Dialog Box (shown for Sphere Analysis)

REVIEWING ANALYSIS RESULTS

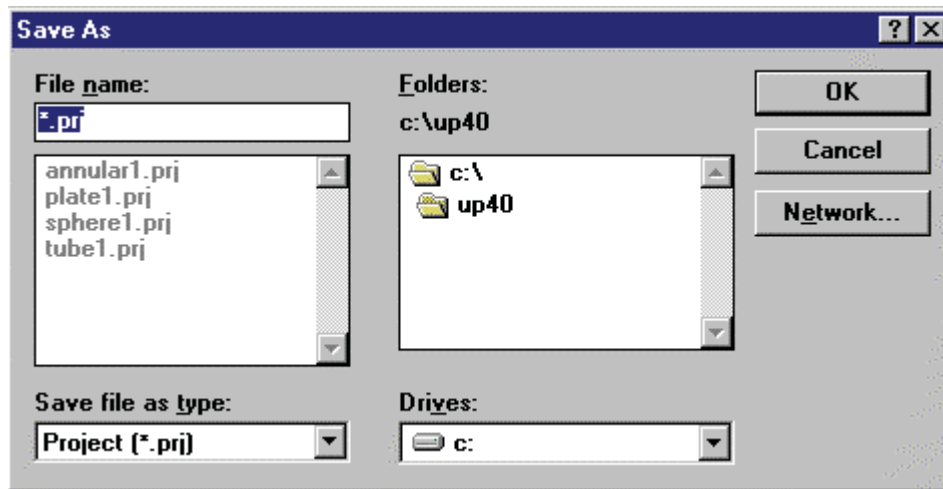
-The Analysis Dialog Box displays calculated results for the user defined design (material, geometry, boundary conditions, type of pressure loading - external or internal, etc.). Various scroll arrows allow the user to change the units used to display the analysis results. The structure of the Analysis Dialog Box is dependent upon the pressure vessel geometry that has been analyzed. Further detail on the specific information contained in the Analysis Dialog box is addressed in the section DETAILS ON ANALYSIS TYPE.

PRINTING THE RESULTS OF A COMPLETED PROJECT ANALYSIS

-Click on **File** on the menu bar of the Application Window, then click on **Print** or enter Alt+F+P from the keyboard. See the section REPORTS for more details.

SAVING A PROJECT

-Click on **File** on the menu bar of the Application Window, then click on **Save As** or enter Alt+F+A from the keyboard to open the Save As Dialog Box.

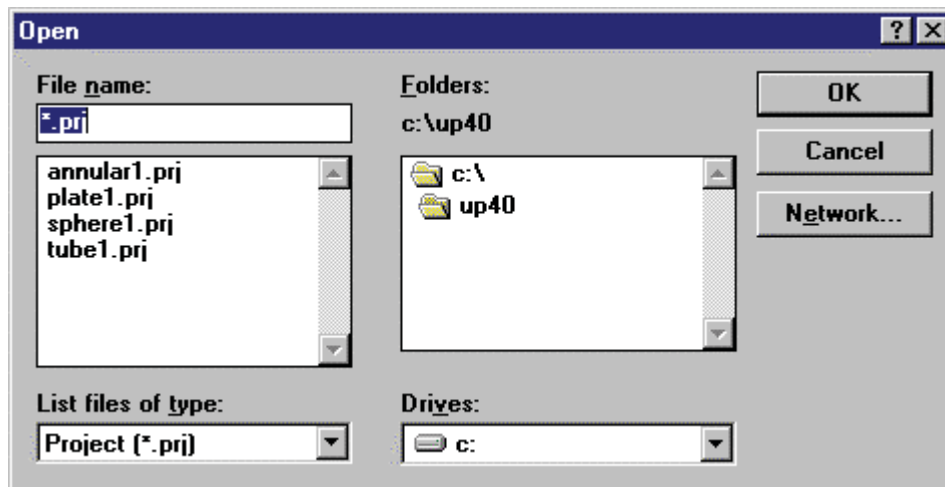


Save As Dialog Box

-In the Save As Dialog Box use the keyboard and cursor to enter a file name (*.prj) and directory and click on **OK**.

OPENING AN EXISTING PROJECT

-Click on **File** on the menu bar of the Application Window, then click on **Open Design** or enter Ctrl+O from the keyboard to open the Open Dialog Box.



Open Dialog Box

-Click on the desired project file name (*.prj) in the Open Dialog Box.
-Click on **OK**.

EXITING UNDER PRESSURE

-Click on **File** on the menu bar of the Application Window, then click on **Quit** or enter Ctrl+Q from the keyboard or click on the “x” in the upper left corner of the main Under Pressure window.

EXAMPLES

Example 1 - Aluminum Tube Design

Requirements:

- Aluminum Alloy Cylindrical Electronics Housing for Undersea service
- 4500 psi maximum external service pressure
- Internal diameter of 6.00"
- Internal length of 24.00"
- Minimum safety factor of 2.0 on buckling and stress (shell material failure)

Procedure:

-Follow the steps outlined in **GETTING STARTED** to set up the Under Pressure Application Window for this example.

-Use the cursor and keyboard to enter the **Project Title**, **Project Description**, and **Project Designer** in the Project Parameters portion of Application Window.

-Click on **CHOOSE** to open the Material Database Dialog Box.

-Select **Main Category - Metals** by clicking on the scroll arrow and clicking on Metals.

-Select **Sub-Category - Aluminum** by clicking on the scroll arrow and clicking on Aluminum.

-Select **Name - 6061-T6** as a first option for this example by clicking on scroll arrow and clicking on 6061-T6. Note 6061-T6 has a yield strength of 35,000 psi.

-Click on **Done** to return to the Application Window.

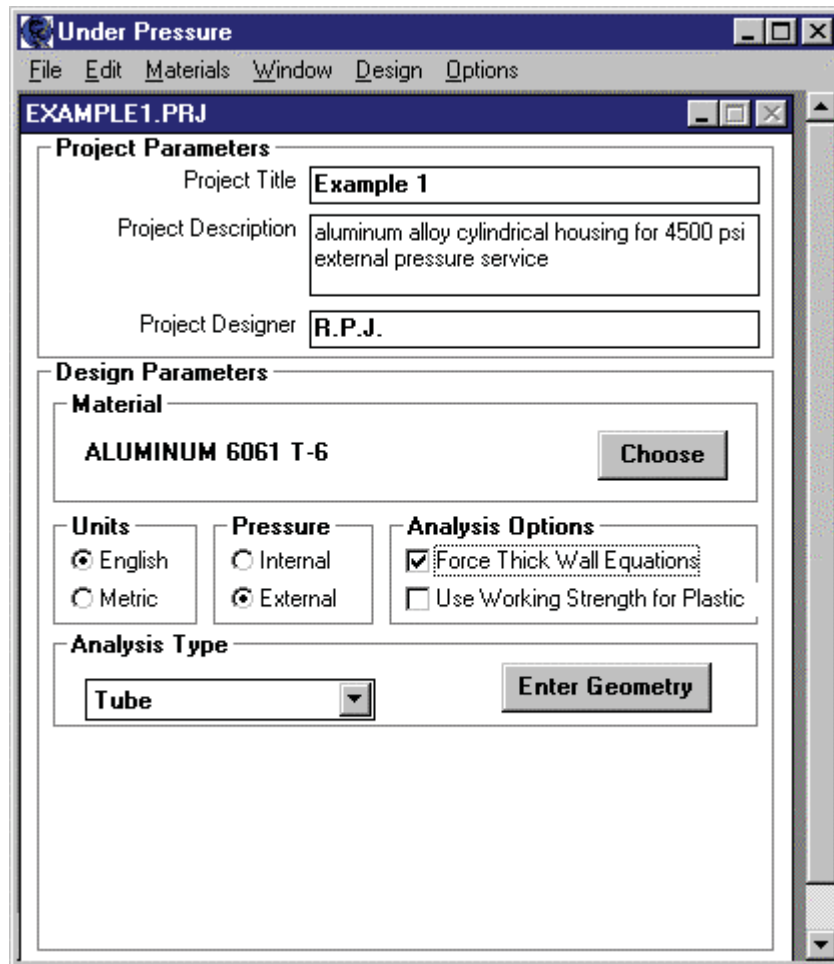
-Click the **Units - English** Option Button.

-Click the **Pressure - External** Option Button.

-Click the **Analysis Options - Force Thick Wall Equations** Check Box.

-Select **Analysis Type - Tube** by clicking on the scroll arrow and clicking on Tube.

- The Under Pressure Application Window should appear as follows:



EXAMPLE 1 APPLICATION WINDOW

-Click on **Enter Geometry** to open the Geometry Dialog Box.

-Use the cursor and keyboard to enter 6.00 inches for **Tube I.D.**

-Estimate the appropriate **Tube O.D.** or **Wall thickness** and enter this corresponding value using the cursor and keyboard.

-Use the cursor and keyboard to enter 24.00 inches for **Tube length**.

-Click the **Maintain I.D. constant** Option button given that the 6.00 inch Tube I.D. is a fixed constraint for this particular example.

-Click on **Perform Analysis** to generate analysis results.

-Review the analysis results in the Analysis Dialog Box, for this example a minimum safety factor of 2.0 was desired on buckling and stress for a design pressure of 4500 psi. Therefore the **Theoretical Failure** portion of the Analysis Dialog Box (titled Tube Analysis-External Pressure for this example) should indicate that **Thin Wall Buckling** and **Shell Failure** occur at a pressure greater than or equal to 9000 psi (Safety Factor equals Failure Pressure/Maximum Service Pressure, or $9000/4500 = 2$). A

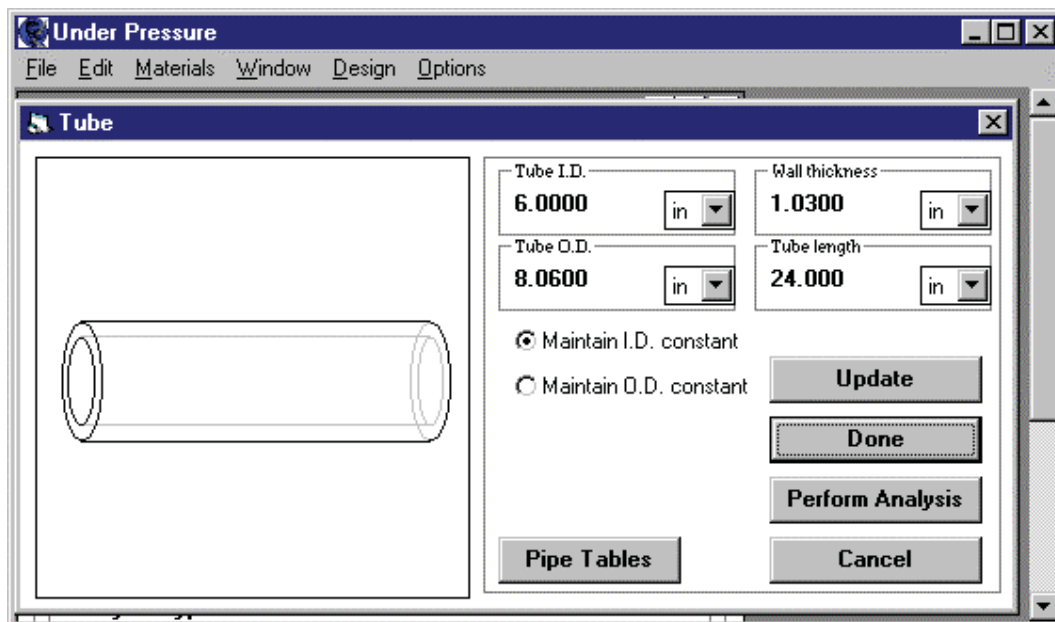
safety factor of two was arbitrarily selected for this example. In actual practice, the safety factor that is selected will depend on the specific requirements of the pressure vessel and the confidence of the pressure vessel designer. In addition to specific requirements that may exist, safety factors should be used by the pressure vessel designer to account for any number of variables that could affect the structural performance of a pressure housing design. Variables that could affect the pressure housing performance could include dimensional tolerances (imperfections), corrosion allowances, material properties, creep behavior (duration of load, temperature effects), cyclic loading (fatigue), dynamic loading, stress concentrations, residual stresses etc.

Variations in the pressure vessel geometry allowed by dimensional tolerances on the pressure boundary components can significantly effect structural performance. In particular, buckling of shells subjected to external pressure is sensitive to any geometric imperfections that may exist. Out of roundness (ID/OD) or concentric and/or wall thickness variations of tubes and variations in thickness such as flat spots on spheres and hemispherical endcaps can impact buckling resistance.

Stress concentrations include any deviations from the idealized pressure vessel geometry analyzed by Under Pressure such as O-ring grooves, through holes, blind holes, screw threads, notches, shoulders and generally any variations or discontinuities in wall thickness or curvature of the pressure vessel **geometry**.

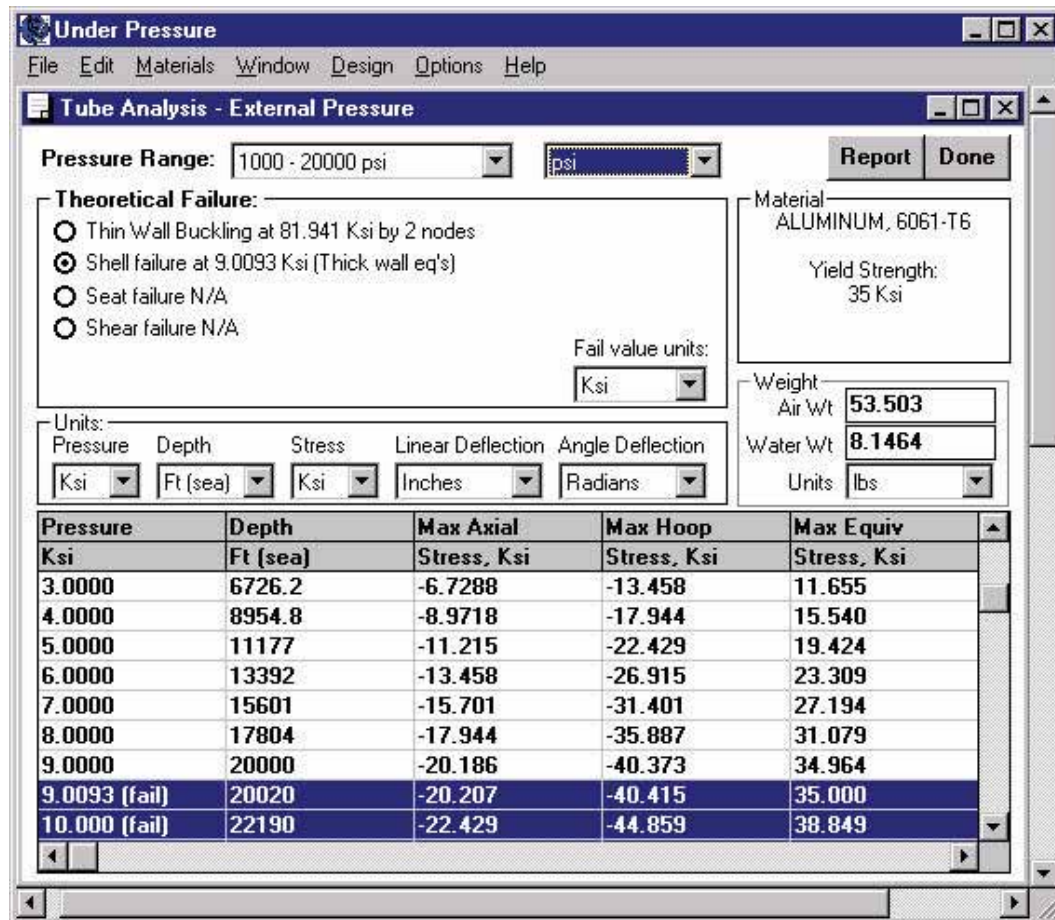
-Select **Done** in the Analysis Dialog Box to return to the Geometry Dialog Box to iterate on required **Wall thickness** (or alternatively **Tube O.D.**) until the requirements of this example aluminum alloy cylindrical housing are met.

-For aluminum alloy 6061-T6, a wall thickness of 1.03" is found to be adequate for this example as shown by setting up the Geometry Dialog Box below:



EXAMPLE 1 GEOMETRY DIALOG BOX (6061-T6)

-The Analysis Dialog Box for the geometry shown above appears as follows:



EXAMPLE 1 ANALYSIS DIALOG BOX (6061-T6)

-This analysis indicates the following results:

- Thin Wall Buckling occurs at 81,941 psi (S.F. = 18.2)
- Shell failure occurs at 9009 psi (S.F. = 2.0)
- Weight in air = 53.5 lb.
- Weight in water = 8.15 lb.

Note: This assumes the ends are capped with “weightless” end closures. Endcap weights must be added to all actual housing weights.

-In this example, the designer might consider a slight safety factor reduction to allow the use of stock 8” OD material.

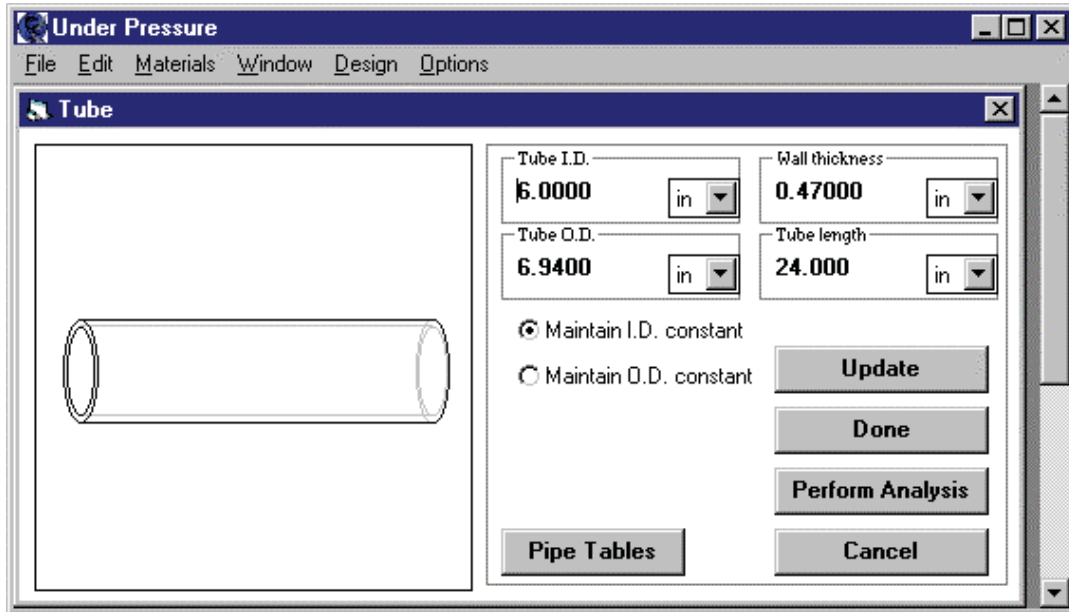
-Return to the Under Pressure Application Window.

-Select **Choose** and select aluminum alloy 7075-T6 from the material database. Select **Done** to return to the Application Window. Note the Yield Strength of 7075-T6 = 62,000 psi.

-Click **Perform Analysis** and note that the thin wall buckling is nearly identical, but that the fail pressure is much greater. Buckling is directly a function of material stiffness, not strength.

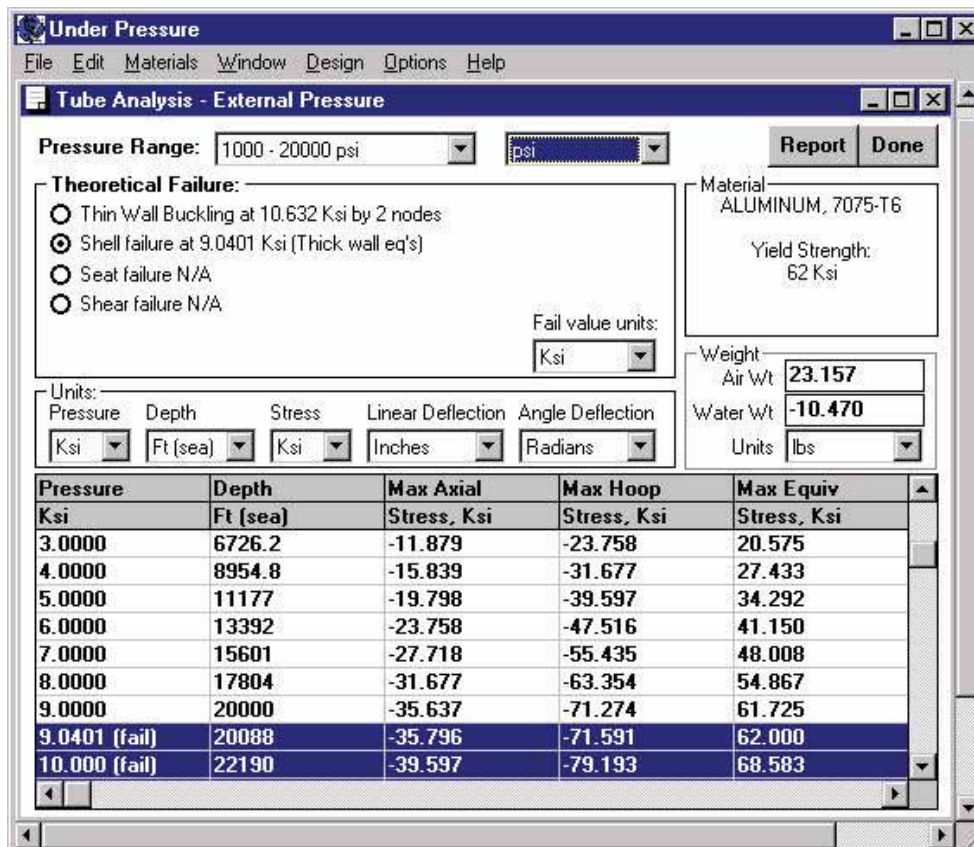
-Now select **Enter Geometry** to iterate on **Wall thickness** (or alternatively Tube O.D. for this new alloy).

-For aluminum alloy 7075-T6, a wall thickness of .47" is found to be adequate for this example as shown by setting up the Geometry Dialog Box below (Further detail on the specific information contained in the Analysis Dialog box is addressed later in this manual):



EXAMPLE 1 GEOMETRY DIALOG BOX (7075-T6)

-The Analysis Dialog Box for the geometry shown above appears as follows:



EXAMPLE 1 ANALYSIS DIALOG BOX (7075-T6)

-This analysis indicates the following results:

- Thin Wall Buckling occurs at 10,632 psi (S.F. = 2.36)
- Shell failure occurs at 9040 psi (S.F. = 2.0)
- Weight in air = 23.16 lb.
- Weight in water = -10.47 lb.

-The use of a higher strength aluminum alloy (7075-T6) for the cylindrical housing results in a design (wall thickness equals .47 inches) that generates 10.47 lb. of positive buoyancy when submerged as compared to a negative buoyancy of 8.15 lb. for the 6061-T6 design (wall thickness equals 1.03"). Again, note that if you change back to 6061-T6 the thin wall buckling does not change appreciably; whereas, the shell failure does. Further detail on the specific information contained in the Analysis Dialog box is addressed later in the section DETAILS ON ANALYSIS TYPE.

Example 2 - Aluminum Flat Circular Endcap Design:

Requirements:

- 7075-T6 Aluminum Alloy Flat Circular Endcap for Aluminum Cylindrical Housing design of Example 1.
- 4500 psi maximum external service pressure
- Plate Free Diameter of 6.00" (equal to Tube I.D. of Example 1)
- Plate Outside Diameter of 6.94" (equal to Tube O.D. of Example 1)
- Minimum safety factor of 2.0 on stress

Procedure:

-Follow the steps outlined in **GETTING STARTED** to set up the Under Pressure Application Window for this example.

-Use the cursor and keyboard to enter the **Project Title**, **Project Description**, and **Project Designer** in the Project Parameters portion of Application Window (optional).

-Click on **CHOOSE** to open the Material Database Dialog Box.

-Select **Main Category - Metals** by clicking on the scroll arrow and clicking on Metals.

-Select **Sub-Category - Aluminum** by clicking on the scroll arrow and clicking on Aluminum.

-Select **Name - 7075-T6** as a first option for this example by clicking on the scroll arrow and clicking on 7075-T6.

-Click on **Done** to return to the Application Window.

-Click the **Units - English** Option Button.

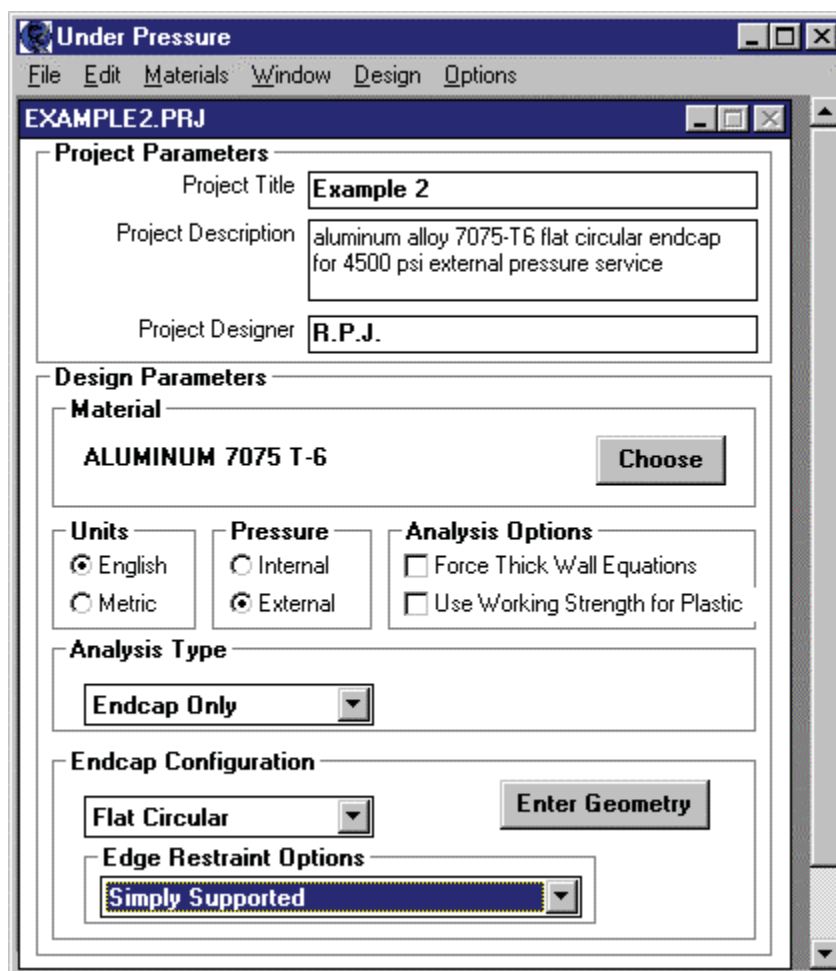
-Click the **Pressure - External** Option Button.

-Select **Analysis Type - Endcap Only** by clicking on the scroll arrow and clicking on Endcap Only.

-Select **Endcap Configuration - Flat Circular** by clicking on the scroll arrow and clicking on Flat Circular.

-Select **Edge Restraint Options - Simply Supported** by clicking on the scroll arrow and clicking on Simply Supported. For this example we will assume that the interface between the cylindrical tube of example 1 and the flat circular endcap of this example is such that the outer edges of the endcap can rotate during pressure loading (simply supported boundary condition). If the pressure housing design was such that the edges of the endcap cannot rotate during pressure loading, a Fixed Edge Restraint Option (clamped boundary condition) would be appropriate (for example, a welded-on endcap or bored solid bar).

-The Under Pressure Application Window should appear as follows:



EXAMPLE 2 APPLICATION WINDOW

-Click on **Enter Geometry** to open the Geometry Dialog Box.

-Use the cursor and keyboard to enter 6.94 inches for **Plate Outside Diameter**.

-Use the cursor and keyboard to enter 6.00 inches for **Plate Free Diameter**. Note: The free diameter is the unsupported diameter. The formulas used by Under Pressure can not account for material

outside the free diameter, except when calculating seat stress. The strengthening effect of this additional material is not considered in these calculations.

-Estimate **Plate thickness** and enter the corresponding value using the cursor and keyboard.

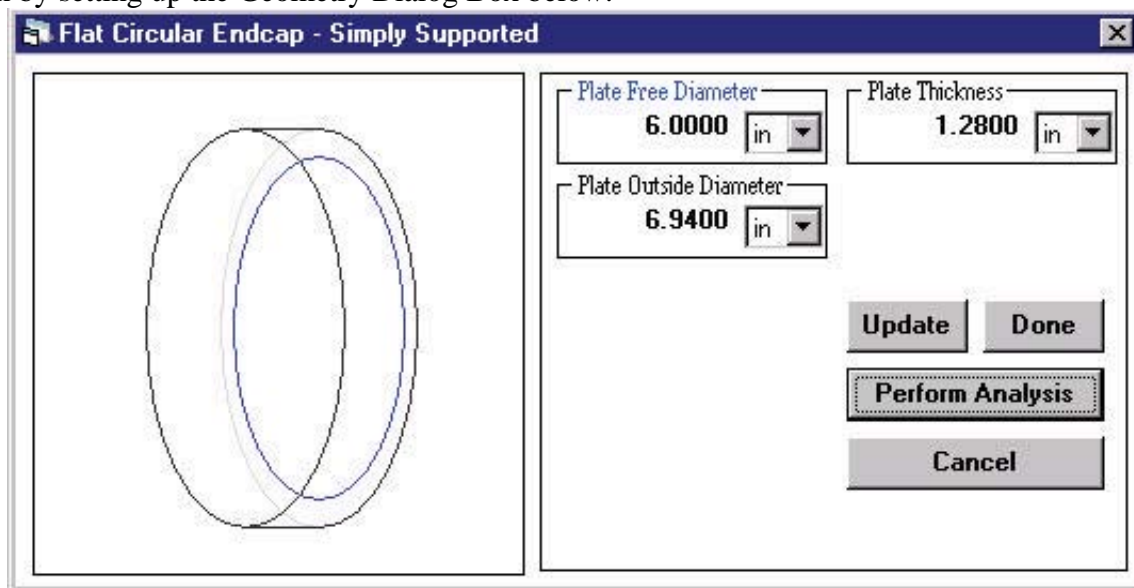
-Click on **Perform Analysis** to generate analysis results.

-Review the analysis results in the Analysis Dialog Box. For this example a minimum safety factor of 2.0 was desired on stress for a design pressure of 4500 psi. Therefore the **Theoretical Failure** portion of the Analysis Dialog Box (titled Flat Circular Endcap Analysis-External Pressure for this example) should indicate that **Radial Stress Failure, Tangential Stress Failure, and Seat Failure** occur at a pressure greater than or equal to 9000 psi (Safety Factor equals Failure Pressure/Maximum Service Pressure or $9000/4500 = 2$).

-Under Pressure provides a calculation of the maximum shear stress in flat circular endcaps. The maximum shear stress occurs at the plate free diameter and is equal to $(Pxd)/(4xt)$ where P = applied pressure, d = plate free diameter and t = plate thickness. Shear stresses are insignificant as compared to radial and tangential stresses for flat circular plates unless the ratio of the plate thickness to plate free diameter (t/d) approaches $3/8$ for a fixed edge restraint or $5/8$ for a simply supported edge restraint (these ratios assume a Poissons's Ratio of .3 and a shear strength equal to $1/2$ of the uniaxial strength).

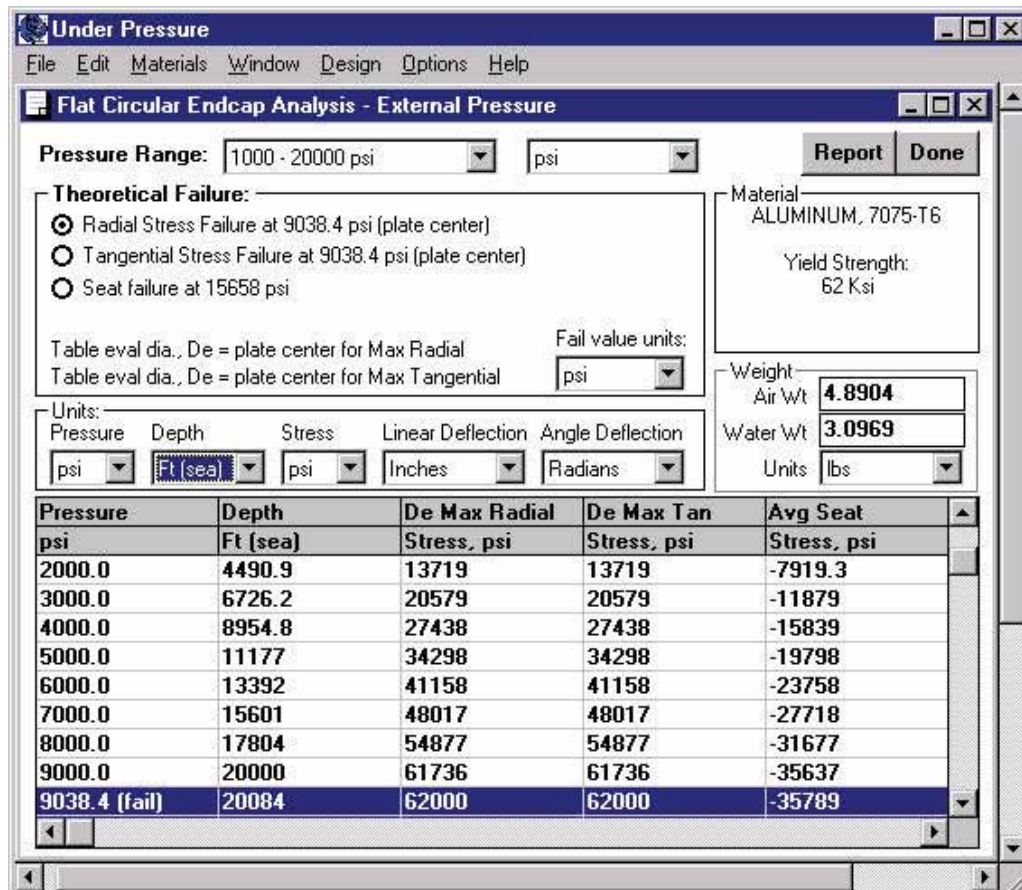
-Select **Done** in the Analysis Dialog Box to return to the Geometry Dialog Box to iterate on the required **Plate thickness** until the requirements of this example aluminum alloy 7075-T6 flat circular endcap are met.

-For aluminum alloy 7075-T6, a plate thickness of 1.28" is found to be adequate for this example as shown by setting up the Geometry Dialog Box below:



EXAMPLE 2 GEOMETRY DIALOG BOX

-The Analysis Dialog Box for the geometry shown above appears as follows:



EXAMPLE 2 ANALYSIS DIALOG BOX

-This analysis indicates the following results:

- Radial Stress Failure = Tangential Stress failure occurs at 9038 psi (S.F. = 2.0)
- Seat failure occurs at 15,658 psi (S.F. = 3.48)
- Weight in air = 4.89 lb.
- Weight in water = 3.10 lb.

-Note that the average seat stress at the outer edges of the flat circular plate is equal to axial stress in the 7075-T6 tube of example 1. Further detail on the specific information contained in the Analysis Dialog box is addressed later in section DETAILS ON ANALYSIS TYPE.

Note: When using certain materials it is often advisable to construct such endcaps from sawn plate (rather than round bar) to ensure fully tempered material at the maximally stressed plate center.

Example 3 - Aluminum Flat Annular Endcap Design:

Requirements:

- 7075-T6 Aluminum Alloy Flat Annular Endcap for Aluminum Cylindrical Housing design of Example 1.
- 4500 psi maximum external service pressure
- Plate Free Diameter of 6.00" (equal to Tube I.D. of Example 1)
- Plate Outside Diameter of 6.94" (equal to Tube O.D. of Example 1)
- 1.00" hole in the center of plate for an electrical connector

- Minimum safety factor of 1.33 on membrane stresses (tangential and radial stress in plate)

Procedure:

-Follow the steps outlined in **GETTING STARTED** to set up the Under Pressure Application Window for this example.

-Use the cursor and keyboard to enter the **Project Title, Project Description, and Project Designer** in the Project Parameters portion of Application Window.

-Click on **CHOOSE** to open the Material Database Dialog Box.

-Select **Main Category - Metals** by clicking on the scroll arrow and clicking on Metals.

-Select **Sub-Category - Aluminum** by clicking on the scroll arrow and clicking on Aluminum.

-Select **Name - 7075-T6** as a first option for this example by clicking on the scroll arrow and clicking on 7075-T6.

-Click on **Done** to return to the Application Window.

-Click the **Units - English** Option Button.

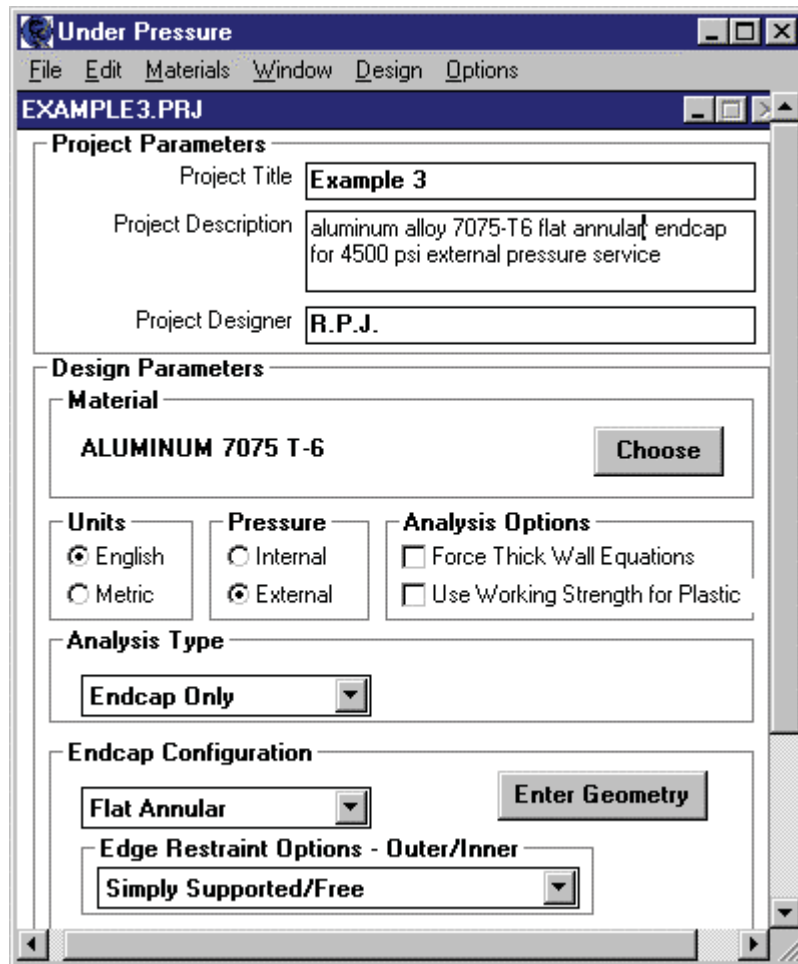
-Click the **Pressure - External** Option Button.

-Select **Analysis Type - Endcap Only** by clicking on the scroll arrow and clicking on Endcap Only.

-Select **Endcap Configuration - Flat Annular** by clicking on the scroll arrow and clicking on Flat Annular.

-Select **Edge Restraint Options-Outer/Inner - Simply Supported/Free** by clicking on the scroll arrow and clicking on Simply Supported. For this example, we will assume that the outer interface between the cylindrical tube of example 1 and the flat annular endcap of this example is such that the outer edges of the endcap can rotate during pressure loading (simply supported boundary condition). For this example, we will also assume that the compliance/clearances between the center hole in the aluminum annular endcap and the radial surfaces of the electrical connector are such that the edges of the hole are essentially unconstrained (i.e. free edge restraint) by the presence of the connector during pressure loading.

-The Under Pressure Application Window should appear as follows:



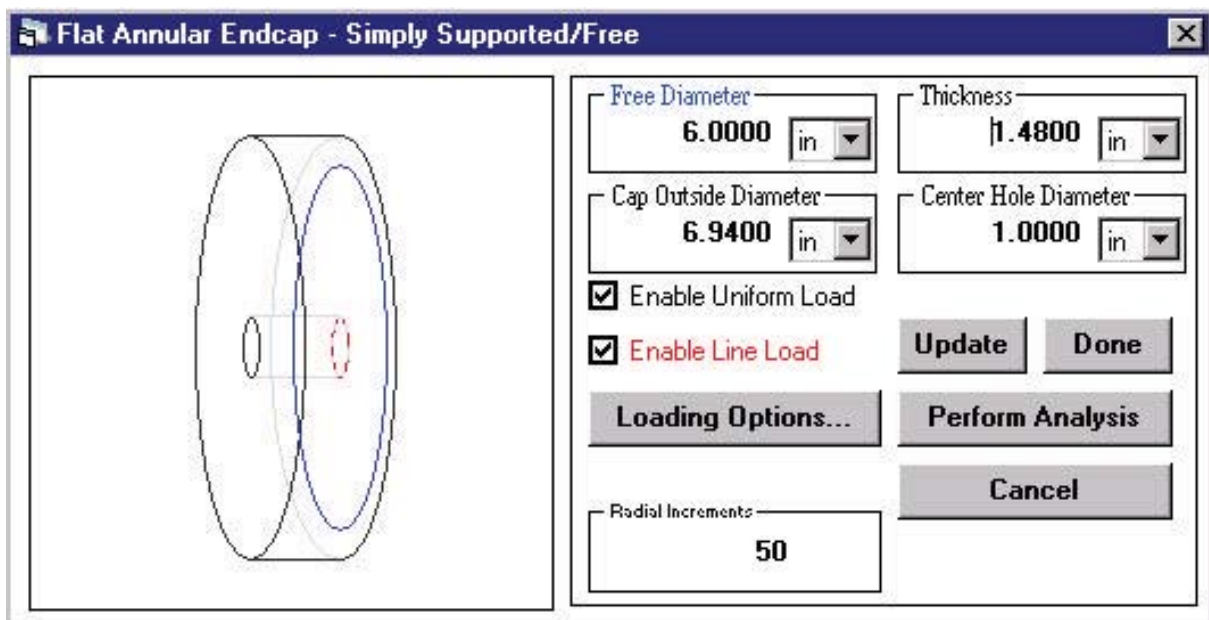
EXAMPLE 3 APPLICATION WINDOW

- Click on **Enter Geometry** to open the Geometry Dialog Box.
- Use the cursor and keyboard to enter 1.00 inches for **Center Hole Diameter**.
- Use the cursor and keyboard to enter 6.94 inches for **Cap Outside Diameter**.
- Use the cursor and keyboard to enter 6.00 inches for **Free Diameter**.
- Estimate **Thickness** and enter the corresponding value using the cursor and keyboard.
- Click on **Enable Uniform Load** and **Enable Line Load** Check Boxes. By checking Enable Uniform Load, the program will apply uniform external pressure to the flat external surface of the annular plate. By checking Enable Line Load, the program will apply an appropriate line load to the circumference of the hole at the center of the plate. This line load is equivalent to the pressure load that exists on the flat external surface of the connector. This line load simulates the load on the edge of the hole in the plate that is generated by the pressure loading on the item installed into the hole in the plate (in this example, the electrical connector).
- Click on **Perform Analysis** to generate analysis results.

-Review the analysis results in the Analysis Dialog Box, for this example, a minimum safety factor of 1.33 was desired on membrane stress (radial and tangential stress) for a design pressure of 4500 psi. Therefore the **Theoretical Failure** portion of the Analysis Dialog Box (titled Flat Annular Endcap Analysis-External Pressure for this example) should indicate that **Radial Stress Failure and Tangential Stress Failure** occur at a pressure greater than or equal to 6000 psi (Safety Factor equals Failure Pressure/Maximum Service Pressure = $6000/4500 = 1.33$). The use of different safety factors in the tube and flat circular endcap of examples 1 and 2 (safety factor = 2.00) than in this example is done arbitrarily and not intended to imply that less safety factor is actually required for flat annular endcap.

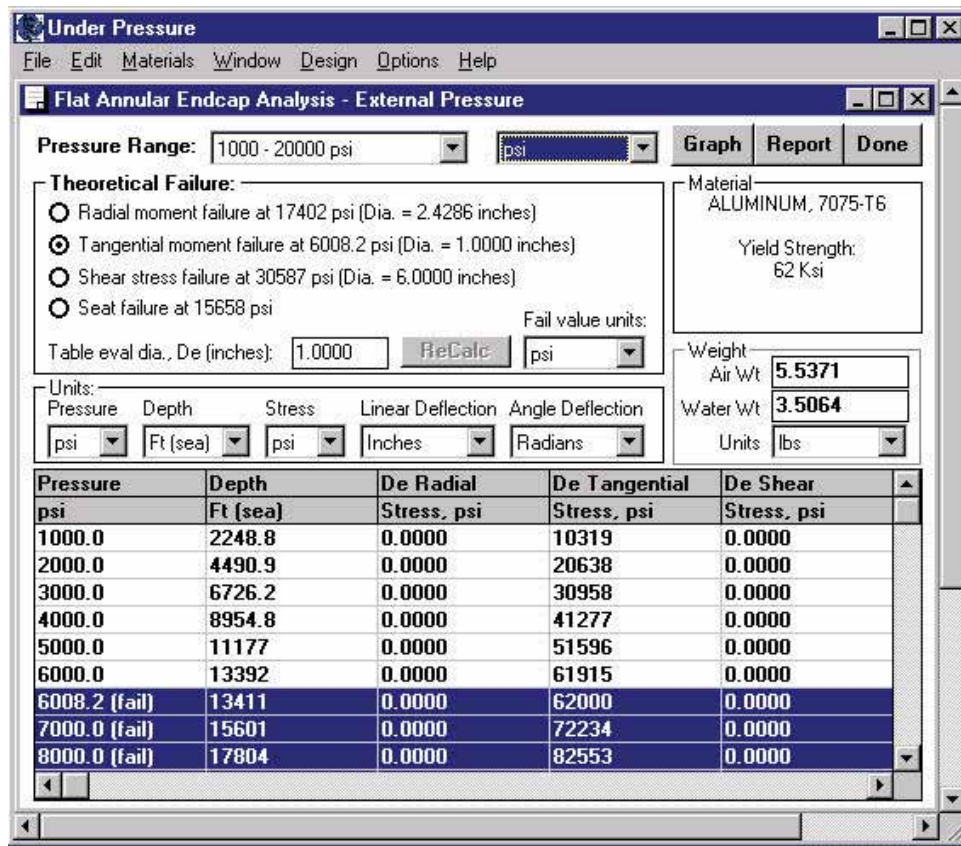
-Select **Done** in the Analysis Dialog Box to return to the Geometry Dialog Box to iterate on required **thickness** until the design requirements of the aluminum alloy 7075-T6 flat annular endcap are met.

-For aluminum alloy 7075-T6, a thickness of 1.48" is found to be adequate for this example as shown by setting up the Geometry Dialog Box below:



EXAMPLE 3 GEOMETRY DIALOG BOX

-The Analysis Dialog Box for the geometry shown above appears as follows:



EXAMPLE 3 ANALYSIS DIALOG BOX

-This analysis indicates the following results:

- Radial Stress Failure occurs at 17,402 psi (safety factor = 3.87) at a diameter of 2.43 inches
- Tangential Stress Failure occurs at 6008 psi (safety factor = 1.34) at a diameter of 1.00 inches (at the hole diameter)
- Weight in air = 5.54 lb.
- Weight in water = 3.51 lb.

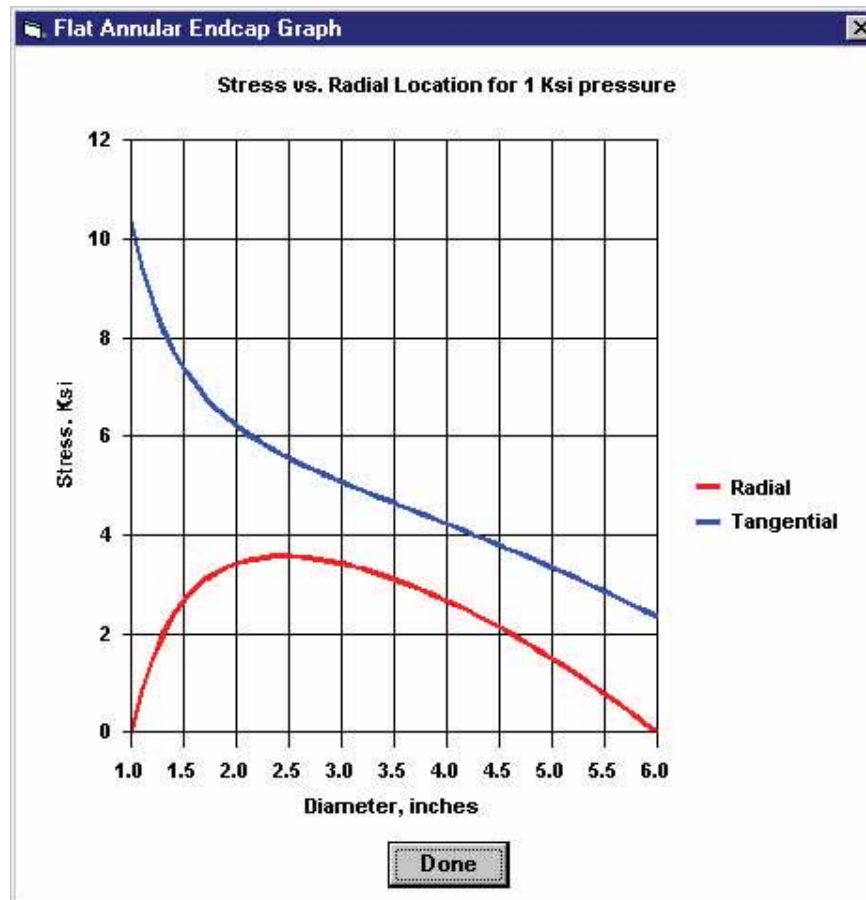
-If you scroll to the right on the horizontal scroll bar, you'll find additional columns with data analysis (this is true for nearly all data analysis screens in Under Pressure):

| De Angular Defl Radians | HD Radial Stress, psi | HD Tangential Stress, psi | HD Shear Stress, psi | HD Vert Defl Inches | HD Angular Defl Radians |
|----------------------------|--------------------------|------------------------------|-------------------------|------------------------|----------------------------|
| 0.00067693 | 0.0000 | 10319 | 0.0000 | -0.0019612 | 0.00067693 |
| 0.0013539 | 0.0000 | 20638 | 0.0000 | -0.0039225 | 0.0013539 |
| 0.0020308 | 0.0000 | 30958 | 0.0000 | -0.0058837 | 0.0020308 |
| 0.0027077 | 0.0000 | 41277 | 0.0000 | -0.0078450 | 0.0027077 |
| 0.0033847 | 0.0000 | 51596 | 0.0000 | -0.0098062 | 0.0033847 |
| 0.0040616 | 0.0000 | 61915 | 0.0000 | -0.011767 | 0.0040616 |
| 0.0040672 | 0.0000 | 62000 | 0.0000 | -0.011784 | 0.0040672 |
| 0.0047385 | 0.0000 | 72234 | 0.0000 | -0.013729 | 0.0047385 |
| 0.0054155 | 0.0000 | 82553 | 0.0000 | -0.015690 | 0.0054155 |

| FD Radial Stress, psi | FD Tangential Stress, psi | FD Shear Stress, psi | FD Vert Defl Inches | FD Angular Defl Radians | Avg Seat Stress, psi |
|--------------------------|------------------------------|-------------------------|------------------------|----------------------------|-------------------------|
| 0.0000 | 2349.7 | -1013.5 | 0.0000 | 0.00092485 | -3959.7 |
| 0.0000 | 4699.5 | -2027.0 | 0.0000 | 0.0018497 | -7919.3 |
| 0.0000 | 7049.2 | -3040.5 | 0.0000 | 0.0027746 | -11879 |
| 0.0000 | 9399.0 | -4054.1 | 0.0000 | 0.0036994 | -15839 |
| 0.0000 | 11749 | -5067.6 | 0.0000 | 0.0046243 | -19798 |
| 0.0000 | 14098 | -6081.1 | 0.0000 | 0.0055491 | -23758 |
| 0.0000 | 14118 | -6089.4 | 0.0000 | 0.0055567 | -23790 |
| 0.0000 | 16448 | -7094.6 | 0.0000 | 0.0064740 | -27718 |
| 0.0000 | 18798 | -8108.1 | 0.0000 | 0.0073988 | -31677 |

EXAMPLE 3 ANALYSIS DIALOG BOX (cont.)

-Clicking on **Graph** in the Analysis Dialog Box generates a plot of radial and tangential stresses in the plate for a 1 Ksi external pressure load as a function of the location along the plate diameter. This plot confirms that the maximum membrane stress in the plate is in the tangential direction and occurs at the edge of the hole in the center of the plate. Note that 50 points were used to generate these stress curves. The number of points used in these curves corresponds to the number of **Radial Increments** that the user selects in the Geometry Dialog Box for Flat Annular Endcaps. Further detail on the specific information contained in the Analysis Dialog box is addressed later in section DETAILS ON ANALYSIS TYPE. The hole at the center of a flat annular endcap is a stress concentrator. The presence of a hole, regardless of size, acts to increase the magnitude of stresses in the plate in the local vicinity of the hole. A direct comparison of a flat circular plate and a flat annular plate reveals the magnitude of stress concentration around the hole (simply supported edge restraint for flat circular plates should be compared to simply supported/free edge restraint for flat annular endcaps, or fixed edge restraint for flat circular plates should be compared to fixed/free edge restraint for flat annular endcaps). For a given material, plate thickness, plate free diameter and pressure loading, the addition of a hole on the plate centerline acts to approximately double tangential stresses in the plate at the edge of the hole (for cases where the hole diameter is small as compared to the plate free diameter). The addition of the hole with free edge restraint acts to relieve the plates ability to carry radial stress at the hole (radial stresses at the edge of the hole go to zero). As a consequence, the tangential stress must bear the portion of the load picked up as radial stress pre-hole, with the result that the tangential stress increases by a factor of approximately two at the edge of the hole. This result can be seen in flat annular endcap graph below.



EXAMPLE 3 FLAT ANNULAR ENDCAP GRAPH

Example 4 - Plastic Tube Design:

Requirements:

- Acetal Plastic (Delrin) Cylindrical Camera Housing for Undersea service
- 200 ft. sea water external pressure
- Minimum internal diameter of 4.00"
- Internal Length of 10.00"
- Minimum safety factor of 3.0 on buckling
- Maximum Membrane Stress in Plastic of 3000 psi at design depth (200 ft. sea water)
- Cylindrical housing to generate 6 lb. of positive buoyancy when submerged

Procedure:

-Follow the steps outlined in **GETTING STARTED** and previous examples to set up the Under Pressure Application Window for this example.

-Click on **CHOOSE** to open the Material Database Dialog Box.

-Select **Main Category - Plastics** by clicking on the scroll arrow and clicking on Plastics.

-Select **Sub-Category - Thermoplastics** by clicking on the scroll arrow and clicking on Thermoplastics.

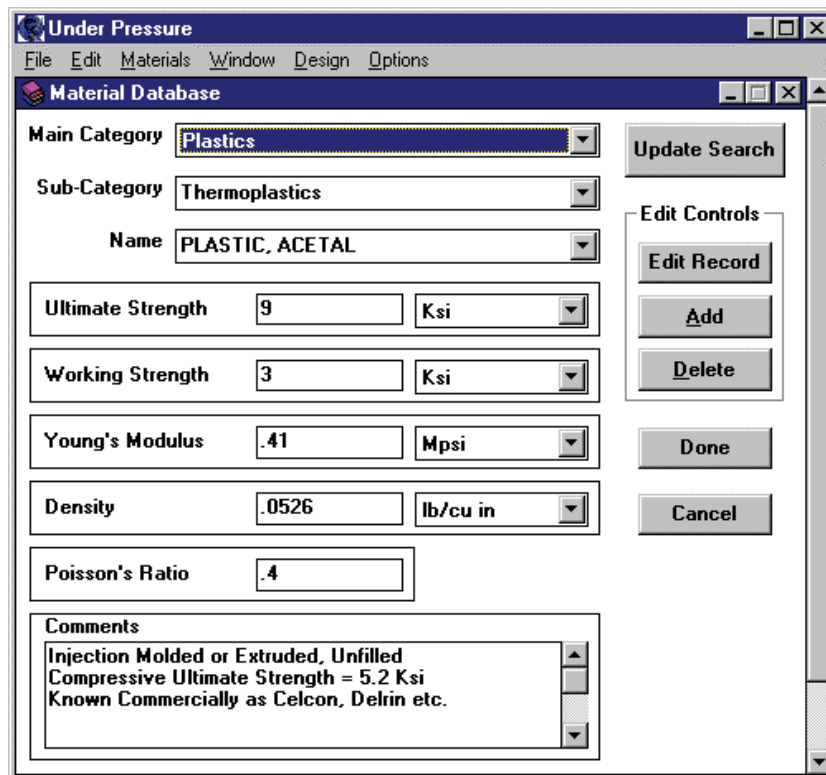
-Select **Name - PLASTIC, ACETAL** as a first option for this example by clicking on the scroll arrow and clicking on PLASTIC, ACETAL.

-Click on **Edit Record**.

-Use the cursor and keyboard to change **Working Strength** to **3 Ksi**.

-Click on **Done**

-The Material Database Dialog Box should appear as follows:



EXAMPLE 4 MATERIAL DATABASE DIALOG BOX

Note: In this instance the designer has elected to use a less conservative working strength of 3ksi versus that given in the provided database.

-Click on **Done** to return to the Application Window.

-Click the **Units - English** Option Button.

-Click the **Pressure - External** Option Button.

-Click the **Analysis Options - Force Thick Wall Equations** Check Box.

-Click the **Analysis Options - Use Working Strength for Plastic** Check Box.

-Select **Analysis Type - Tube** by clicking on the scroll arrow and clicking on Tube.

-The Under Pressure Application Window should appear as follows:

The screenshot shows the 'Under Pressure' application window with the following content:

- Project Parameters:**
 - Project Title: Example 4
 - Project Description: acetal plastic cylindrical housing for 200 feet sea water service
 - Project Designer: R.P.J.
- Design Parameters:**
 - Material:** PLASTIC, ACETAL (with a 'Choose' button)
 - Units:** English (selected), Metric
 - Pressure:** Internal, External (selected)
 - Analysis Options:** Force Thick Wall Equations (checked), Use Working Strength for Plastic (checked)
 - Analysis Type:** Tube (selected in a dropdown menu, with an 'Enter Geometry' button)

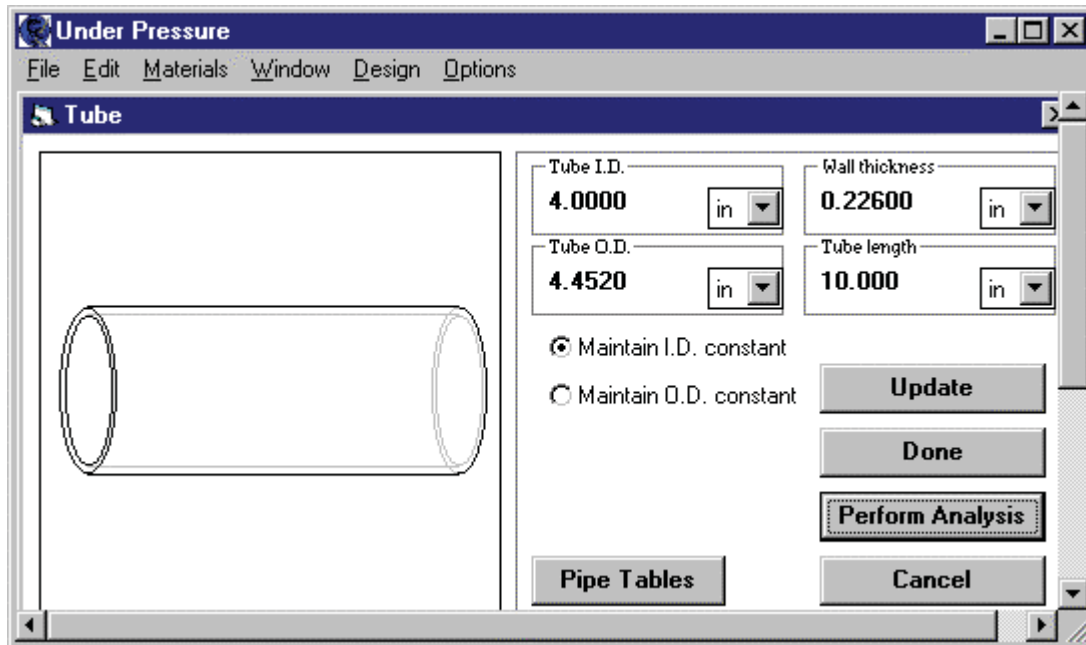
EXAMPLE 4 APPLICATION WINDOW

- Click on **Enter Geometry** to open the Geometry Dialog Box.
- Use the cursor and keyboard to enter 4.00 inches for **Tube I.D.**
- Estimate appropriate **Tube O.D.** or **Wall thickness** and enter the corresponding value using the cursor and keyboard.
- Use the cursor and keyboard to enter 10.00 inches for **Tube length.**
- Click on **Perform Analysis** to generate analysis results.
- Review the analysis results in Analysis Dialog Box, for this example a minimum safety factor of 3.0 was desired on buckling at a depth of 200 feet seawater. Therefore the **Theoretical Failure** portion of the Analysis Dialog Box (titled Tube Analysis-External Pressure for this example) should indicate that **Thin Wall Buckling** occurs at a depth greater than or equal to 600 feet (Safety Factor equals Failure Depth/Maximum Service Depth = $600/200 = 3$). A safety factor of 3 was arbitrarily selected for this example. In actual practice, the safety factor that is selected will depend on specific requirements of the pressure vessel and the confidence of the pressure vessel designer.

-The Analysis Dialog Box should also indicate that the maximum hoop stress at a depth of 200 feet does not exceed 3000 psi (selected Working Strength of plastic for this example)

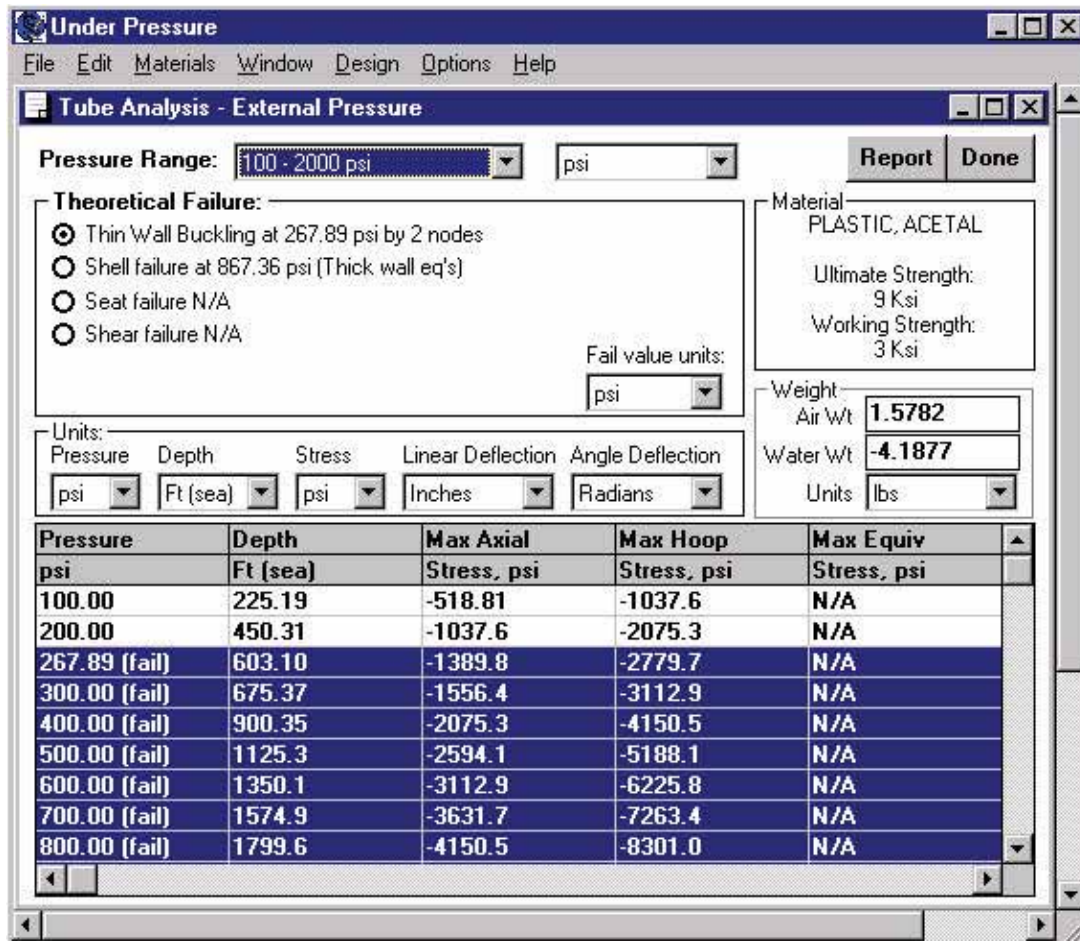
-Select **Done** in the Analysis Dialog Box to return to Geometry Dialog Box to iterate on required **Wall thickness** (or alternatively **Tube O.D.**) until requirements of this example Acetal cylindrical housing are met.

-For an Acetal tube with a Tube I.D. of 4.00 inches, a wall thickness of .226" is found to be adequate for the structural design requirements (stress, buckling) of this example as shown by setting up the Geometry Dialog Box below:



EXAMPLE 4 GEOMETRY DIALOG BOX (4.00" I.D.)

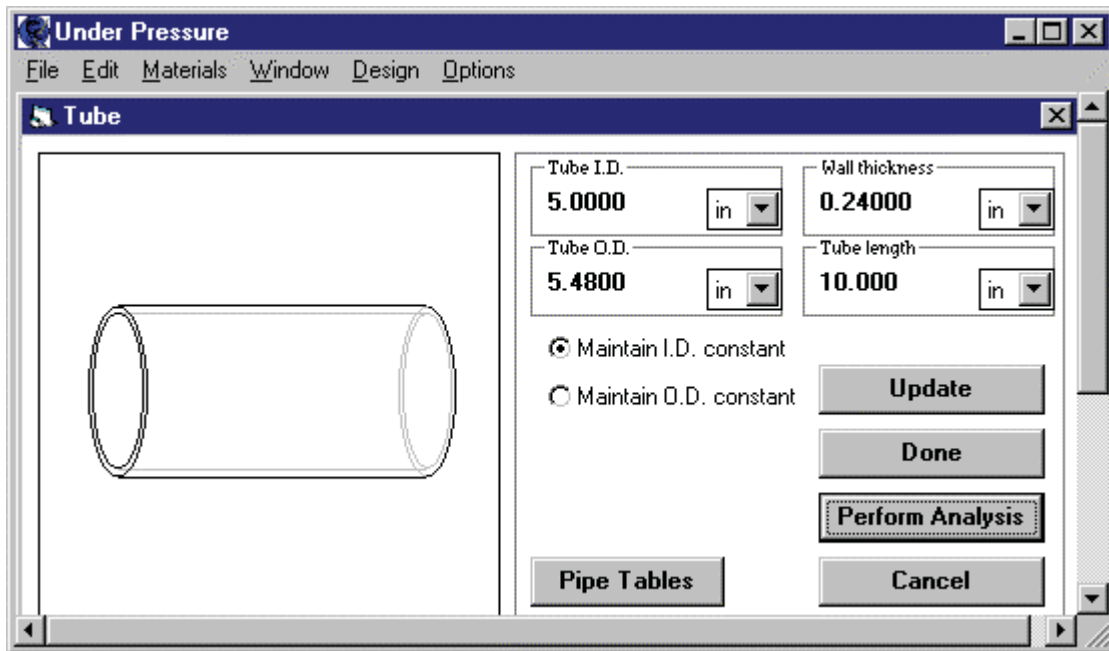
-The Analysis Dialog Box for the geometry shown above appears as follows:



EXAMPLE 4 ANALYSIS DIALOG BOX (4.00" I.D.)

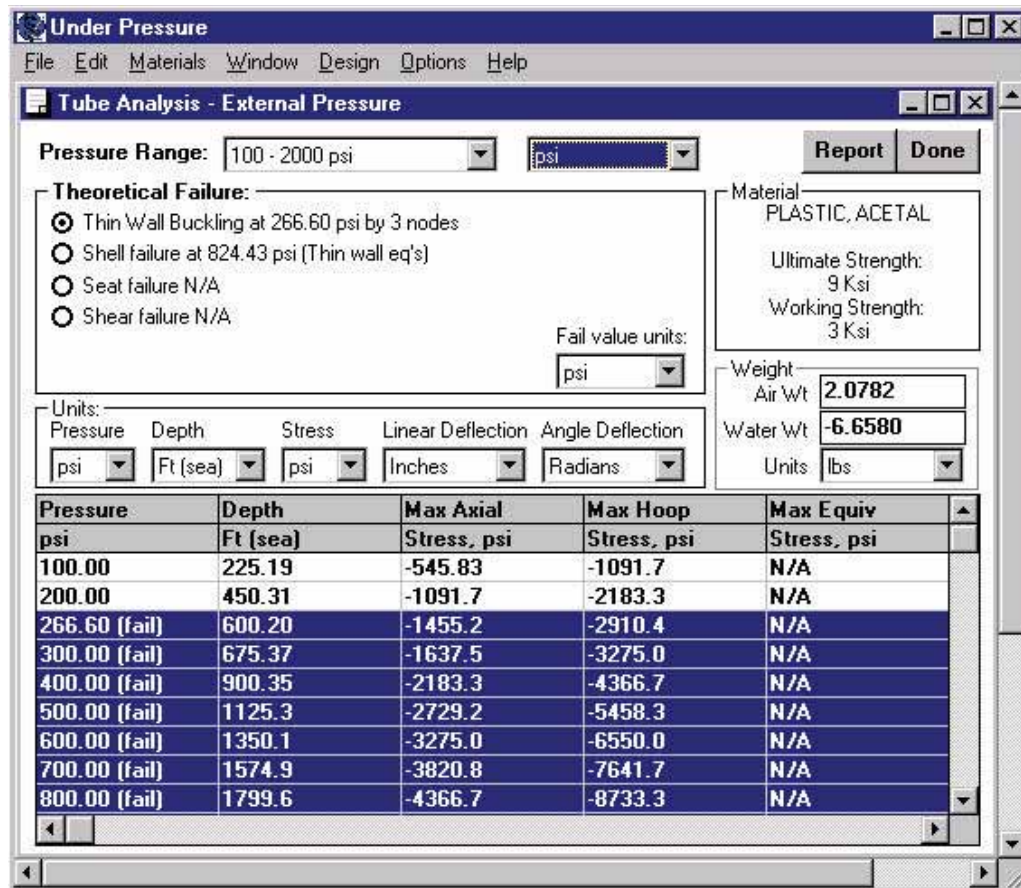
-This Analysis Dialog Box indicates that the requirements for stress and buckling have been met, but that the Acetal tube only generates 4.19 lb. of positive buoyancy. Since reducing the Wall Thickness further to generate more buoyancy will result in a housing that does not meet a buckling safety factor of 3.0, the Tube I.D. must be increased and the required Wall Thickness rechecked.

-By increasing the Acetal Tube I.D. to 5.0 inches and iterating on Wall Thickness until all requirements have been met, a Wall Thickness of .24 inches is found to be adequate for this example as shown by setting up the Geometry Dialog Box below:



EXAMPLE 4 GEOMETRY DIALOG BOX (5.00” I.D.)

-The Analysis Dialog Box for the geometry shown above appears as follows:



EXAMPLE 4 ANALYSIS DIALOG BOX (5.00” I.D.)

-This analysis indicates the following results:

- Thin Wall Buckling occurs at 600 feet (safety factor equals 3.0)
- Hoop stress in tube is below selected Working Strength (3000 psi) at a maximum service depth of 200 feet sea water
- Weight in air = 2.08 lb.
- Weight in water = -6.66 lb. (the tube generates in excess of 6 lb. of positive buoyancy when submerged)

Further detail on the specific information contained in the Analysis Dialog box is addressed later in section DETAILS ON ANALYSIS TYPE.

DETAILS ON MATERIALS

MAIN CATEGORIES

Clicking on **CHOOSE** in the Under Pressure Application Window allows the user to select the pressure vessel material. Under Pressure comes with a database of commonly used pressure vessel materials, but also allows the user to edit the material's data base for their own specific needs. Materials are defined by the following parameters:

- **Main Category**
- **Sub-Category**
- **Name**

Main Categories are pre-defined and cannot be edited by the program user. Five Main Categories exist for the material's database:

- **All**
- **Ceramics**
- **Glass**
- **Metals**
- **Plastics**

Main Categories are pre-set by the program because they define the material properties needed by the program to perform an analysis. The material properties needed for each of the Main Categories are as follows:

Ceramics

- Ultimate Strength (tensile)
- Ultimate Strength (compressive)
- Young's Modulus
- Density
- Poisson's Ratio

Glass

- Ultimate Strength (tensile)
- Ultimate Strength (compressive)
- Young's Modulus
- Density
- Poisson's Ratio

Metals

- Yield Strength
- Young's Modulus
- Density
- Poisson's Ratio

Plastics

- Ultimate Strength
- Working Strength

- Young's Modulus
- Density
- Poisson's Ratio

The All Main Category allows the user to view all materials in all Main Categories at once. The All Main Category defaults to material properties for the Metals Main Category.

SUB-CATEGORIES

The **Sub-Category** parameter allows the user to further organize materials that fall into the same Main Category. Under Pressure comes with the following Sub Categories:

Main Category: Ceramics
Sub Categories: All, Alumina, Silicon Carbide

Main Category: Glass
Sub Categories: All, Glass

Main Category: Metals
Sub Categories: All, Aluminum, Nickel, Stainless Steel, Steel, Titanium

Main Category: Plastics
Sub Categories: All, Composites, Thermoplastics

NAME

The **Name** parameter designates the specific material alloy or composition.

MATERIAL DATABASE

Under Pressure comes with the following materials database where the following abbreviations are used:

U.S.T. = Ultimate Strength (tensile)
U.S.C. = Ultimate Strength (compressive)
Y.M. = Young's Modulus
Den. = Density
P.R. = Poisson's Ratio
Y.S. = Yield Strength
U.S. = Ultimate Strength
W.S. = Working Strength

Main Category: Ceramics

| Sub-Cat | Name | U.S.T. Ksi | U.S.C. Ksi | Y.M. Mpsi | Den. lb/ in ³ | P.R. |
|-----------------|-----------------|---------------|---------------|--------------|-----------------------------|------|
| Alumina | 94% | 28 | 305 | 44 | .13 | .21 |
| Alumina | 96% | 32 | 300 | 47 | .134 | .23 |
| Alumina | 99.5% | 38 | 380 | 54 | .14 | .22 |
| Alumina | Sapphire | 40 | 300 | 50 | .143 | .29 |
| Silicon Carbide | Silicon Carbide | 44.5 | 362 | 57 | .11 | .19 |

Main Category: Glass

| Sub-Cat | Name | U.S.T. Ksi | U.S.C. Ksi | Y.M. Mpsi | Den. lb/in ³ | P.R. |
|---------|--------|---------------|---------------|--------------|----------------------------|------|
| Glass | BK-7 | 5 | 210 | 11.9 | .0906 | .206 |
| Glass | Pyrex | 5 | 210 | 8.9 | .081 | .2 |
| Glass | Quartz | 5 | 210 | 10.57 | .079 | .19 |
| Glass | Vycor | 5 | 210 | 10.57 | .079 | .19 |

Main Category: Metals

| | | | | | |
|--------------|-----------------|-----|------|------|------|
| Aluminum | 2024-T3 | 36 | 10.5 | .101 | .33 |
| Aluminum | 5052-H34 | 25 | 10.1 | .097 | .33 |
| Aluminum | 5082-H32 | 22 | 10.1 | .097 | .33 |
| Aluminum | 5456-H111 | 26 | 10.2 | .096 | .33 |
| Aluminum | 6061-T6 | 35 | 9.9 | .098 | .33 |
| Aluminum | 6262-T9 | 55 | 9.0 | .098 | .345 |
| Aluminum | 7075-T6 | 62 | 10.3 | .101 | .33 |
| Nickel | K Monel | 90 | 26 | .306 | .32 |
| Nickel | Monel | 25 | 26 | .319 | .32 |
| Stain. Steel | 17-4PH H1150 | 100 | 28.5 | .284 | .32 |
| Stain. | 304, 303, | 26 | 29 | .286 | .27 |

| | | | | | |
|--------------|-------------------------|-------------|--------------|----------------------------|------|
| Steel | 304L, 316L (see 316) | | | | |
| Stain. Steel | 316 | 26 | 29 | .286 | .27 |
| Stain. Steel | 17-4PH H1075 | 125 | 28.5 | .283 | .27 |
| Stain. Steel | 17-4PH H900 | 170 | 28.5 | .282 | .27 |
| Steel | Carbon | 36 | 29 | .284 | .32 |
| Steel | Low Alloy | 70 | 29 | .283 | .32 |
| Sub-Cat | Name | Y.S. Ksi | Y.M. Mpsi | Den. lb/in ³ | P.R. |
| Titanium | Comm. Pure | 55 | 15.5 | .163 | .34 |
| Titanium | Ti-5Al- 2.5Sn | 110 | 15.5 | .162 | .31 |
| Titanium | Ti-6Al-4V | 119 | 16 | .16 | .31 |

Main Category: Plastics

| Sub-Cat | Name | U.S. Ksi | W.S. Ksi | Y.M. Mpsi | Den. lb/in ³ | P.R. |
|--------------------|--------------------|-------------|-------------|--------------|----------------------------|------|
| Compo- site | Glass/ Epoxy | 10 | 1 | 2 | .0667 | .4 |
| Thermo- plastic | Acetal | 9 | .9 | .41 | .0526 | .4 |
| Thermo- plastic | Acrylic | 8 | .8 | .35 | .0417 | .35 |
| Thermo- plastic | Nylon-6 | 9 | .9 | .2 | .04 | .4 |
| Thermo- plastic | Polycar- Bonate | 8 | .8 | .3 | .0435 | .4 |
| Thermo- plastic | Poly prop. | 4.3 | .43 | .16 | .0323 | .4 |
| Thermo- plastic | PVC | 6 | .6 | .35 | .0476 | .36 |

DEFINITION OF MATERIAL PROPERTIES

-Ultimate Strength (tensile) (for Main Categories Glass, Ceramics): Maximum uniaxial tensile stress material can withstand without failure.

-Ultimate Strength (compressive) (for Main Categories Glass, Ceramics): Maximum uniaxial compressive stress material can withstand without failure.

-Yield Strength (for Main Category Metals): Uniaxial stress at which yield (permanent deformation) of the material is initiated.

-Ultimate Strength (for Main Category Plastics): Maximum uniaxial stress material can withstand without failure.

-Working Strength (for Main Category Plastics): Maximum stress allowed in material during service as defined by pressure vessel designer.

-Young's Modulus (for all Main Categories): Average ratio of stress to strain for stress below the proportional limit, measurement of material stiffness.

-Density (for all Main Categories): Mass or weight per unit volume of material.

-Poisson's Ratio (for all Main Categories): Absolute value of the ratio of lateral strain over axial strain.

MATERIAL DATABASE REFERENCES

Glass:

-See Comments in Material Database Dialog Box for specific vendor technical data sheet references.

Ceramics:

-See Comments in Material Database Dialog Box for specific vendor technical data sheet references.

Metals:

-“MIL-HDBK-5, Metallic Materials and Elements for Aerospace Vehicle Structures,” Department of Defense, United States of America, Washington, D.C.

-“Engineering Data For Aluminum Structures,” the Aluminum Association Incorporated, 900 19th St., N.W., Washington, D.C. 20006.

-“Metals Handbook,” American Society for Metals, Metals Park, Ohio.

Plastics:

-“Plastics, Edition 8, Thermoplastics and Thermosets,” D.A.T.A. Inc., A Cordura Company, 9889 Willow Creek Road, P.O. Box 26875, San Diego, CA 92126.

MATERIAL PROPERTIES DISCUSSION

The strength properties of a material are used by Under Pressure to predict the pressure (depth) at which material failure will occur. Bearing stresses (average seat stresses) and membrane stresses (axial, hoop and meridional stresses in shells and tangential and radial stresses in plates) are compared to uniaxial strengths of the material to predict failure. Shear stresses are compared to shear strengths of the material to predict failure. Uniaxial strengths (Yield Strength for Metals, Ultimate Tensile and Compressive Strengths for Ceramics and Glass, and Ultimate and Working Strengths for Plastics) are provided by the program material database or added by the user to predict material failure due to bearing and membrane stresses. Under Pressure predicts material failure due to shear stresses based on the criteria that the shear strength of the material is equal to 1/2 of its uniaxial strength. This failure criteria for shear is known as the “Maximum Shear Stress Theory.” The type of uniaxial

material strength used by Under Pressure to perform a pressure vessel analysis depends on the behavior of the material.

Main Category materials such as **Ceramics** and **Glass** are characterized by large differences in the magnitudes of the material's tensile strength and compressive strength. Consequently, in evaluating stresses in Ceramic, Glass and similar brittle materials, it is essential to compare tensile stresses to tensile strengths, and to compare compressive stresses to compressive strengths.

Main Category Materials such as **Metals** are typically characterized as ductile materials. Ductile materials are defined by a value of stress (yield strength) at which permanent deformation of the material is initiated. The commencement of permanent deformation (yielding) of a ductile material is generally considered to be the point at which material failure occurs for the purposes of performing structural analysis. The magnitude of uniaxial stress that initiates yield of ductile materials is essentially the same for either a compressive or tensile load.

Main Category Materials such as **Plastics** are typically characterized by material strengths that are heavily dependent upon service temperatures and duration of applied load (creep behavior). For this reason, it is often convenient to define these materials in terms of an Ultimate Strength and a Working Strength when performing structural analysis of plastic materials. The Ultimate Strength of a plastic is the stress required to fail a material for a short-term load applied at room temperatures. The Working Strength of a plastic is the maximum allowable stress selected by the designer to account for the effects of creep behavior or any other factors that could effect the structural performance of the plastic. In general, the material database provided by Under Pressure uses a working strength equal to 1/10 of the ultimate strength. The working strengths given in Under Pressure may be conservative for some applications. It is recommended that the program user consult supplier's technical data sheets for the specific plastic composition of interest. Plastic data sheets will typically recommend working strengths for a material as a function of service conditions. Service conditions are normally defined in terms of maximum design temperature and duration of load.

Analysis of plastic composite materials (e.g. fiberglass tubes) using Under Pressure should be approached with caution. The properties of many composite material are directional such that large variations in strength and modulus exist depend on the orientation with respect to fibers, cloth etc. Analysis of this material directionality along with unique composite failure modes such as delamination are beyond the scope of Under Pressure.

The Young's Modulus (also known as Elastic Modulus) and Poisson's Ratio of a material are known as the material's elastic constants. The elastic constants are used in the evaluation of stresses and deflections of a pressure vessel geometry. Under Pressure also uses the elastic constants to predict the pressure at which thin wall buckling will occur for pressure vessel shell geometries such as tubes, spheres, and hemispheres. Thin wall buckling of tubes, spheres, and hemispheres is dependent on the material's elastic constants and geometry (i.e. the material and geometric "stiffness") and is independent of the material's strength. The density of a material is used by Under Pressure to calculate the in-air weight and water weight (weight when submerged) of a pressure vessel geometry.

ADDING NEW MATERIALS TO THE DATABASE

[Note: you can not create new main categories.]

In general, Main Category Glass should be used to define glass-like materials, Main Category Ceramics should be used to define ceramic-like materials or brittle fracture types of metal, crystalline, and sapphire, Main Category Metals should be used to define metallic materials, and Main Category Plastics should be used to define thermoplastics and thermosets. This approach is valid as long as the failure criteria that is appropriate for the new material corresponds with the failure criteria that is used by Under Pressure to perform structural analysis for the selected Main Category:

- brittle materials that are characterized by significant differences in the magnitudes of their compressive strength and tensile strength should be defined using Main Category **Glass** or **Ceramics**.
- ductile materials that are characterized by yielding (permanent deformation) prior to ultimate failure should be defined using Main Category **Metals**.
- materials that are characterized by strengths that are heavily dependent upon duration of applied load, and service temperature should be defined using Main Category **Plastics** .

The reason for reinforcing the above discussion is that it is possible, for example, to have a metallic material that exhibits brittle behavior, that may be better defined using Main Category Glass or Ceramics for the purposes of performing structural analysis within Under Pressure.

SELECTING A PRESSURE VESSEL MATERIAL

-Click on **CHOOSE** in the Application Window to open the Material Database Dialog Box (alternatively the Material Database Dialog Box can be accessed by clicking on **M**aterials in the Application Window and clicking on **V**iew Material or by entering Alt+M+V from the keyboard.

-Click on appropriate **Main Category** of desired material by clicking on the scroll arrow and clicking on choice.

-Click on appropriate **Sub-Category** of desired material by clicking on the scroll arrow and clicking on choice.

-Click on **Name** of desired material by clicking on the scroll arrow and clicking on choice.

-Click on **Done**.

VIEWING ALL MATERIALS IN THE DATABASE AT ONCE

-Click on **CHOOSE** in the Application Window to open the Material Database Dialog Box.

-Click on All for **Main Category**.

-Click on All for **Sub-Category**.

-Click on the scroll arrow adjacent to the **Name** box to scroll through all materials (listed in alphabetic order) in the database.

EDITING THE PROPERTIES OF AN EXISTING MATERIAL

- Click on **CHOOSE** in the Application Window to open the Material Database Dialog Box.
- Click on appropriate **Main Category** of the material to be edited.
- Click on appropriate **Sub-Category** of the material to be edited.
- Click on **Name** of the material to be edited.
- Click on **Edit Record**.
- Click on the item to be edited (material property value, material property units, comments) and edit item as desired.
- Click on **Done**.

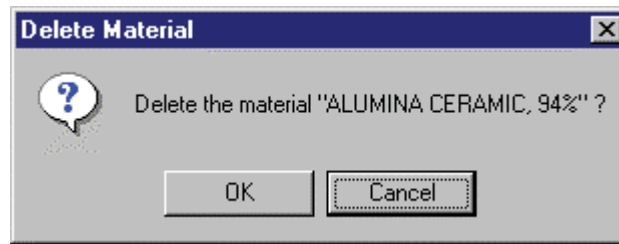
NOTE: The Under Pressure material database resides in the files DESMAT.LDB and DESMAT.MDB. Any edits performed by the user modify these two files. It is recommended that the original material database be backed up prior to editing the material database. An update or reinstallation of Under Pressure will overwrite these files. If you have added material to the database, back up these files **before** reinstalling Under Pressure.

ADDING A NEW MATERIAL

- Click on **CHOOSE** in the Application Window to open the Material Database Dialog Box.
- Click on **ADD**.
- Click on appropriate **Main Category** of the material to be added.
- Click on appropriate **Sub-Category** of the material to be added, **or** use the cursor and keyboard to enter a new user-defined Sub-Category.
- Use the cursor and keyboard to fill in **Name** of new material, material properties, material property units, and comments.
- Click on **DONE**.

DELETING AN EXISTING MATERIAL

- Click on **CHOOSE** in the Application Window to open the Material Database Dialog Box.
- Click on appropriate **Main Category** of the material to be deleted.
- Click on appropriate **Sub-Category** of the material to be deleted.
- Click on **Name** of the material to be deleted.
- Click on **Delete** to open the Delete Material List Box.



Delete Material List Box

-Click on **OK**.

CLOSING THE MATERIAL DATABASE DIALOG BOX

To close the Material Database Dialog Box without completing any operations, select **CANCEL** or press the Escape key from the keyboard.

UNITS FOR MATERIAL PROPERTIES

Materials in the database provided with Under Pressure are defined using English Units. When editing or adding materials to the database, Under Pressure allows the user to use the following units:

Strengths, Young's Modulus:

- Ksi
- psi
- Mpsi
- Kbar
- MPa
- GPa
- Mbar

Density:

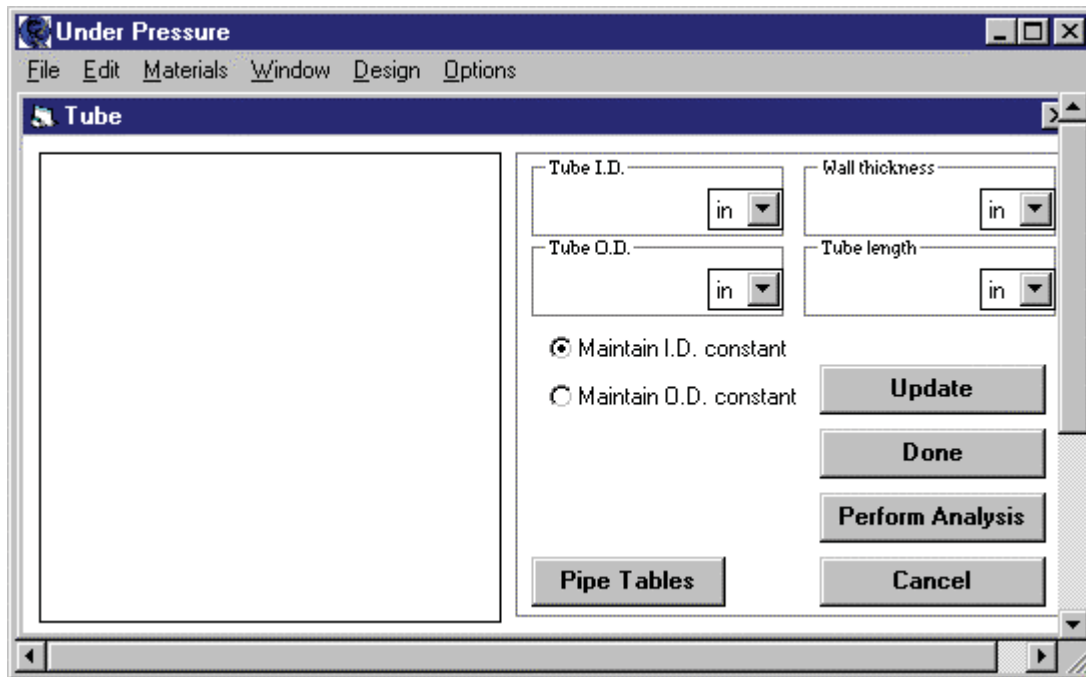
- lb/in³
- lb/ft³
- kg/m³
- gr/cm³
- kg/cm³

DETAILS ON ANALYSIS TYPE

TUBE ANALYSIS

Tube Geometry:

-By selecting Analysis Type **Tube** in the Application Window and clicking on **Enter Geometry**, the program user can access the Tube Geometry Dialog Box:



TUBE GEOMETRY DIALOG BOX

-Tube geometry is defined by the variables **Tube I.D.** (inner diameter), **Tube O.D.** (outer diameter) and **Tube length**. Alternatively, the variable **Wall thickness** can be used in conjunction with either Tube I.D. or Tube O.D. to define the tube geometry. APPENDIX A: PRESSURE VESSEL GEOMETRIES shows a figure of a tube and the variables used to define its geometry.

-While analyzing a tube design, the option buttons **Maintain I.D. constant** and **Maintain O.D. constant** can be used to constrain either the Tube I.D. or the Tube O.D. for successive iterations of a tube design.

-Clicking on **Pipe Tables** allows the user to load standard pipe cross section geometries for analysis.

-After using the cursor and keyboard or Pipe Tables to define a tube geometry, a three dimensional view of the resulting tube geometry is generated on the left hand side of the Tube Geometry Dialog Box.

-Clicking on **Done** saves the tube geometry and closes the Tube Geometry Dialog Box.

-Clicking on **Cancel** closes the Tube Geometry Dialog Box without saving user input.

-Clicking on **Perform Analysis** analyzes the tube geometry and opens the Tube Analysis Dialog Box.

Tube Formulas:

-Formulas used for Tube stress analysis are presented in APPENDIX C: FORMULAS USED BY UNDER PRESSURE.

-If the ratio of the tube mean radius/tube wall thickness is greater than 10, the program uses thin wall formulas to calculate stresses. If the ratio of the tube mean radius/tube wall thickness is less than or equal to 10, the program uses thick wall formulas to calculate stresses.

-Thick wall formulas can be used for all ratios of tube mean radius/tube wall thickness by the program user by clicking on the **Force Thick Wall Equations** check box in the Application Window.

-If thin wall equations are used for an analysis, the membrane stresses (axial and hoop stress) are assumed to be uniform (constant magnitude) throughout the shell wall thickness for either external or internal pressure loading.

-If thin wall equations are used for an analysis, the displacement of the Tube I.D. and the Tube O.D. are assumed to be equal during pressure loading.

-If thick wall equations are used for an analysis, the axial stress is constant throughout the shell wall thickness for either external or internal pressure loading.

-If thick wall equations are used for an analysis, the hoop stress varies throughout the shell wall thickness, with the maximum magnitude of stress occurring at the Tube I.D. for either external or internal pressure loading.

-If thick wall equations are used for an analysis, the displacement of the Tube I.D. and the Tube O.D. differ from one another during pressure loading.

-Formulas used for Tube buckling analysis are presented in APPENDIX C: FORMULAS USED BY UNDER PRESSURE.

-Number of nodes for a Tube buckling analysis refers to the number of circumferential lobes that develop around the tube in its buckled configuration.

Tube Formula Assumptions:

-Stress Analysis: Stress analysis results for a tube analysis assume that the ends of the tube are capped (closed by endcaps). The membrane (axial and hoop) stress results that are presented for a tube analysis are only valid for locations in the tube away from the tube/endcap interface. An analysis of the stresses in the tube at the tube/endcap interface is beyond the scope of Under Pressure.

-Buckling Analysis: Buckling analysis results presented for a tube analysis assume that the ends of the tube are capped (closed by endcaps) and that the endcaps hold the ends of the tube circular. The

validity of this assumption is dependent upon the type of endcap used and the amount of support it provides the ends of the tube. For example, this assumption would be better approximated by a relatively rigid thick flat circular endcap than by a relatively compliant thin walled hemispherical endcap. Buckling of tubes is dependent on the tube material's elastic constants and geometry and is independent of the tube material's strength. As discussed in Example 1 of this manual, out of roundness of tubes can impact buckling resistance. If the ratio of the tube mean radius/tube wall thickness is less than or equal to 10, the results of the buckling analysis may not be valid.

Tube Analysis Results:

-The following Tube Analysis Dialog Box (from Example 1 of this manual) will be used to highlight the data and options available to the user after a tube analysis has been performed:

The screenshot shows the 'Under Pressure' software interface with the 'Tube Analysis - External Pressure' dialog box open. The 'Pressure Range' is set to '1000 - 20000 psi'. Under 'Theoretical Failure', 'Shell failure at 9040.1 psi (Thick wall eq's)' is selected. The material is 'ALUMINUM, 7075-T6' with a 'Yield Strength' of '62 Ksi'. The weight is '23.157' (Air Wt) and '-10.470' (Water Wt). The units are 'psi' for pressure, 'Ft (sea)' for depth, 'psi' for stress, 'Inches' for linear deflection, and 'Radians' for angle deflection.

| Pressure psi | Depth Ft (sea) | Max Axial Stress, psi | Max Hoop Stress, psi | Max Equiv Stress, psi |
|-----------------|-------------------|--------------------------|-------------------------|--------------------------|
| 3000.0 | 6726.2 | -11879 | -23758 | 20575 |
| 4000.0 | 8954.8 | -15839 | -31677 | 27433 |
| 5000.0 | 11177 | -19798 | -39597 | 34292 |
| 6000.0 | 13392 | -23758 | -47516 | 41150 |
| 7000.0 | 15601 | -27718 | -55435 | 48008 |
| 8000.0 | 17804 | -31677 | -63354 | 54867 |
| 9000.0 | 20000 | -35637 | -71274 | 61725 |
| 9040.1 (fail) | 20088 | -35796 | -71591 | 62000 |
| 10000 (fail) | 22190 | -39597 | -79193 | 68583 |

TUBE ANALYSIS DIALOG BOX EXAMPLE

-Clicking on the scroll arrow adjacent to the two boxes titled **Distortion Pressure Range** (at the top of Tube Analysis Dialog Box) allows the user to select the range of pressure (in units of psi or Bar) that will be used with the data in the results table below. For example if psi is selected as the pressure units of choice, one of the following pressure ranges can be selected for presenting the results:

- 0.1-2 psi
- 1-20 psi
- 10-200 psi
- 1000-20,000 psi
- 10,000-200,000 psi

-The **Theoretical Failure** portion indicates the pressure or depth at which the two modes of failure for a tube analysis, **Thin Wall Buckling** and **Shell Failure** occur. The mode of failure that occurs at the lowest (least in magnitude) pressure is highlighted with the aid of an option button. Since failure by buckling requires at least some compressive loading, this failure mode will be non-applicable (N/A) for an internal pressure analysis of a tube design. Clicking on the scroll arrow adjacent to the **Fail value units** box allows the user to select the units that are used for presenting the Theoretical failure values:

- Ksi
- Kbar
- psi
- Bar
- Ft (sea)
- Ft (fresh)
- m (sea)
- m (fresh)

-The upper right hand portion of the Tube Analysis Dialog Box provides weights in air and water (**Air Wt** and **Water Wt**) with the option of expressing these values in either lb. or kg. The weight in water (when completely submerged) assumes that the internal volume of the cylinder is empty (i.e. the cylinder is sealed at both ends with some type of endcap).

-The bottom portion of the Tube Analysis Dialog Box provides a results table of applicable stresses and deflections for a tube analysis that are displayed as a function of the user selected **Distortion Pressure Range**. The following information is provided for a tube analysis as a function of the selected pressure range:

Depth (equivalent water Depth of selected pressure)

Max Axial Stress (Maximum Stress in tube wall in direction of tube centerline axis)

Max Hoop Stress (Maximum Stress in tube wall in direction normal to tube cross section)

Max Equiv Stress (Maximum uniaxial stress that is equivalent to the three dimensional stress state that exists in the tube wall for predicting failure of ductile materials by comparison to the uniaxial yield strength)

dID (displacement of tube Inner Diameter)

dOD (displacement of tube Outer Diameter)

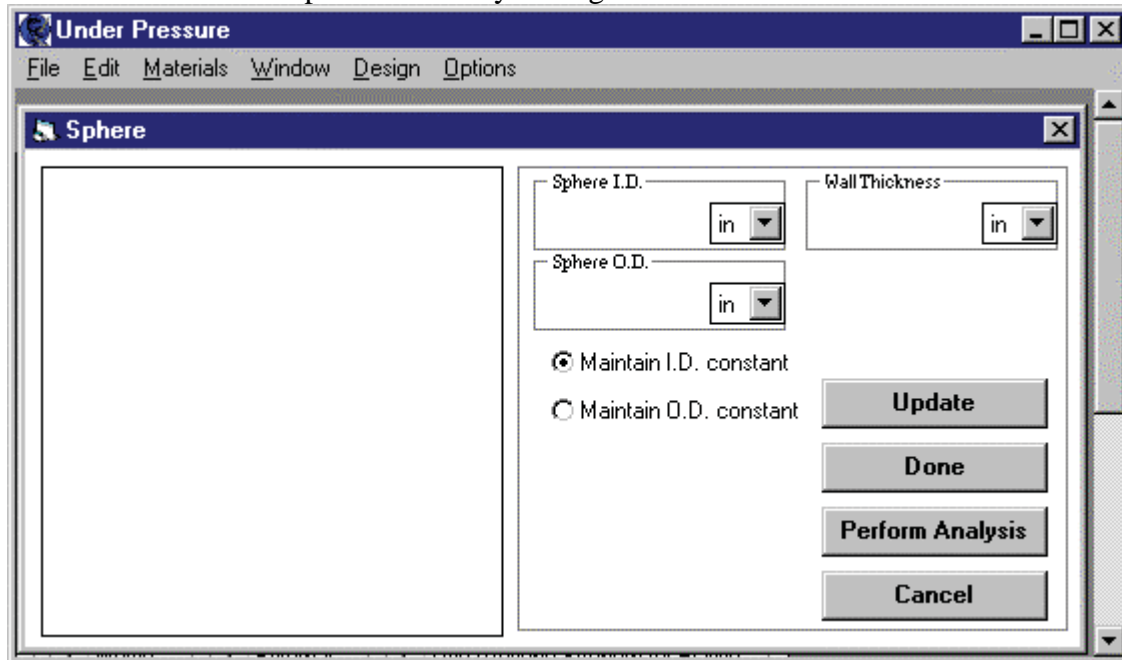
dLength (Change in tube length)

An explanation of the orientation of the above stresses is provided in Appendix D: PRESSURE VESSEL STRESSES. Again, the user has an option as to which units will be used to display the data in the results table. Rows of data for an applied pressures at or greater than the calculated failure pressure are highlighted by the program as a warning to the user. As discussed in Appendix C: FORMULAS USED BY UNDER PRESSURE, Maximum Equivalent Stress is used as a failure criteria for Metal Tubes, and is not applicable to tubes fabricated from other Main Category Materials (Ceramics, Glass, and Plastics).

SPHERE ANALYSIS:

Sphere Geometry:

-By selecting Analysis Type **Sphere** in the Application Window and clicking on **Enter Geometry**, the program user can access the Sphere Geometry Dialog Box:



SPHERE GEOMETRY DIALOG BOX

-Sphere geometry is defined by the variables **Sphere I.D.** (inner diameter) and **Sphere O.D.** (outer diameter). Alternatively, the variable **Wall thickness** can be used in conjunction with either Sphere I.D. or Sphere O.D. to define the sphere geometry. APPENDIX A: PRESSURE VESSEL GEOMETRIES shows a figure of a sphere and the variables used to define its geometry.

-While analyzing a sphere design, the option buttons **Maintain I.D. constant** and **Maintain O.D. constant** can be used to constrain either the Sphere I.D. or the Sphere O.D. for successive iterations of a sphere design.

-After using the cursor and keyboard to define a sphere geometry, a three dimensional view of the resulting sphere geometry is generated on the left hand side of the Sphere Geometry Dialog Box.

-Clicking on **Done** saves the sphere geometry and closes the Sphere Geometry Dialog Box.

-Clicking on **Cancel** closes the Sphere Geometry Dialog Box without saving user input.

-Clicking on **Perform Analysis** analyzes the Sphere geometry and opens the Sphere Analysis Dialog Box.

Sphere Formulas:

-Formulas used for Sphere stress analysis are presented in APPENDIX C: FORMULAS USED BY UNDER PRESSURE.

-If the ratio of the sphere mean radius/sphere wall thickness is greater than 10, the program uses thin wall formulas to calculate stresses. If the ratio of the sphere mean radius/sphere wall thickness is less than or equal to 10, the program uses thick wall formulas to calculate stresses.

-Thick wall formulas can be used for all ratios of sphere mean radius/tube wall thickness by the program user by clicking on the **Force Thick Wall Equations** check box in the Application Window.

-If thin wall equations are used for an analysis, the membrane stresses (meridional and hoop stress) are assumed to be uniform (constant magnitude) throughout the shell wall thickness for either external or internal pressure loading.

-If thin wall equations are used for an analysis, the displacement of the Sphere I.D. and the Sphere O.D. are assumed to be equal during pressure loading.

-If thick wall equations are used for an analysis, the meridional and hoop stresses vary throughout the shell wall thickness, with the maximum magnitude of stress occurring at the Sphere I.D. for either external or internal pressure loading.

-If thick wall equations are used for an analysis, the displacement of the Sphere I.D. and the Sphere O.D. differ from one another during pressure loading.

-Shear stresses in the sphere wall are calculated when thick wall equations are used and the load case is for internal pressure loading.

-Formulas used for Sphere buckling analysis are presented in APPENDIX C: FORMULAS USED BY UNDER PRESSURE.

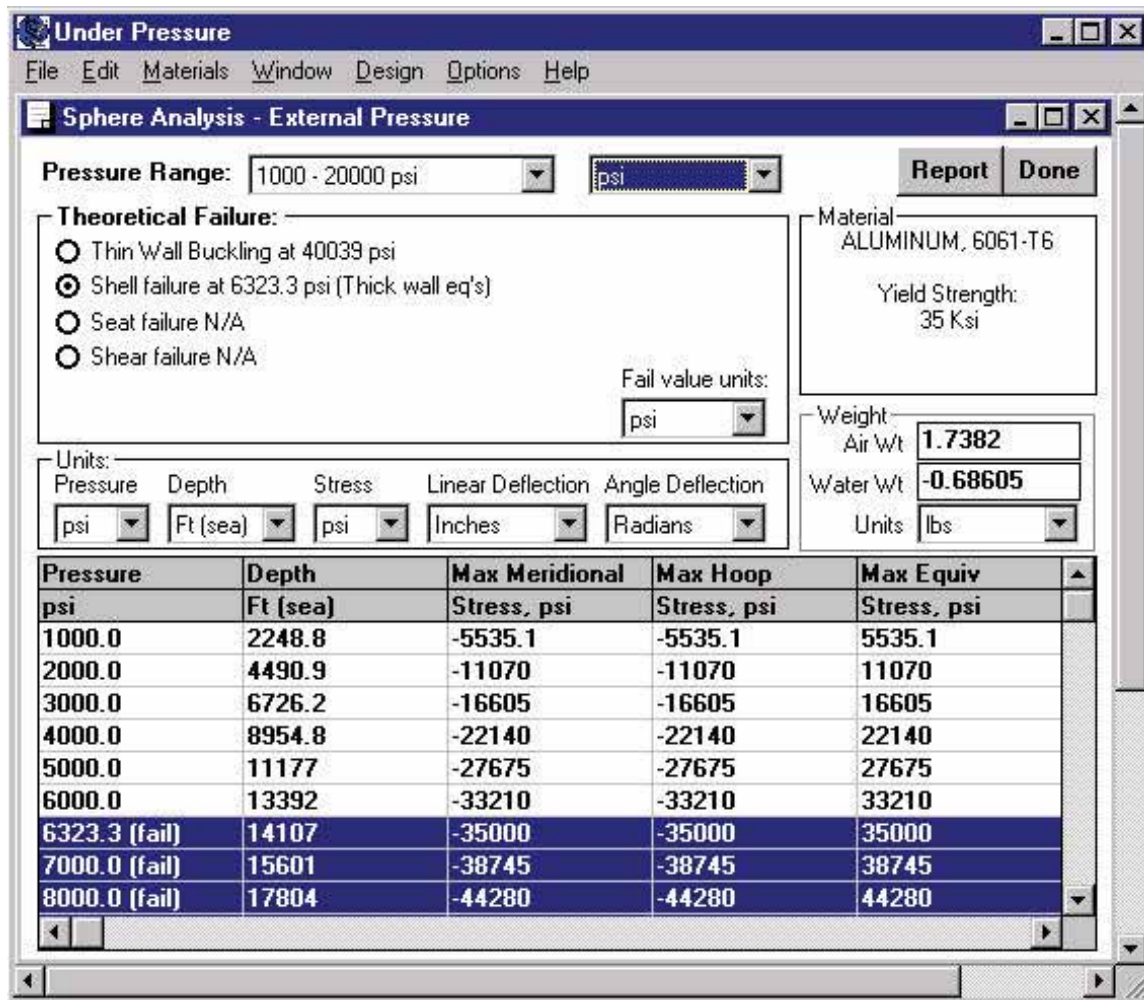
-Spheres buckle by dimpling of the shell wall and therefore the number of nodes generated for a tube buckling analysis are not presented.

Sphere Formula Assumptions:

-Buckling Analysis: If the ratio of the sphere mean radius/sphere wall thickness is less than or equal to 10, the results of the buckling analysis may not be valid. Buckling of spheres is dependent on the sphere material's elastic constants and geometry and is independent of the sphere material's strength. As discussed in Example 1 of this manual, variations in thickness and "flat spots" can impact buckling resistance of spherical shells.

Sphere Analysis Results:

-The following Sphere Analysis Dialog Box will be used to highlight the data and options available to the user after a sphere analysis has been performed:



SPHERE ANALYSIS DIALOG BOX EXAMPLE

-Clicking on the scroll arrow adjacent to the two boxes titled **Distortion Pressure Range** (at the top of Sphere Analysis Dialog Box) allows the user to select the range of pressure (in units of psi or Bar) that will be used with the data in the results table below. For example if psi is selected as the pressure units of choice, one of the following pressure ranges can be selected for presenting the results:

- 0.1-2 psi
- 1-20 psi
- 10-200 psi
- 1000-20,000 psi
- 10,000-200,000 psi

-The **Theoretical Failure** portion indicates the pressure or depth at which the three modes of failure for a sphere analysis, **Thin Wall Buckling**, **Shell Failure** and **Shear Failure** occur. The mode of failure that occurs at the lowest (least in magnitude) pressure is highlighted with the aid of an option button. Since failure by buckling requires at least some compressive loading, this failure mode will be non-applicable (N/A) for an internal pressure analysis of a sphere design. Clicking on the scroll arrow adjacent to the **Fail value units** box allows the user to select the units for at which failure will be presented:

- Ksi

- Kbar
- psi
- Bar
- Ft (sea)
- Ft (fresh)
- m (sea)
- m (fresh)

-The upper right hand portion of the Sphere Analysis Dialog Box provides weights in air and water (**Air Wt** and **Water Wt**) with the option of expressing these values in either lb. or kg. The weight in water (when completely submerged) assumes that the internal volume of the sphere is empty.

-The bottom portion of the Sphere Analysis Dialog Box provides a results table of applicable stresses and deflections for a sphere analysis that are displayed as a function of the user selected **Distortion Pressure Range**. The following information is provided for a sphere analysis as a function of the selected pressure range:

Depth (equivalent water Depth of selected pressure)

Max Meridional Stress (Maximum Stress in sphere wall in direction of sphere cross section meridian)

Max Hoop Stress (Maximum Stress in sphere wall in direction normal to sphere cross section)

Max Equiv Stress (Maximum uniaxial stress that is equivalent to the three dimensional stress state that exists in the sphere wall for predicting failure of ductile materials by comparison to the uniaxial yield strength)

Max Shear Stress (Maximum Shear Stress in sphere wall, N/A for external pressure)

Avg Seat Stress (Average Seat Stress, N/A for spheres)

dID (displacement of sphere Inner Diameter)

dOD (displacement of sphere Outer Diameter)

An explanation of the orientation of the above stresses is provided in Appendix D: PRESSURE VESSEL STRESSES. Again, the user has an option as to which units will be used to display the data in the results table. Rows of data for an applied pressures at or greater than the calculated failure pressure are highlighted by the program to warn the user. As discussed in Appendix C: FORMULAS USED BY UNDER PRESSURE, Maximum Equivalent Stress is used as a failure criteria for Metal Spheres, and is not applicable to spheres fabricated from other Main Category Materials Ceramics, Glass, and Plastics.

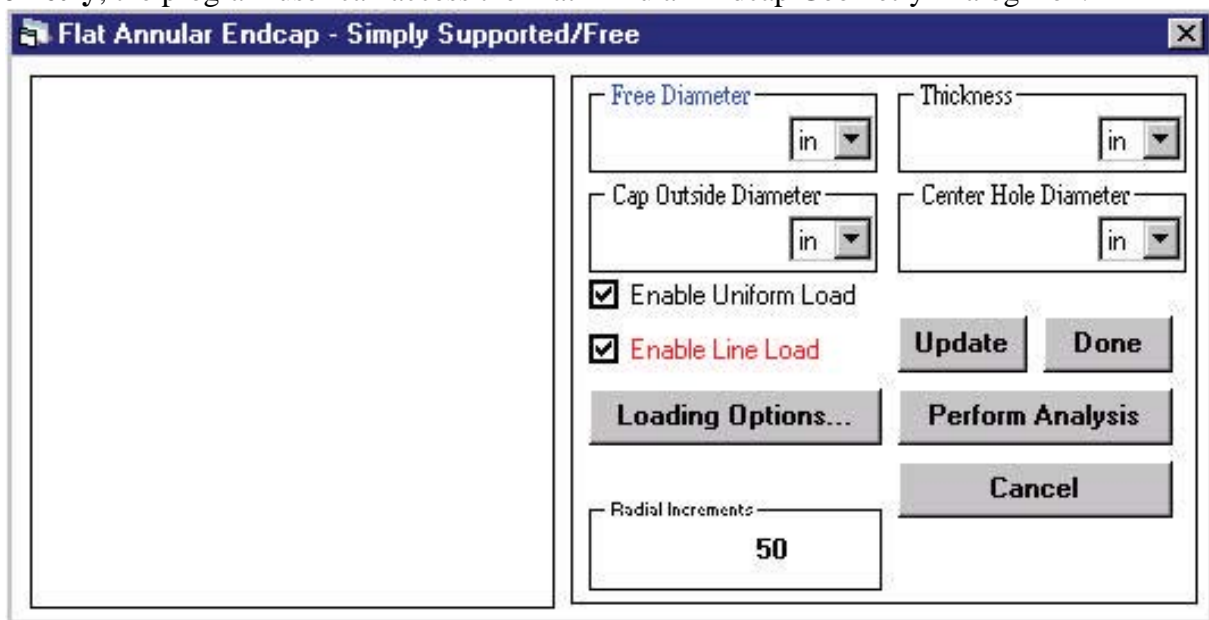
FLAT ANNULAR ENDCAP ANALYSIS

Flat Annular Endcap Edge Restraint Options:

-The significance of **Edge Restraint Options - Outer/Inner** for Flat Annular Plates is defined in APPENDIX B: FLAT ENDCAP BOUNDARY CONDITIONS. If in doubt about which edge restraint is most appropriate for a given Flat Annular Endcap design, the use of the edge restraint option that results in the highest stresses in the plate would be the conservative approach for the designer to use.

Flat Annular Endcap Geometry:

-By selecting Analysis Type **Endcap Only** in the Application Window, selecting **Endcap Configuration** Flat Annular, selecting appropriate **Edge Restraint Options** and clicking on **Enter Geometry**, the program user can access the Flat Annular Endcap Geometry Dialog Box:



FLAT ANNULAR ENDCAP GEOMETRY DIALOG BOX

-Flat Annular Endcap geometry is defined by the variables **Center Hole Diameter**, **Cap Outside Diameter**, **Free Diameter** and **Thickness**. APPENDIX A: PRESSURE VESSEL GEOMETRIES shows a figure of a flat annular endcap and the variables used to define its geometry.

-After using the cursor and keyboard to define the endcap geometry, a three dimensional view of the resulting flat annular endcap geometry is generated (using an additional line to denote the Plate Free Diameter) on the left hand side of the Flat Annular Endcap Geometry Dialog Box.

-Clicking on **Enable Uniform Load** applies a uniform pressure to the appropriate flat surface (external surface for external pressure, internal surface for internal pressure) of the annular plate. Clicking on **Enable Line Load** applies a line load to the circular edge of the center hole that is equivalent to the pressure load that exists on any insert mounted in the hole. $(\text{Line load}) \times (\text{circumference of hole}) = (\text{pressure load}) \times (\text{cross sectional area of hole})$. By selecting **Enable**

Uniform Load, a uniformly distributed pressure is applied to the flat annular surface of the plate from the center hole diameter to the cap outside diameter. By selecting **Enable Line Load**, a uniform circular line load is applied to the flat annular surface of the plate at a diameter equal to the hole diameter. A typical analysis of a flat annular endcap for pressure loading should be performed with BOTH the **Enable Uniform Load** and **Enable Line Load** check boxes active at the same time.

Under Pressure allows further options for applying loads to flat annular endcaps by clicking on **Loading Options**. Selecting **Loading Options** allows the user to increase the inner diameter of the annular area to which uniform pressure loading is applied and/or allows the user to increase the diameter at which a uniform circular line load is applied. These loading diameters can be adjusted by clicking on Loading Options, clicking the Custom Option Button, and using the cursor and keyboard to input values for Uniform Loading I.D., Line Load Application Diameter and Line Loading O.D.. If the **Loading Options** is not used, the inner diameter of the uniform load and the application diameter of the line load default to the plate's center hole diameter.

Using a Uniform Loading I.D., Line Load Application Diameter and Line Loading O.D. larger than the hole diameter may be appropriate for endcap designs where some type of circular cover or connector is installed over the center hole that seals a circular area on the flat plate that is larger in diameter than the center hole diameter. For this situation, the Uniform Loading I.D., Line Load Application Diameter and Line Loading O.D. could be adjusted to equal the diameter of the endcap flat surface that is sealed from pressure by the cover or connector.

-Clicking on **Done** saves the geometry and closes the Flat Annular Endcap Geometry Dialog Box.

-Clicking on **Cancel** closes the Flat Annular Endcap Geometry Dialog Box without saving user input.

-Clicking on **Perform Analysis** analyzes the Flat Annular Endcap geometry and opens the Flat Annular Endcap Analysis Dialog Box.

Flat Annular Endcap Formulas:

-Formulas used for Annular Circular Endcap stress analysis are presented in APPENDIX C: FORMULAS USED BY UNDER PRESSURE.

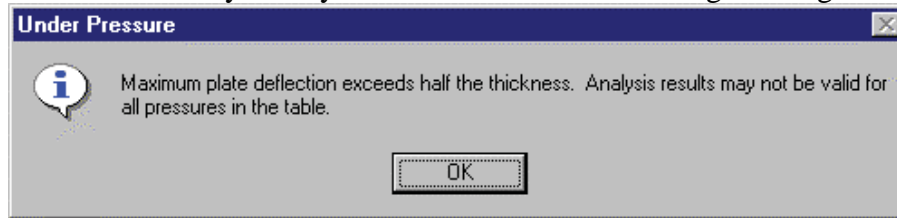
Flat Annular Endcap Formula Assumptions:

-Stress analysis formulas assume that the plate is flat and has a constant thickness

-The ratio of the Plate Free Diameter to the Plate Thickness is greater than or equal to 4. If this assumption is violated, the results of the analysis may not be valid and the following warning will appear:

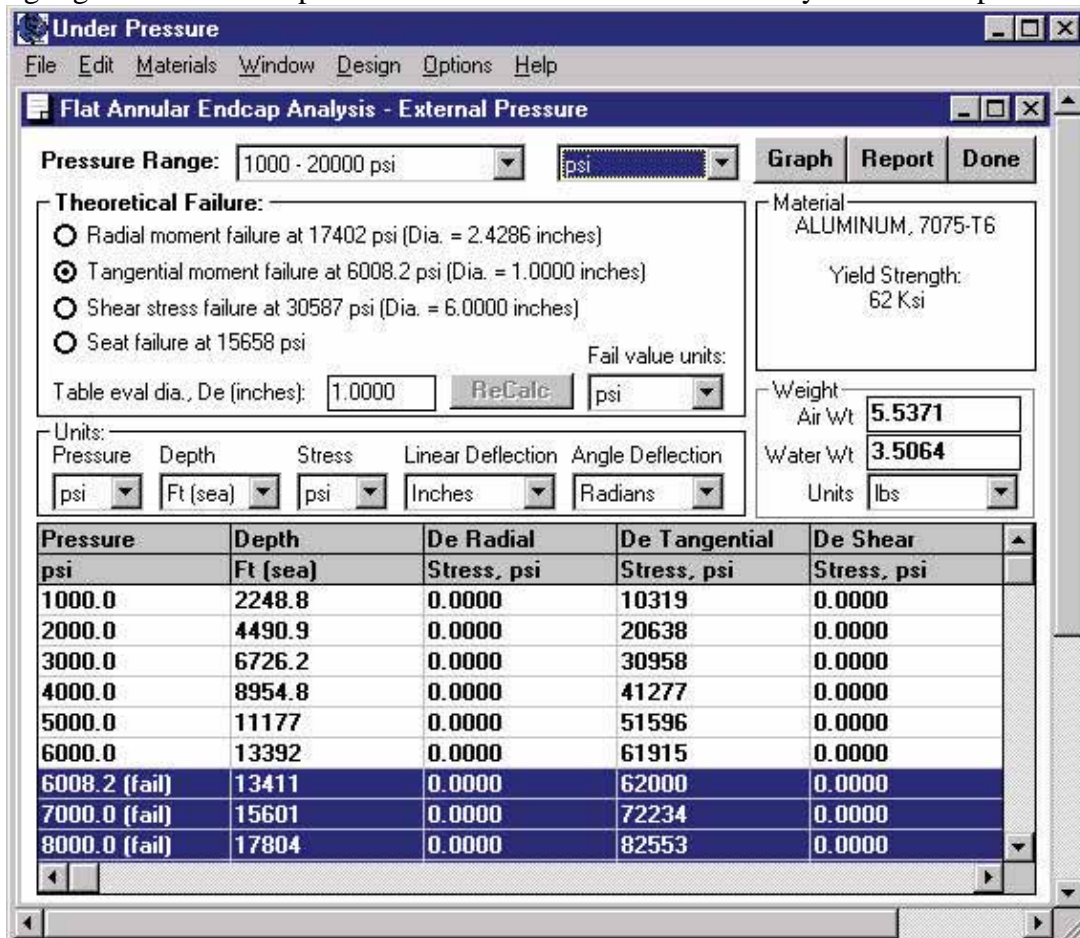


-The maximum deflection of the plate does not exceed one half the plate thickness. If this assumption is violated, the results of the analysis may not be valid and the following warning will appear:



Flat Annular Endcap Analysis Results:

-The following Flat Annular Endcap Analysis Dialog Box (from Example 3 of this manual) will be used to highlight the data and options available to the user after an analysis has been performed:



FLAT ANNULAR ENDCAP ANALYSIS DIALOG BOX EXAMPLE

-Clicking on the scroll arrow adjacent to the two boxes titled **Distortion Pressure Range** (at the top of Flat Annular Endcap Analysis Dialog Box) allows the user to select the range of pressure (in units of psi or Bar) that will be used with the data in the results table below. For example if psi is selected as the pressure units of choice, one of the following pressure ranges can be selected for presenting the results:

- 0.1-2 psi
- 1-20 psi
- 10-200 psi

- 100-2000 psi
- 1000-20,000 psi
- 10,000-200,000 psi

-The **Theoretical Failure** portion indicates the pressure or depth and radial location at which the four modes of failure for a Flat Annular Endcap analysis, **Radial moment fail (Stress) Failure**, **Tangential moment fail (Stress) Failure**, **Shear Stress Failure** and **Seat Fail Pressure** occur. The mode of failure that occurs at the lowest (least in magnitude) pressure is highlighted with the aid of an option button. Clicking on the scroll arrow adjacent to the **Fail value units** box allows the user to select the units for at which failure will be presented:

- Ksi
- Kbar
- psi
- Bar
- Ft (sea)
- Ft (fresh)
- m (sea)
- m (fresh)

The bottom of the Theoretical Failure portion of the Flat Annular Endcap Analysis Dialog Box provides the radial location of the Plate (i.e. the diameter of evaluation = D_e) for which the maximum radial and tangential stresses are presented in the results table. The user can change the radial location used in the results table using the box adjacent to **Table eval dia, D_e** and clicking on **ReCalc**.

-The upper right hand portion of the Flat Annular Endcap Analysis Dialog Box provides weights in air and water (**Air Wt** and **Water Wt**) with the option of expressing these values in either lbs or kg.

-The bottom portion of the Flat Annular Endcap Analysis Dialog Box provides a results table of applicable stresses and deflections for an analysis that are displayed as a function of the user selected **Distortion Pressure Range**. The following information is provided for an analysis as a function of the selected pressure range:

Depth (equivalent water Depth of selected pressure)

D_e Radial Stress (maximum Stress in plate cross section in direction normal to plate center line at user selected evaluation Diameter)

D_e Tangential Stress (maximum Stress in plate in direction normal to plate cross section at user selected evaluation Diameter)

D_e Shear Stress (Shear Stress in plate at user selected evaluation Diameter)

D_e Vert Defl (Deflection of plate in direction of plate centerline at user selected evaluation Diameter)

D_e Angular (Angular rotation of plate cross section at user selected evaluation Diameter)

HD Radial Stress (maximum Hole Diameter Radial Stress)

HD Tangential Stress (maximum Hole Diameter Tangential Stress)

HD Shear Stress (Hole Diameter Shear Stress)

HD Vert Defl (Hole Diameter Vertical Deflection)

HD Angular (Hole Diameter Angular rotation)

FD Radial Stress (maximum Free Diameter Radial Stress)

FD Tangential Stress (maximum Free Diameter Tangential Stress)
FD Shear Stress (Free Diameter Shear Stress)
FD Vert Defl (Free Diameter Vertical Deflection)
FD Angular (Free Diameter Angular rotation)
Avg Seat Stress (Average bearing Stress on annular area of plate between plate outside diameter and plate free diameter)

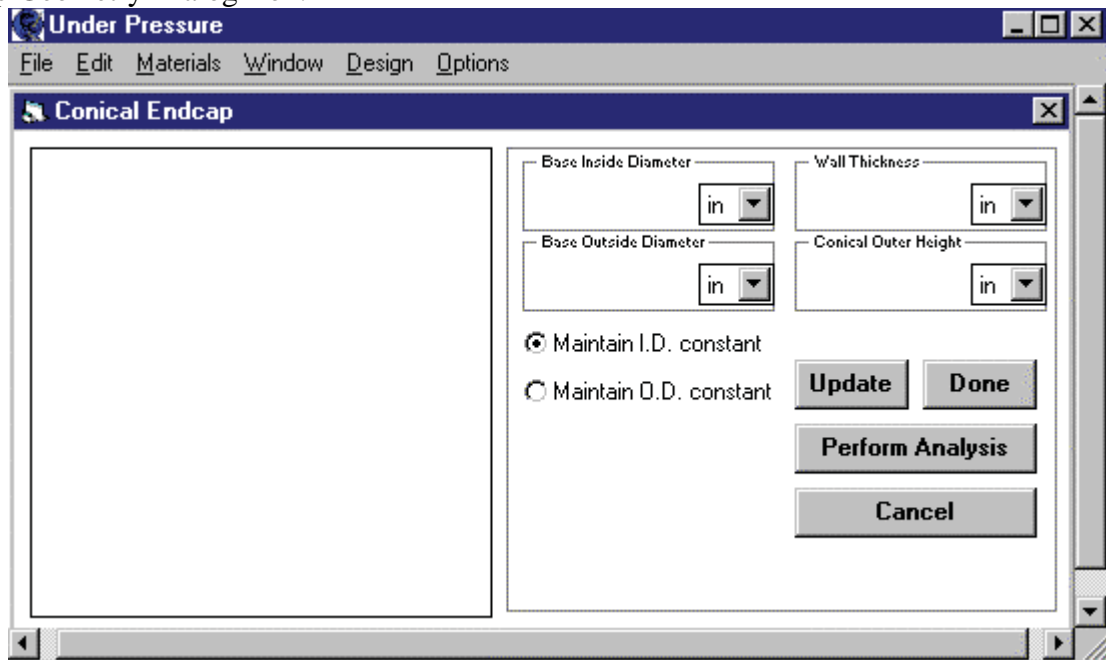
An explanation of the orientation of the above stresses is provided in Appendix D: PRESSURE VESSEL STRESSES. Again, the user has an option as to which units will be used to display the data in the results table. Rows of data for an applied pressures at or greater than the calculated failure pressure are highlighted by the program to warn the user. As discussed in Appendix C: FORMULAS USED BY UNDER PRESSURE, the maximum radial and tangential membrane stresses occur on the flat surfaces of the flat annular plate with the concave side of the plate loaded in compression and convex side of the plate loaded in tension (the radial and tangential membrane stresses are zero at the plate midthickness).

-Selecting **GRAPH** generates a plot of the radial and tangential stresses in the plate for a 1 Ksi external pressure load as a function of the location along the plate diameter. These graphs aid the user in visualizing the state of stress in the Flat Annular Endcap.

CONICAL ENDCAP ANALYSIS

Conical Endcap Geometry:

-By selecting Analysis Type **Endcap Only** in the Application Window, selecting Endcap Configuration **Conical** and clicking on **Enter Geometry**, the program user can access the Conical Endcap Geometry Dialog Box:



CONICAL ENDCAP GEOMETRY DIALOG BOX

-Conical Endcap geometry is defined by the variables **Conical Inside Diameter** (I.D.), **Conical Outside Diameter** (O.D.) and **Conical Outer Height**. Alternatively, the variable **Wall thickness** can be used in conjunction with either I.D. or O.D. to define the conical endcap geometry. APPENDIX A: PRESSURE VESSEL GEOMETRIES shows a figure of a Conical Endcap and the variables used to define its geometry.

-While analyzing a Conical Endcap design, the option buttons **Maintain I.D. constant** and **Maintain O.D. constant** can be used to constrain either the I.D. or the O.D. for successive iterations of a Conical endcap design analysis.

-After using the cursor and keyboard to define the endcap geometry, a three dimensional view of the resulting conical shell geometry is generated on the left hand side of the Conical Endcap Geometry Dialog Box.

-Clicking on **Done** saves the geometry and closes the Conical Endcap Geometry Dialog Box.

-Clicking on **Cancel** closes the Conical Endcap Geometry Dialog Box without saving user input.

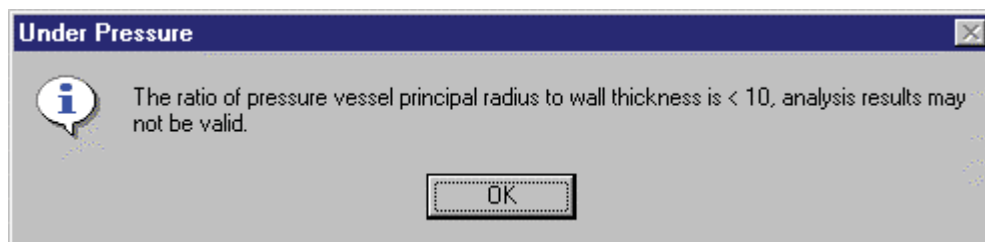
-Clicking on **Perform Analysis** analyzes the Conical Endcap geometry and opens the Conical Endcap Analysis Dialog Box.

Conical Endcap Formulas:

-Formulas used for Conical Endcap stress analysis are presented in APPENDIX C: FORMULAS USED BY UNDER PRESSURE.

Conical Endcap Formula Assumptions:

-If the ratio of the cone open end mean radius (principal radius)/cone wall thickness is less than or equal to 10, the results of the stress analysis may not be valid. If this assumption is violated the following warning will appear:



- Calculation of average seat stress results for a conical endcap assume that the open end of the cone is supported with axial edge support (see conical endcap figure of Appendix D). Calculation of meridional and hoop membrane stress results for a conical endcap assume that the open end of the cone is supported with tangential edge support (see support of conical endcap FEA model, econelhoop.mod, in Appendix E).

Conical Endcap Analysis Results:

-The following Conical Endcap Analysis Dialog Box will be used to highlight the data and options available to the user after an analysis has been performed:

Conical Endcap Analysis - External Pressure

Pressure Range: 1000 - 20000 psi psi

Theoretical Failure:

- Thin Wall Buckling N/A
- Shell failure at 7630.1 psi (Thin wall eq's)
- Seat failure at 17143 psi
- Shear failure N/A

Fail value units: psi

Material: ALUMINUM, 6061-T6
Yield Strength: 35 Ksi

Weight:
Air Wt: 3.1962
Water Wt: 1.2955
Units: lbs

Units:
Pressure: psi Depth: Ft (sea) Stress: psi Linear Deflection: Inches Angle Deflection: Degrees

| Pressure | Depth | Max Meridional | Max Hoop | Max Equiv |
|---------------|----------|----------------|-------------|-------------|
| psi | Ft (sea) | Stress, psi | Stress, psi | Stress, psi |
| 1000.0 | 2248.8 | -2648.4 | -5296.7 | 4587.1 |
| 2000.0 | 4490.9 | -5296.7 | -10593 | 9174.2 |
| 3000.0 | 6726.2 | -7945.1 | -15890 | 13761 |
| 4000.0 | 8954.8 | -10593 | -21187 | 18348 |
| 5000.0 | 11177 | -13242 | -26484 | 22935 |
| 6000.0 | 13392 | -15890 | -31780 | 27523 |
| 7000.0 | 15601 | -18539 | -37077 | 32110 |
| 7630.1 (fail) | 16990 | -20207 | -40415 | 35000 |
| 8000.0 (fail) | 17804 | -21187 | -42374 | 36697 |

CONICAL ENDCAP ANALYSIS DIALOG BOX EXAMPLE

-Clicking on the scroll arrow adjacent to the two boxes titled **Distortion Pressure Range** (at the top of Conical Endcap Analysis Dialog Box) allows the user to select the range of pressure (in units of psi or Bar) that will be used with the data in the results table below. For example if psi is selected as the pressure units of choice, one of the following pressure ranges can be selected for presenting the results:

- 0.1-2 psi
- 1-20 psi
- 10-200 psi
- 100-2000 psi
- 1000-20,000 psi
- 10,000-200,000 psi

-The **Theoretical Failure** portion indicates the pressure or depth at which the two modes of failure for a conical endcap analysis, **Shell Failure**, and **Seat Failure** occur. Thin wall buckling failure is non-applicable (N/A), not because buckling of a conical shell won't occur, but because an explicit buckling formula for conical shells does not exist in the APPENDIX C formula reference. An approximate buckling formula for truncated conical shells with closed ends based on the formulas used for cylindrical shells (tubes) is referenced in APPENDIX C, but is not programmed into Under Pressure. The mode of failure that occurs at the lowest (least in magnitude) pressure is highlighted with the aid of an option button. Clicking on the scroll arrow adjacent to the **Fail value units** box allows the user to select the units for at which failure will be presented:

- Ksi
- Kbar
- psi
- Bar
- Ft (sea)
- Ft (fresh)
- m (sea)
- m (fresh)

-The upper right hand portion of the Conical Endcap Analysis Dialog Box provides weights in air and water (**Air Wt** and **Water Wt**) with the option of expressing these values in either lbs or kg. The weight in water (when completely submerged) assumes that the internal volume of the conical endcap is empty.

-The bottom portion of the Conical Endcap Analysis Dialog Box provides a results table of applicable stresses and deflections for an analysis that are displayed as a function of the user selected **Distortion Pressure Range**. The following information is provided for an analysis as a function of the selected pressure range:

Depth (equivalent water Depth of selected pressure)

Max Meridional Stress (Maximum Stress in cone wall in direction of cone cross section meridian)

Max Hoop Stress (Maximum Stress in cone wall in direction normal to cone cross section)

Max Equiv Stress (Maximum uniaxial stress that is equivalent to the three dimensional stress state that exists in the cone wall for predicting failure of ductile materials by comparison to the uniaxial yield strength)

Avg Seat Stress (Average Stress on flat annular bearing area at cone open end)

dRadius (radial displacement of cone open end)

dHeight (change in cone height)

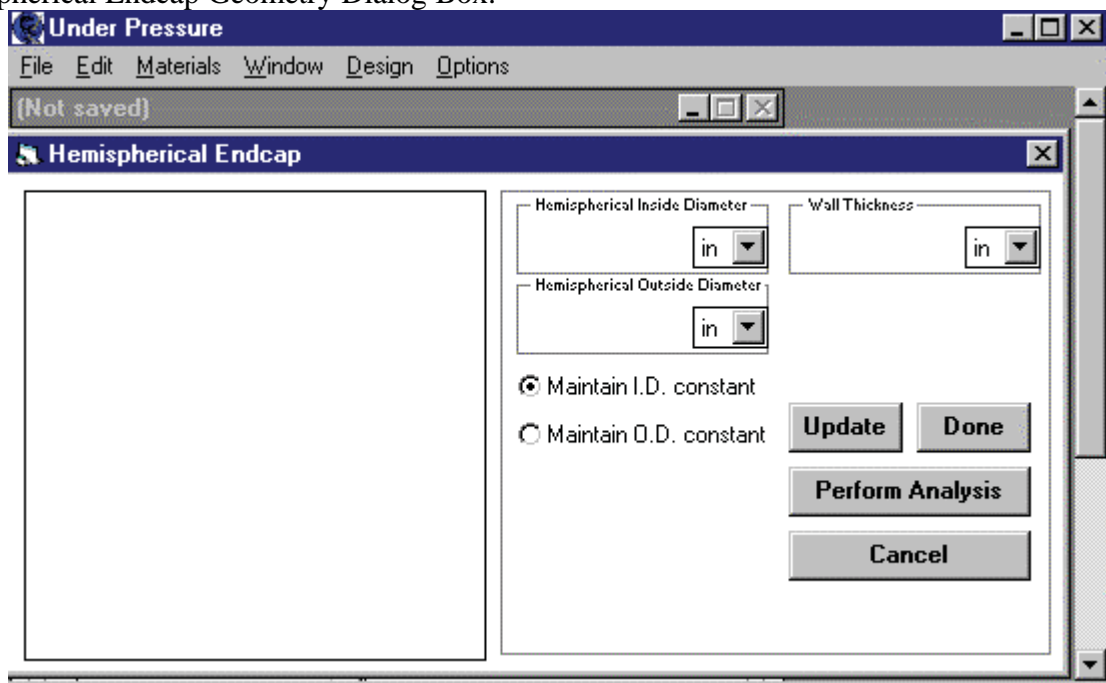
dMeridian (rotation of cone meridian)

An explanation of the orientation of the above stresses is provided in Appendix D: PRESSURE VESSEL STRESSES. Again, the user has an option as to which units will be used to display the data in the results table. Rows of data for an applied pressures at or greater than the calculated failure pressure are highlighted by the program to warn the user. As discussed in Appendix C: FORMULAS USED BY UNDER PRESSURE, Maximum Equivalent Stress is used as a failure criteria for Metal Conical Endcaps, and is not applicable to conical shells fabricated from other Main Category Materials Ceramics, Glass, and Plastics.

HEMISPHERICAL ENDCAP ANALYSIS:

Hemispherical Endcap Geometry:

-By selecting Analysis Type **Endcap Only** in the Application Window, selecting Endcap Configuration **Hemispherical** and clicking on **Enter Geometry**, the program user can access the Hemispherical Endcap Geometry Dialog Box:



HEMISPHERICAL ENDCAP GEOMETRY DIALOG BOX

-Hemispherical Endcap geometry is defined by the variables **Hemispherical Inside Diameter** (I.D.) and **Hemispherical Outside Diameter** (O.D.). Alternatively, the variable **Wall thickness** can be used in conjunction with either I.D. or O.D. to define the hemispherical endcap geometry. APPENDIX A: PRESSURE VESSEL GEOMETRIES shows a figure of a hemispherical endcap and the variables used to define its geometry.

-While analyzing a hemispherical endcap design, the option buttons **Maintain I.D. constant** and **Maintain O.D. constant** can be used to constrain either the I.D. or the O.D. for successive iterations of a hemispherical endcap design.

-After using the cursor and keyboard to define the endcap geometry, a three dimensional view of the resulting hemispherical shell geometry is generated on the left hand side of the Hemispherical Endcap Geometry Dialog Box.

- Clicking on **Done** saves the geometry and closes the Hemispherical Endcap Geometry Dialog Box.
- Clicking on **CANCEL** closes the Hemispherical Endcap Geometry Dialog Box without saving user input.
- Clicking on **Perform Analysis** analyzes the Hemispherical Endcap geometry and opens the Hemispherical Endcap Analysis Dialog Box.

Hemispherical Endcap Formulas:

- Formulas used for Hemispherical Endcap stress analysis are presented in APPENDIX C: FORMULAS USED BY UNDER PRESSURE. These formulas are the same as those used for a Sphere stress analysis as discussed previously.
- If the ratio of the hemisphere mean radius/hemisphere wall thickness is greater than 10, the program uses thin wall formulas to calculate stresses. If the ratio of the hemisphere mean radius/hemisphere wall thickness is less than or equal to 10, the program uses thick wall formulas to calculate stresses.
- Thick wall formulas can be used for all ratios of hemisphere mean radius/tube wall thickness by the program user by clicking on the **Force Thick Wall Equations** check box in the Application Window.
- If thin wall equations are used for an analysis, the membrane stresses (meridional and hoop stress) are assumed to be uniform (constant magnitude) throughout the shell wall thickness for either external or internal pressure loading.
- If thin wall equations are used for an analysis, the displacement of the Hemisphere I.D. and the Hemisphere O.D. are assumed to be equal during pressure loading.
- If thick wall equations are used for an analysis, the meridional and hoop stresses vary throughout the shell wall thickness, with the maximum magnitude of stress occurring at the Hemisphere I.D. for either external or internal pressure loading.
- If thick wall equations are used for an analysis, the displacement of the Hemisphere I.D. and the Hemisphere O.D. differ from one another during pressure loading.
- Shear stresses in the Hemisphere wall are calculated when thick wall equations are used and the load case is for internal pressure loading.
- Formulas used for Hemisphere buckling analysis are presented in APPENDIX C: FORMULAS USED BY UNDER PRESSURE.

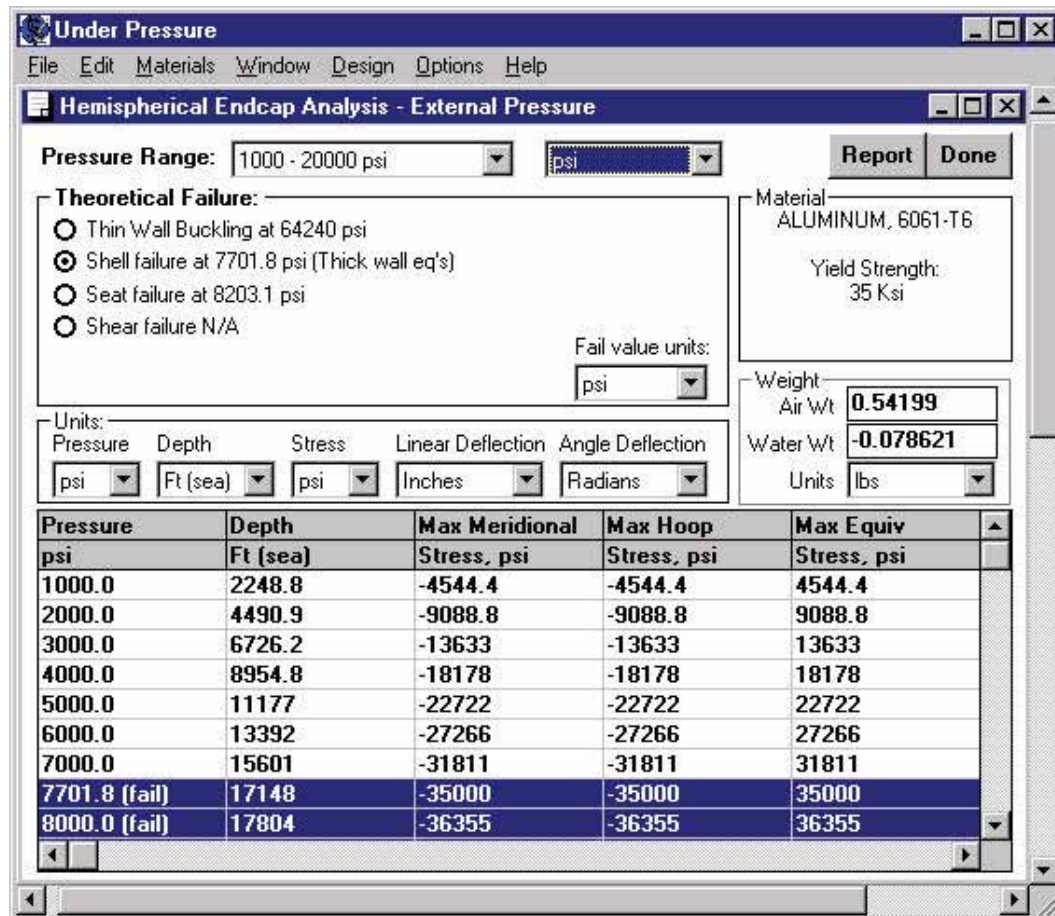
-Hemispherical Endcaps buckle by dimpling of the shell wall and therefore the number of nodes generated for a tube buckling analysis are not presented

Hemispherical Endcap Formula Assumptions:

-Buckling Analysis: If the ratio of the hemisphere mean radius/hemisphere wall thickness is less than or equal to 10, the results of the buckling analysis may not be valid.

Hemispherical Endcap Analysis Results:

-The following Hemispherical Endcap Analysis Dialog Box will be used to highlight the data and options available to the user after an analysis has been performed:



HEMISPHERICAL ENDCAP ANALYSIS DIALOG BOX EXAMPLE

-Clicking on the scroll arrow adjacent to the two boxes titled **Distortion Pressure Range** (at the top of Hemispherical Endcap Analysis Dialog Box) allows the user to select the range of pressure (in units of psi or Bar) that will be used with the data in the results table below. For example if psi is selected as the pressure units of choice, one of the following pressure ranges can be selected for presenting the results:

- 0.1-2 psi
- 1-20 psi
- 10-200 psi
- 100-2000 psi
- 1000-20,000 psi
- 10,000-200,000 psi

-The **Theoretical Failure** portion indicates the pressure or depth at which the four modes of failure for a hemispherical endcap analysis, **Thin Wall Buckling**, **Shell Failure**, **Seat Failure**, and **Shear Failure** occur. The mode of failure that occurs at the lowest (least in magnitude) pressure is highlighted with the aid of an option button. Since failure by buckling requires at least some compressive loading, this failure mode will be non-applicable (N/A) for an internal pressure analysis of a sphere design. Clicking on the scroll arrow adjacent to the **Fail value units** box allows the user to select the units at which failure will be presented:

- Ksi
- Kbar

- psi
- Bar
- Ft (sea)
- Ft (fresh)
- m (sea)
- m (fresh)

-The upper right hand portion of the Hemispherical Endcap Analysis Dialog Box provides weights in air and water (**Air Wt** and **Water Wt**) with the option of expressing these values in either lbs or kg. The weight in water (when completely submerged) assumes that the internal volume of the hemispherical endcap is empty.

-The bottom portion of the Hemispherical Endcap Analysis Dialog Box provides a results table of applicable stresses and deflections for an analysis that are displayed as a function of the user selected **Distortion Pressure Range**. The following information is provided for an analysis as a function of the selected pressure range:

Depth (equivalent water Depth of selected pressure)

Max Meridional Stress (Maximum Stress in hemisphere wall in direction of hemisphere cross section meridian)

Max Hoop Stress (Maximum Stress in hemisphere wall in direction normal to hemisphere cross section)

Max Equiv Stress (Maximum uniaxial stress that is equivalent to the three dimensional stress state that exists in the hemisphere wall for predicting failure of ductile materials by comparison to the uniaxial yield strength)

Max Shear Stress (Maximum Shear Stress in hemisphere wall, N/A for external pressure)

Avg Seat Stress (Average bearing Stress on flat annular area at hemisphere equator)

dID (displacement of hemisphere Inner Diameter)

dOD (displacement of hemisphere Outer Diameter)

An explanation of the orientation of the above stresses is provided in Appendix D: PRESSURE VESSEL STRESSES. Again, the user has an option as to which units will be used to display the data in the results table. Rows of data for an applied pressures at or greater than the calculated failure pressure are highlighted by the program to warn the user. As discussed in Appendix C: FORMULAS USED BY UNDER PRESSURE, Maximum Equivalent Stress is used as a failure criteria for Metal Hemispherical Endcaps, and is not applicable to hemispheres fabricated from other Main Category Materials Ceramics, Glass, and Plastics.

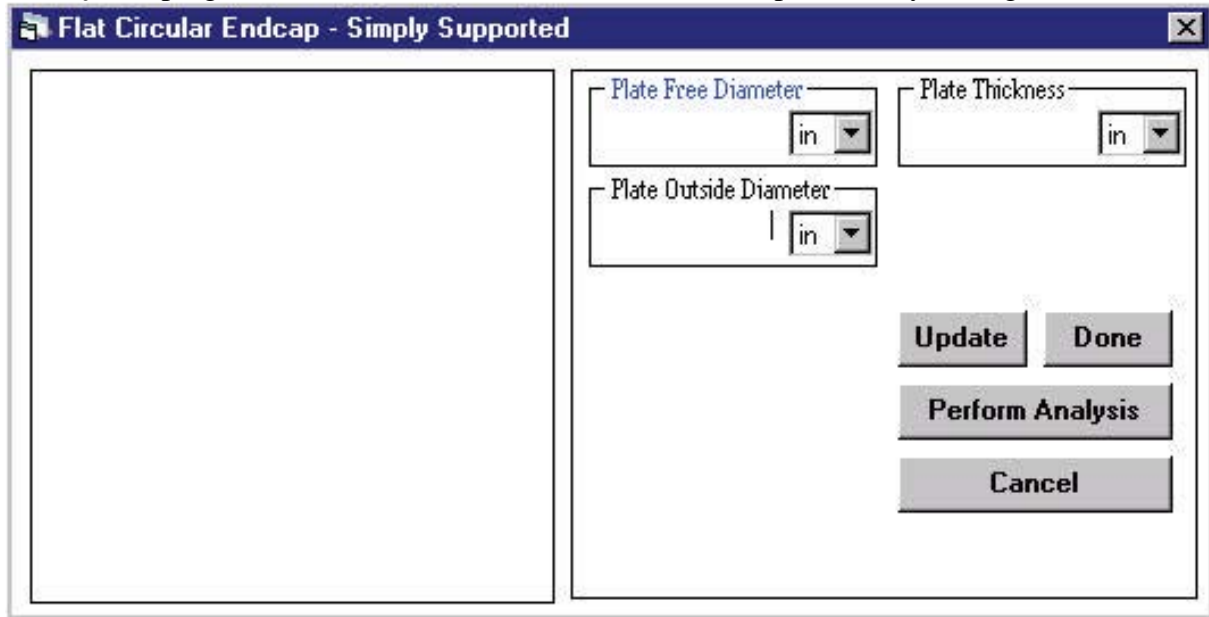
FLAT CIRCULAR ENDCAP ANALYSIS

Flat Circular Endcap Edge Restraint Options:

-The significance of Edge Restraint Options (Simply Supported or Fixed) for Flat Circular Plates is defined in APPENDIX B: FLAT ENDCAP BOUNDARY CONDITIONS. In practical application, the edge restraint of a flat circular plate is not likely to be purely simply supported or purely fixed. If in doubt about which edge restraint is most appropriate for a given Flat Circular Endcap design, the use of the simply supported edge restraint results in higher stresses in the plate and consequently is a more conservative approach for the designer to use.

Flat Circular Endcap Geometry:

-By selecting Analysis Type **Endcap Only** in the Application Window, selecting **Endcap Configuration Flat Circular**, selecting appropriate **Edge Restraint Options** and clicking on **Enter Geometry**, the program user can access the Flat Circular Endcap Geometry Dialog Box:



FLAT CIRCULAR ENDCAP GEOMETRY DIALOG BOX

-Flat Circular Endcap geometry is defined by the variables **Plate Outside Diameter**, **Plate Free Diameter** and **Plate Thickness**. APPENDIX A: PRESSURE VESSEL GEOMETRIES shows a figure of a flat circular endcap and the variables used to define its geometry.

-After using the cursor and keyboard to define the endcap geometry, a three dimensional view of the resulting flat circular endcap geometry is generated (using a dashed line to denote the Plate Free Diameter) on the left hand side of the Flat Circular Endcap Geometry Dialog Box.

-Clicking on **Done** saves the geometry and closes the Flat Circular Endcap Geometry Dialog Box

-Clicking on **Cancel** closes the Flat Circular Endcap Geometry Dialog Box without saving user inputs.

-Clicking on **Perform Analysis** analyzes the Flat Circular Endcap geometry and opens the Flat Circular Endcap Analysis Dialog Box.

Flat Circular Endcap Formulas:

-Formulas used for Flat Circular Endcap stress analysis are presented in APPENDIX C: FORMULAS USED BY UNDER PRESSURE.

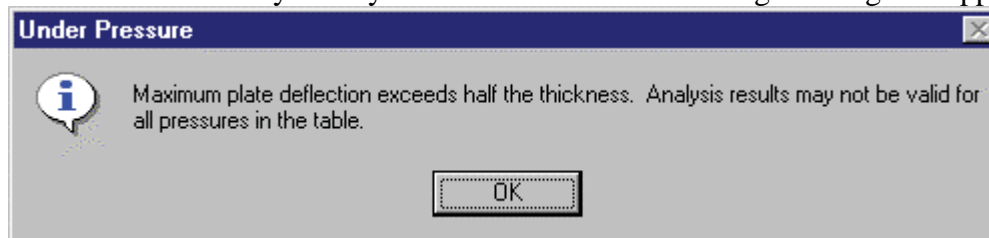
Flat Circular Endcap Formula Assumptions:

-Stress analysis formulas assume that the plate is flat and has a constant thickness

-The ratio of the Plate Free Diameter to the Plate Thickness is greater than or equal to 4. If this assumption is violated, the results of the analysis may not be valid and the following warning will appear:

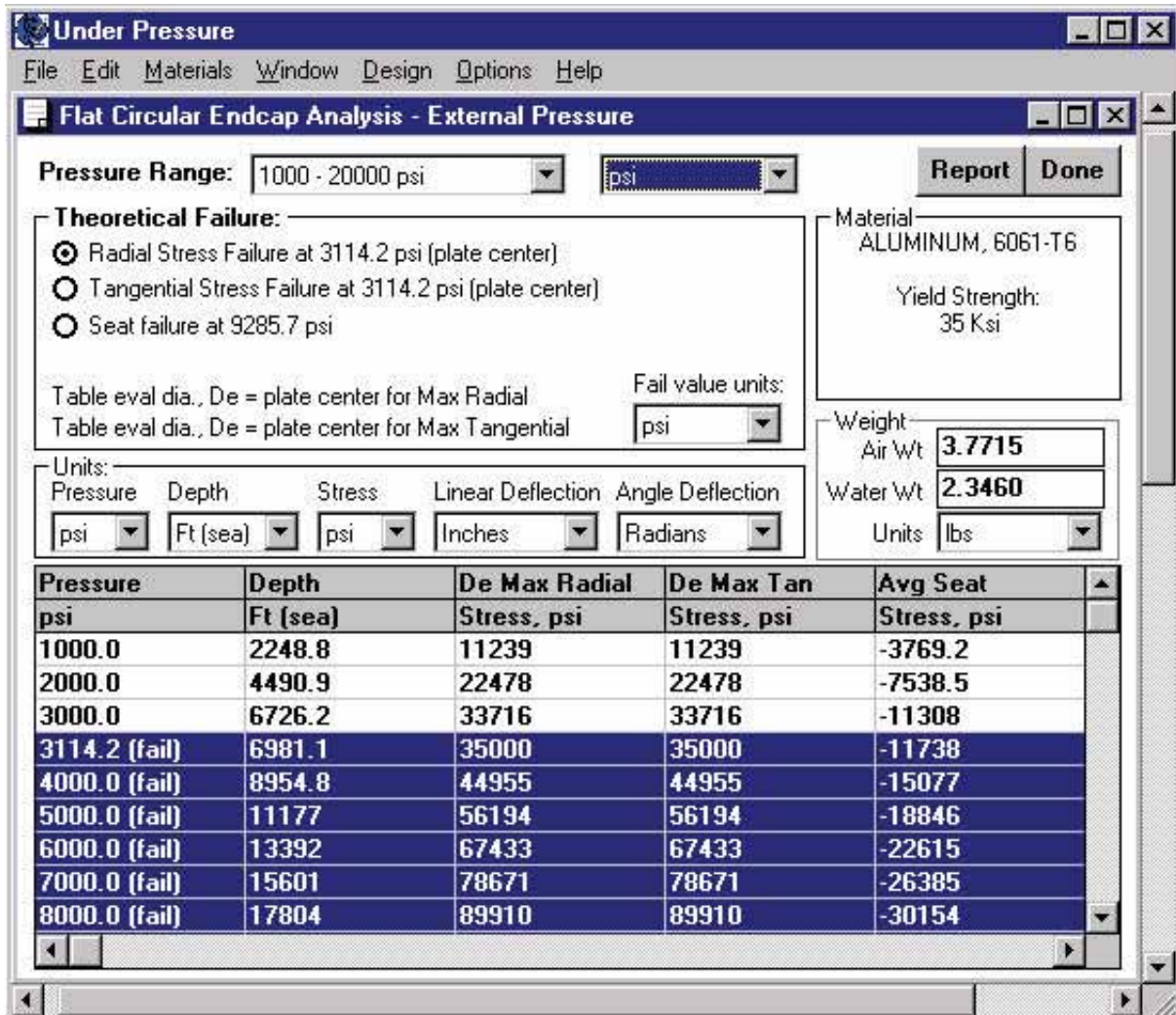


-The maximum deflection of the plate does not exceed one half the plate thickness. If this assumption is violated, the results of the analysis may not be valid and the following warning will appear:



Flat Circular Endcap Analysis Results:

-The following Flat Circular Endcap Analysis Dialog Box (from Example 2 of this manual) will be used to highlight the data and options available to the user after an analysis has been performed:



FLAT CIRCULAR ENDCAP ANALYSIS DIALOG BOX EXAMPLE

-Clicking on the scroll arrow adjacent to the two boxes titled **Distortion Pressure Range** (at the top of Flat Circular Endcap Analysis Dialog Box) allows the user to select the range of pressure (in units of psi or Bar) that will be used with the data in the results table below. For example if psi is selected as the pressure units of choice, one of the following pressure ranges can be selected for presenting the results:

- 0.1-2 psi
- 1-20 psi
- 10-200 psi
- 100-2000 psi
- 1000-20,000 psi
- 10,000-200,000 psi

-The **Theoretical Failure** portion indicates the pressure or depth and radial location at which the three modes of failure for a Flat Circular Endcap analysis, **Radial Stress Failure**, **Tangential Stress Failure**, and **Seat Failure** occur. The mode of failure that occurs at the lowest (least in magnitude) pressure is highlighted with the aid of an option button. Clicking on the scroll arrow adjacent to the **Fail value units** box allows the user to select the units at which failure will be presented:

- Ksi
- Kbar
- psi
- Bar
- Ft (sea)
- Ft (fresh)
- m (sea)
- m (fresh)

The bottom of the Theoretical Failure portion of the Flat Circular Endcap Analysis Dialog Box provides the radial location of the Plate (i.e. the diameter of evaluation = D_e) for which the maximum radial and tangential stresses are presented in the results table.

-The upper right hand portion of the Flat Circular Endcap Analysis Dialog Box provides weights in air and water (**Air Wt** and **Water Wt**) with the option of expressing these values in either lbs or kg.

-The bottom portion of the Flat Circular Endcap Analysis Dialog Box provides a results table of applicable stresses and deflections for an analysis that are displayed as a function of the user selected **Distortion Pressure Range**. The following information is provided for an analysis as a function of the selected pressure range:

Depth (equivalent water Depth of selected pressure)

De Radial Stress (maximum Stress in plate cross section in direction normal to plate center line at noted evaluation Diameter)

De Tangential Stress (maximum Stress in plate in direction normal to plate cross section at noted evaluation Diameter)

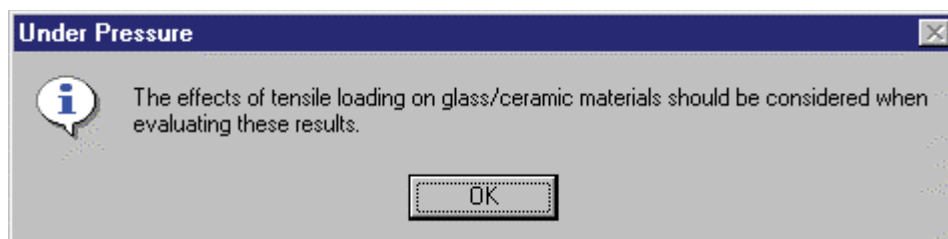
Avg Seat Stress (Average bearing Stress on annular area of plate between plate outside diameter and plate free diameter)

CL Deflection (Deflection of Center of plate in direction of plate centerline)

An explanation of the orientation of the above stresses is provided in Appendix D: PRESSURE VESSEL STRESSES. Again, the user has an option as to which units will be used to display the data in the results table. Rows of data for an applied pressures at or greater than the calculated failure pressure are highlighted by the program to warn the user. As discussed in Appendix C: FORMULAS USED BY UNDER PRESSURE, the maximum radial and tangential membrane stresses occur on the flat surfaces of the flat circular plate with the concave side of the plate loaded in compression and convex side of the plate loaded in tension (the radial and tangential membrane stresses are zero at the plate midthickness).

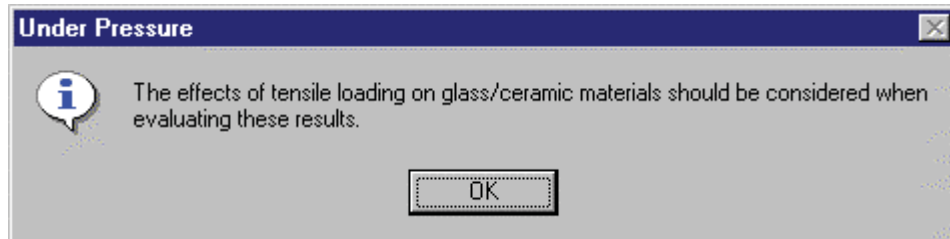
ANALYSIS RESULTS WARNING MESSAGES:

-If Main Category Glass or Ceramic is selected for the design of a pressure vessel subjected to internal pressure, the following warning will appear with the Analysis results Dialog Box:



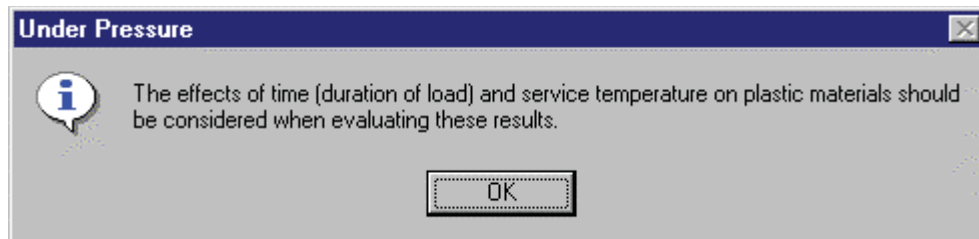
Explanation: Brittle materials such as Glass or Ceramics are generally intended for use in compression only, subjecting these materials to internal pressure results in tensile loads.

-If Main Category Glass or Ceramic is selected for the design of flat circular or flat annular endcaps, the following warning will appear with the Analysis results Dialog Box.



Explanation: Brittle materials such as Glass or Ceramics are generally intended for use in compression only, the convex side of pressure and/or line loaded flat endcaps will be subjected to tensile stresses.

-If Main Category Plastic is selected for the design of a pressure vessel, the following warning will appear with the Analysis results Dialog Box.



Explanation: Stress and buckling analysis of plastic pressure vessel components can have a significant dependence on service temperature and time (creep behavior) and should therefore be considered by the pressure vessel designer.

REPORT GENERATION

Part of the process of evaluation of a design is the comparison of the results that are achieved using different design parameters (materials, shapes, etc.). Under Pressure supports saving the files for easy retrieval, but it is often more convenient to have a printed copy of the information.

Furthermore, it is often useful to have the ability to transfer the generated results of the analysis into another program for additional analysis to include in a report, or to send a copy of the results electronically via e-mail. Under Pressure supports the following types of report generation:

- 1) Printed report (on paper, overheads, etc). This generates a printed report of the analysis to your current printer, as set up in the print dialog box.
- 2) Print report to file. This is used to make a file copy of the analysis report. This file can then be opened with a word processor, or other program to format the contents.
- 3) “Copy Analysis to Clipboard”. This is useful to quickly copy the entire report from the analysis window into other programs. In particular, this is useful for placing the report in e-mails, or to put it in a spreadsheet such as Microsoft Excel.

This chapter will cover how these reports are generated, and how the information can be formatted or sent. In the following sample design, we want to generate a printout of the properties of a tube type pressure vessel made from 4”, schedule 80 PVC, 10” in length.

The screenshot shows the 'Under Pressure' software interface. The main window is titled 'Under Pressure' and contains a 'Tube' sub-window. The 'Tube' window displays a 3D model of a tube and the following input parameters:

- Tube I.D.: 3.8260 in
- Wall thickness: 0.33700 in
- Tube O.D.: 4.5000 in
- Tube length: 10.000 in
- Material: Maintain I.D. constant, Maintain O.D. constant
- Buttons: Update, Done, Perform Analysis, Cancel, Pipe Tables

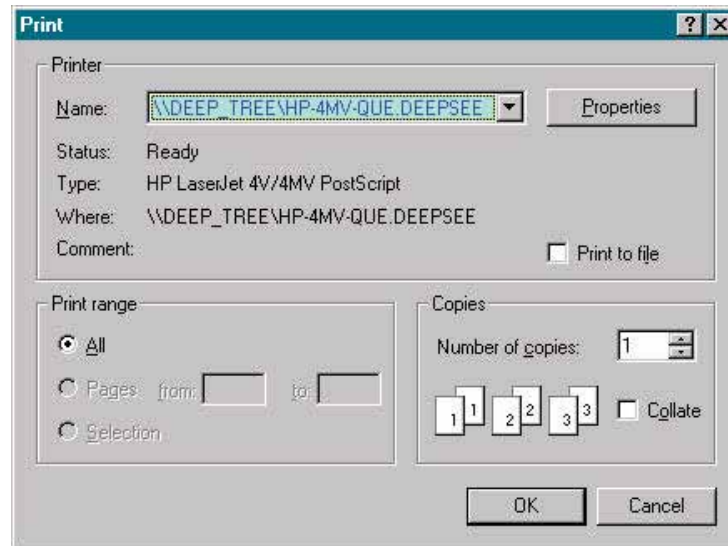
Below the 'Tube' window is the 'Tube Analysis - External Pressure' window. It displays the following information:

- Pressure Range: 100 - 2000 psi
- Theoretical Failure: Thin Wall Buckling at 0.62962 Ksi by 2 nodes, Shell failure at 0.83137 Ksi (Thick wall eqn), Seat failure N/A, Shear failure N/A
- Material: PLASTIC, POLYVINYL CHLORIDE (PVC), Ultimate Strength: 6 Ksi, Working Strength: 0.6 Ksi
- Weight: Air Wt: 2.8979, Water Wt: -3.7930
- Units: Pressure: Ksi, Depth: Ft (sea), Stress: Ksi, Linear Deflection: Inches, Angle Deflection: Radians
- Table of Results:

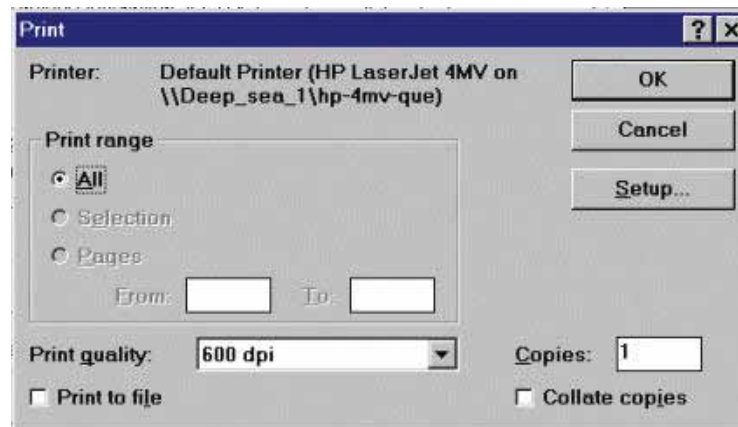
| Pressure | Depth | Max Axial Stress, Ksi | Max Hoop Stress, Ksi | Max Equiv Stress, Ksi |
|----------------|--------|-----------------------|----------------------|-----------------------|
| 0.10000 | 225.19 | -0.36085 | -0.72170 | N/A |
| 0.20000 | 450.31 | -0.72170 | -1.4434 | N/A |
| 0.30000 | 675.37 | -1.0826 | -2.1651 | N/A |
| 0.40000 | 900.35 | -1.4434 | -2.8868 | N/A |
| 0.50000 | 1125.3 | -1.8043 | -3.6085 | N/A |
| 0.60000 | 1350.1 | -2.1651 | -4.3302 | N/A |
| 0.62962 (fail) | 1416.7 | -2.2720 | -4.5440 | N/A |
| 0.70000 (fail) | 1574.9 | -2.5260 | -5.0519 | N/A |
| 0.80000 (fail) | 1733.6 | -2.8868 | -5.7736 | N/A |

A printed report can be generated only when the Analysis window is active. When it is active, the “Print Setup...” and “Print...” menu options under the file menu become available.

In order to print, you can select “Print...” from the File menu. It will pull up the print dialog box similar to that seen below. Note that the printer information will depend on your printer.



Windows NT and '98 Print Screen



Windows 95 Print Screen

The top selection specifies what printer you want to print to (similar to if you had selected the “Print Setup...” dialog box). You can specify the number of copies that you want to generate, and also specify “Print to file” (more on that later). Clicking on the “OK” button will start the generation of the printout. It will vary in number of pages depending on the type of analysis that is being printed.

Another way to quickly print the results to the printer is to simply depress the “Report” button that appears on the Analysis form, next to the “Done” button. Pressing this button will force an immediate print of the analysis to your default printer.

Note: Whenever you either print or copy the report, all the values are recalculated and any error or warning messages will be repeated.

In this case, the information that was generated in the report appears on the following page:

Under Pressure Ver.4.05 15:52:2107-17-1998
TUBE CONFIGURATION (External Pressure)

Inner Diameter : 3.4380 in
Outer Diameter : 4.5000 in
Wall Thickness : 0.53100 in
Tube Length : 10.000 in

Weight in air : 3.1516 lbs
Weight in water : -2.7393 lbs

Failure Mode: Shell failure at 1.2489 Ksi (Thick wall eq's)

Thin Wall Buckling at 2.4984 Ksi by 2 nodes
Seat failure N/A
Shear failure N/A

PLASTIC, POLYVINYL CHLORIDE (PVC) Properties:

Ultimate Strength : 6 Ksi
Working Strength : 0.6 Ksi
Elastic Modulus : 0.35 Mpsi
Density : 0.0476 lb/cu in
Poisson's Ratio : 0.36
Comments :
Molded or Extruded, Rigid
Compressive Ultimate Strength = 8 Ksi

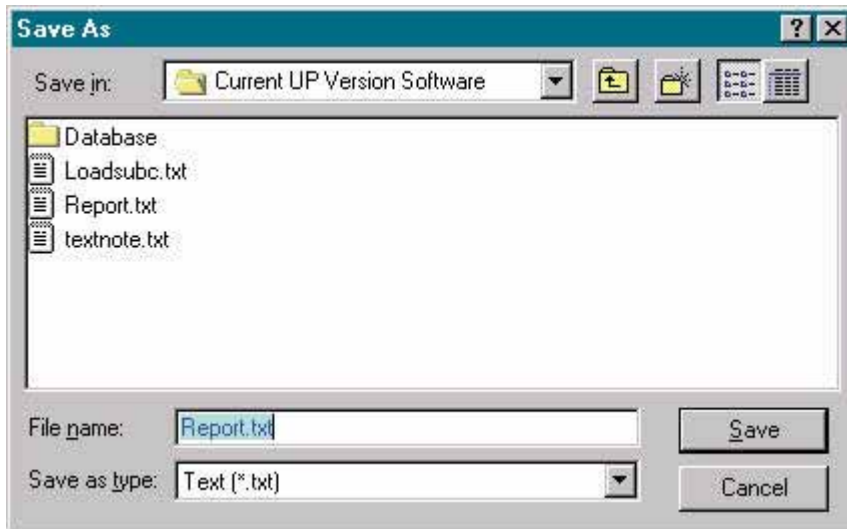
Tube distortion values:

| Pressure Ksi | Depth Ft (sea) | Max Axial Stress, Ksi | Max Hoop Stress, Ksi | Max Equiv Stress, Ksi | d ID Inches | d OD Inches | d Length Inches |
|-----------------|-------------------|--------------------------|-------------------------|--------------------------|----------------|----------------|--------------------|
| 0.10000 | 225.19 | -0.24021 | -0.48042 | N/A | -0.0038696 | -0.0033164 | -0.0019217 |
| 0.20000 | 450.31 | -0.48042 | -0.96084 | N/A | -0.0077393 | -0.0066328 | -0.0038433 |
| 0.30000 | 675.37 | -0.72063 | -1.4413 | N/A | -0.011609 | -0.0099492 | -0.0057650 |
| 0.40000 | 900.35 | -0.96084 | -1.9217 | N/A | -0.015479 | -0.013266 | -0.0076867 |
| 0.50000 | 1125.3 | -1.2010 | -2.4021 | N/A | -0.019348 | -0.016582 | -0.0096084 |
| 0.60000 | 1350.1 | -1.4413 | -2.8825 | N/A | -0.023218 | -0.019898 | -0.011530 |
| 0.70000 | 1574.9 | -1.6815 | -3.3629 | N/A | -0.027088 | -0.023215 | -0.013452 |
| 0.80000 | 1799.6 | -1.9217 | -3.8433 | N/A | -0.030957 | -0.026531 | -0.015373 |
| 0.90000 | 2024.3 | -2.1619 | -4.3238 | N/A | -0.034827 | -0.029848 | -0.017295 |
| 1.0000 | 2248.8 | -2.4021 | -4.8042 | N/A | -0.038696 | -0.033164 | -0.019217 |
| 1.1000 | 2473.3 | -2.6423 | -5.2846 | N/A | -0.042566 | -0.036480 | -0.021138 |
| 1.2000 | 2697.8 | -2.8825 | -5.7650 | N/A | -0.046436 | -0.039797 | -0.023060 |
| 1.2489* | 2807.5 | -3.0000 | -6.0000 | N/A | -0.048328 | -0.041419 | -0.024000 |
| 1.3000* | 2922.2 | -3.1227 | -6.2454 | N/A | -0.050305 | -0.043113 | -0.024982 |
| 1.4000* | 3146.5 | -3.3629 | -6.7259 | N/A | -0.054175 | -0.046430 | -0.026903 |
| 1.5000* | 3370.7 | -3.6031 | -7.2063 | N/A | -0.058045 | -0.049746 | -0.028825 |
| 1.6000* | 3594.9 | -3.8433 | -7.6867 | N/A | -0.061914 | -0.053063 | -0.030747 |
| 1.7000* | 3819.0 | -4.0836 | -8.1671 | N/A | -0.065784 | -0.056379 | -0.032668 |
| 1.8000* | 4043.0 | -4.3238 | -8.6475 | N/A | -0.069654 | -0.059695 | -0.034590 |
| 1.9000* | 4267.0 | -4.5640 | -9.1279 | N/A | -0.073523 | -0.063012 | -0.036512 |
| 2.0000* | 4490.9 | -4.8042 | -9.6084 | N/A | -0.077393 | -0.066328 | -0.038433 |

* = after housing failure

Note: The “default text font” for different printers varies and so your actual results may appear somewhat different from that shown.

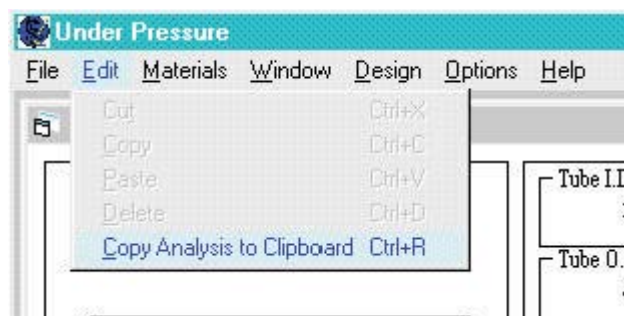
If instead of printing to paper, you want to make a file copy of this information, then you can select “Print...” from the “File” menu, and then choose the “Print to file” option. When you select “Ok” a “Save As” dialog box opens as below:



The default file name is “Report.txt”. The information is saved in a tab delimited ASCII format so that it can be imported into applications such as spreadsheets or tables easily. Because there are different title lengths in the different column titles, the information may appear to overlap or be mixed up if brought up in a word processor that is not formatting the information as a table. In order to solve this problem, the tab stops need to be spaced wider (if in a word processor), or the information can be placed imported into a table or spreadsheet.

Copy to clipboard

When the analysis screen is active, not only can you print the results, but also the results can be copied to the clipboard for insertion into another program.



Selecting the “Copy Analysis to Clipboard” option will copy the entire analysis (contents of the printed report) into the clipboard to allow it to be pasted into another application.

An obvious place to paste this information is into a spreadsheet or word processor. Since the information is tab delimited, pasting it into a spreadsheet will cause the information to all fall into its own cell:

The screenshot shows a Microsoft Excel spreadsheet with the following content:

Under Pressure Ver.4.0x

15:05:58 7/23/98

TUBE CONFIGURATION (Internal Pressure)

Inner Diameter : 3.8260 in
 Outer Diameter : 4.5000 in
 Wall Thickness : 0.33700 in
 Tube Length : 10.000 in

Weight in air : 2.0979 lbs
 Weight in water : -3.7930 lbs

Failure Mode: Shell failure at 96.509 psi (Thick wall eq's)

Thin Wall Buckling N/A
 Seat failure N/A
 Shear failure N/A

PLASTIC, POLYVINYL CHLORIDE (PVC) Properties:

Ultimate Strength : 6 Ksi
 Working Strength : 0.6 Ksi
 Elastic Modulus : 0.35 Mpsi
 Density : 0.0476 lb/cu in
 Poisson's Ratio : 0.36
 Comments :
 Molded or Extruded, Rigid
 Compressive Ultimate Strength = 8 Ksi

Tube distortion values:

| Pressure psi | Depth Ft (sea) | Max Axial Stress, psi | Max Hoop Stress, psi | Max Equiv Stress, psi | d ID Inches | d OD Inches | d Length Inches |
|-----------------|-------------------|--------------------------|-------------------------|--------------------------|----------------|----------------|--------------------|
| 10 | 22.522 | 26.085 | 62.17 | N/A | 0.000616 | 0.00055 | 0.000209 |
| 20 | 45.044 | 52.17 | 124.34 | N/A | 0.001233 | 0.0011 | 0.000417 |
| 30 | 67.564 | 78.256 | 186.51 | N/A | 0.001849 | 0.00165 | 0.000626 |
| 40 | 90.085 | 104.34 | 248.68 | N/A | 0.002465 | 0.0022 | 0.000835 |
| 50 | 112.6 | 130.43 | 310.85 | N/A | 0.003082 | 0.00275 | 0.001043 |
| 60 | 135.12 | 156.51 | 373.02 | N/A | 0.003698 | 0.0033 | 0.001252 |
| 70 | 157.64 | 182.6 | 435.19 | N/A | 0.004314 | 0.00385 | 0.001461 |
| 80 | 180.16 | 208.68 | 497.36 | N/A | 0.004931 | 0.0044 | 0.00167 |
| 90 | 202.67 | 234.77 | 559.53 | N/A | 0.005547 | 0.00495 | 0.001878 |
| 96.509 | 217.33 | 251.75 | 600 | N/A | 0.005948 | 0.005308 | 0.002014 |
| 100 | 225.19 | 260.85 | 621.7 | N/A | 0.006163 | 0.0055 | 0.002087 |
| 110 | 247.71 | 286.94 | 683.87 | N/A | 0.006779 | 0.00605 | 0.002296 |
| 120 | 270.22 | 313.02 | 746.04 | N/A | 0.007396 | 0.0066 | 0.002504 |
| 130 | 292.73 | 339.11 | 808.21 | N/A | 0.008012 | 0.00715 | 0.002713 |
| 140 | 315.25 | 365.19 | 870.38 | N/A | 0.008628 | 0.0077 | 0.002922 |
| 150 | 337.76 | 391.28 | 932.56 | N/A | 0.009245 | 0.00825 | 0.00313 |
| 160 | 360.27 | 417.36 | 994.73 | N/A | 0.009861 | 0.0088 | 0.003339 |
| 170 | 382.78 | 443.45 | 1056.9 | N/A | 0.010477 | 0.00935 | 0.003548 |
| 180 | 405.29 | 469.53 | 1119.1 | N/A | 0.011094 | 0.0099 | 0.003756 |
| 190 | 427.8 | 495.62 | 1181.2 | N/A | 0.01171 | 0.01045 | 0.003965 |
| 200 | 450.31 | 521.7 | 1243.4 | N/A | 0.012326 | 0.011 | 0.004174 |

* = after housing failure

On the other hand, pasting it into a word-processor will have varying results:

Under Pressure Ver.4.05 16:41:26 07-17-1998
 TUBE CONFIGURATION (External Pressure)

Inner Diameter : 3.4380 in
 Outer Diameter : 4.5000 in
 Wall Thickness : 0.53100 in
 Tube Length : 10.000 in

Weight in air : 3.1516 lbs
 Weight in water: -2.7393 lbs

Failure Mode: Shell failure at 1.2489 Ksi (Thick wall eq's)

Thin Wall Buckling at 2.4984 Ksi by 2 nodes
 Seat failure N/A
 Shear failure N/A

PLASTIC, POLYVINYL CHLORIDE (PVC) Properties:
 Ultimate Strength : 6 Ksi
 Working Strength : 0.6 Ksi
 Elastic Modulus : 0.35 Mpsi
 Density : 0.0476 lb/cu in
 Poisson's Ratio : 0.36
 Comments :
 Molded or Extruded, Rigid
 Compressive Ultimate Strength = 8 Ksi

Tube distortion values:

| Pressure Ksi | Depth Ft (sea) | Max Axial Stress, Ksi | Max Hoop Stress, Ksi | Max Equiv d Stress, Ksi | ID Inches | d OD Inches | d Length Inches |
|-----------------|-------------------|--------------------------|-------------------------|----------------------------|--------------|----------------|--------------------|
| 0.10000 | 225.19 | -0.24021 | -0.48042 | N/A | -0.0038696 | -0.0033164 | -0.0019217 |
| 0.20000 | 450.31 | -0.48042 | -0.96084 | N/A | -0.0077393 | -0.0066328 | -0.0038433 |
| 0.30000 | 675.37 | -0.72063 | -1.4413 | N/A | -0.011609 | -0.0099492 | -0.0057650 |
| 0.40000 | 900.35 | -0.96084 | -1.9217 | N/A | -0.015479 | -0.013266 | -0.0076867 |
| 0.50000 | 1125.3 | -1.2010 | -2.4021 | N/A | -0.019348 | -0.016582 | -0.0096084 |
| 0.60000 | 1350.1 | -1.4413 | -2.8825 | N/A | -0.023218 | -0.019898 | -0.011530 |
| 0.70000 | 1574.9 | -1.6815 | -3.3629 | N/A | -0.027088 | -0.023215 | -0.013452 |
| 0.80000 | 1799.6 | -1.9217 | -3.8433 | N/A | -0.030957 | -0.026531 | -0.015373 |
| 0.90000 | 2024.3 | -2.1619 | -4.3238 | N/A | -0.034827 | -0.029848 | -0.017295 |
| 1.0000 | 2248.8 | -2.4021 | -4.8042 | N/A | -0.038696 | -0.033164 | -0.019217 |
| 1.1000 | 2473.3 | -2.6423 | -5.2846 | N/A | -0.042566 | -0.036480 | -0.021138 |
| 1.2000 | 2697.8 | -2.8825 | -5.7650 | N/A | -0.046436 | -0.039797 | -0.023060 |
| * 1.2489 | 2807.5 | -3.0000 | -6.0000 | N/A | -0.048328 | -0.041419 | -0.024000 |
| * 1.3000 | 2922.2 | -3.1227 | -6.2454 | N/A | -0.050305 | -0.043113 | -0.024982 |
| * 1.4000 | 3146.5 | -3.3629 | -6.7259 | N/A | -0.054175 | -0.046430 | -0.026903 |
| * 1.5000 | 3370.7 | -3.6031 | -7.2063 | N/A | -0.058045 | -0.049746 | -0.028825 |
| * 1.6000 | 3594.9 | -3.8433 | -7.6867 | N/A | -0.061914 | -0.053063 | -0.030747 |
| * 1.7000 | 3819.0 | -4.0836 | -8.1671 | N/A | -0.065784 | -0.056379 | -0.032668 |
| * 1.8000 | 4043.0 | -4.3238 | -8.6475 | N/A | -0.069654 | -0.059695 | -0.034590 |
| * 1.9000 | 4267.0 | -4.5640 | -9.1279 | N/A | -0.073523 | -0.063012 | -0.036512 |
| * 2.0000 | 4490.9 | -4.8042 | -9.6084 | N/A | -0.077393 | -0.066328 | -0.038433 |

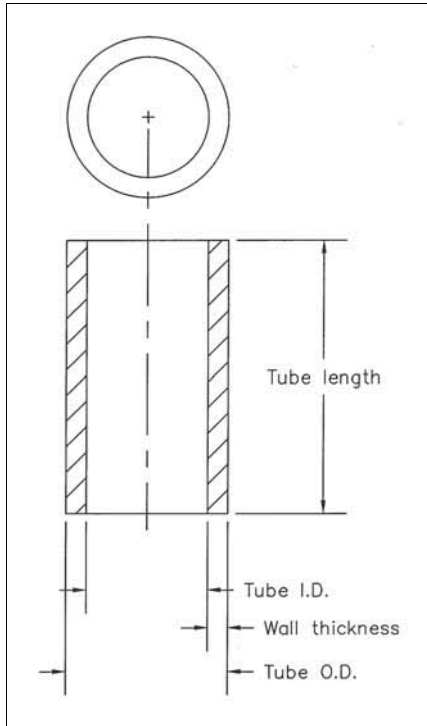
* = after housing failure

Notice how the last few columns appear mis-aligned? To correct this, the information in the table can be selected and either reformatted so that the tabs are wider, or it can be converted into a table. In Microsoft Word, converting it into a table is a matter of selecting the table contents (noted by the dashed line above), and then selecting "Convert Text to Table". This will yield a formatted table.

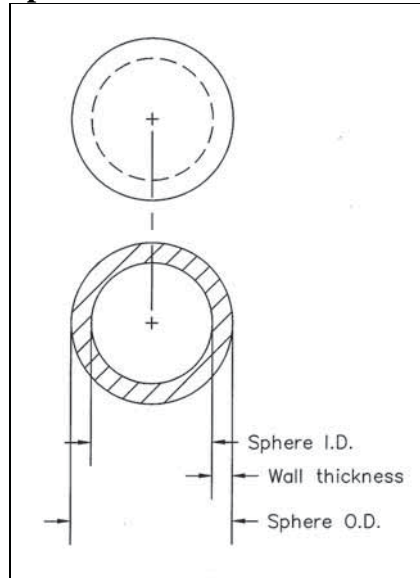
APPENDIX A: PRESSURE VESSEL GEOMETRIES

HOUSINGS

Tube:

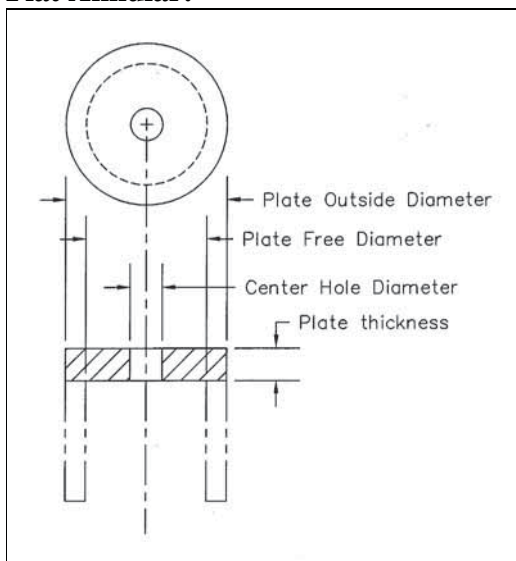


Sphere:

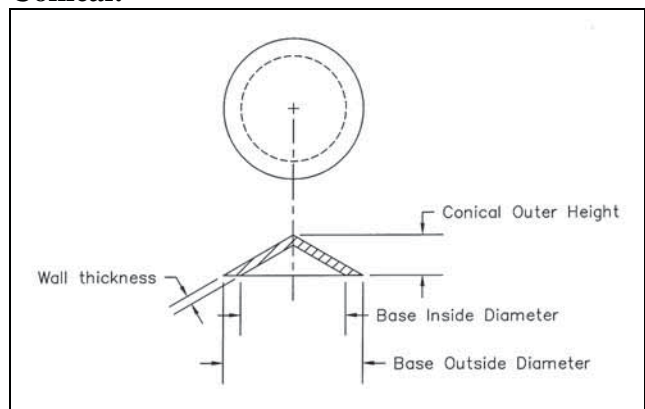


ENDCAP

Flat Annular:

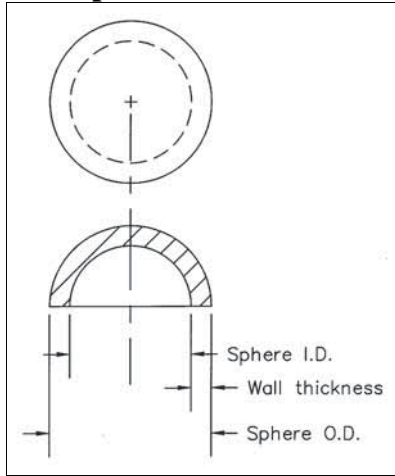


Conical:

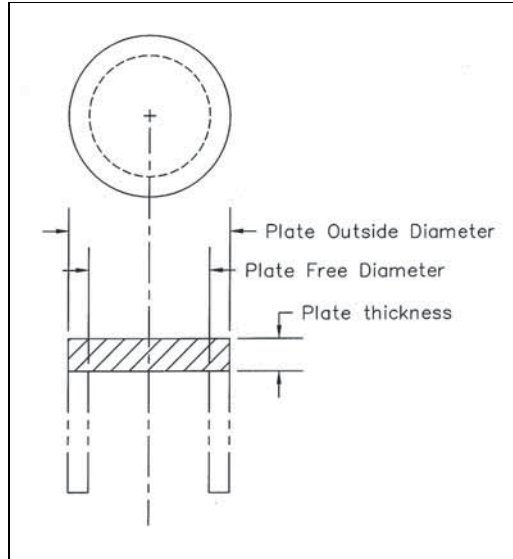


ENDCAP (Continued)

Hemispherical:



Flat Circular:



APPENDIX B: FLAT ENDCAP BOUNDARY CONDITIONS

DEFINITIONS:

Fixed: Boundary condition at the circumferential edge of a plate that prevents radial rotations and transverse deflections but allows for radial displacements.

Free: Boundary condition at the circumferential edge of a plate that allows for radial rotations and both transverse and radial displacements.

Guided: Boundary condition at the circumferential edge of a plate that prevents radial rotations but allows for transverse and radial displacements.

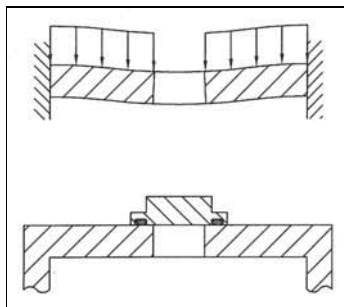
Simply Supported: Boundary condition at the circumferential edge of a plate that prevents transverse deflections but allows for radial rotations and displacements.

FIGURES:

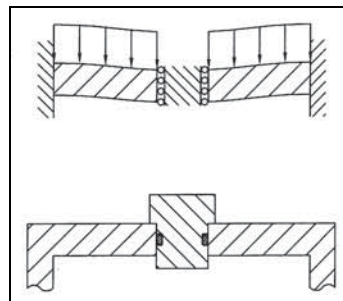
The following figures show a symbolic representation of the various flat endcap boundary conditions adjacent to a pressure vessel assembly cross-section for which the boundary condition might be applicable:

FLAT ANNULAR

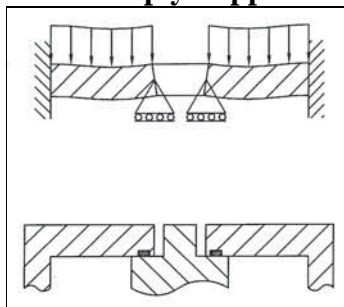
Fixed/Free:



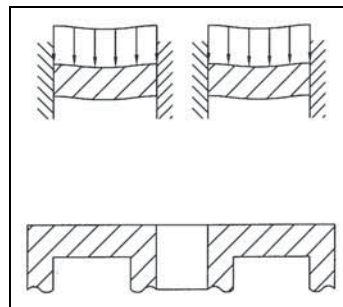
Fixed/Guided:



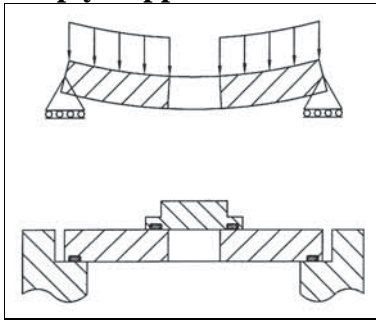
Fixed/Simply Supported:



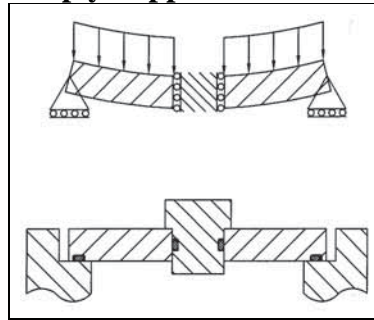
Fixed/Fixed:



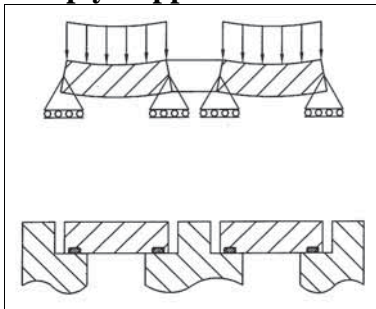
Simply Supported/Free:



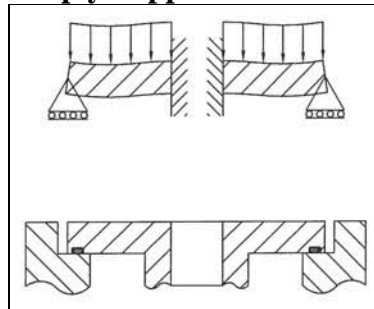
Simply Supported/Guided:



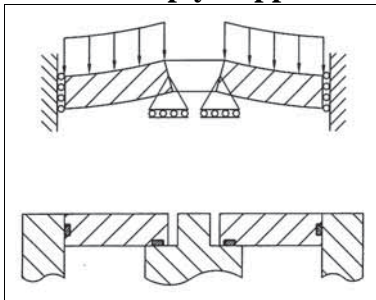
Simply Supported /Simply Supported:



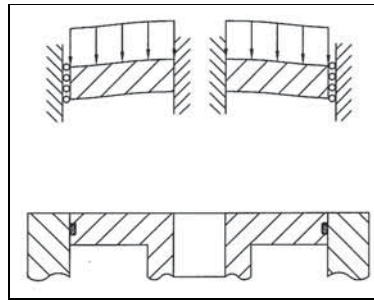
Simply Supported/Fixed:



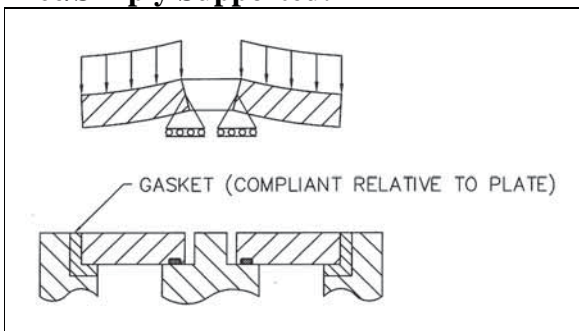
Guided/Simply Supported:



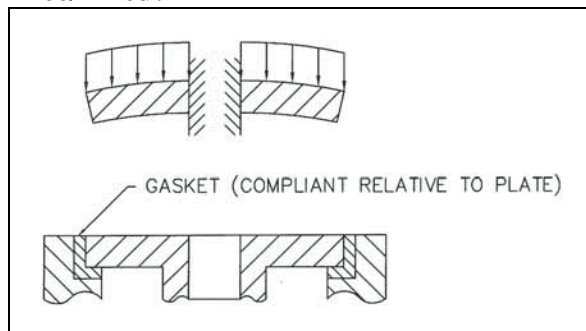
Guided/Fixed:



Free/Simply Supported:

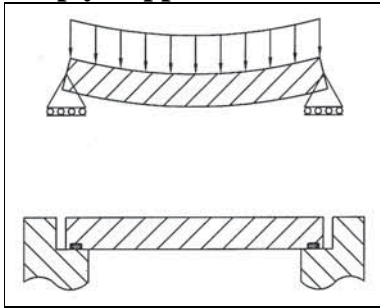


Free/Fixed:

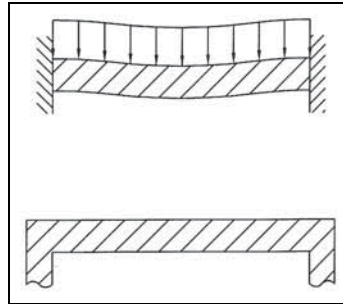


FLAT CIRCULAR

Simply Supported:



Fixed:



APPENDIX C: FORMULAS USED BY UNDER PRESSURE

All formulas are based on “Roark’s Formulas for Stress & Strain, Sixth Edition,” by Warren C. Young, McGraw-Hill, Inc., 1989.

TUBE:

Thin wall stress formulas: Table 28, case 1c, pg. 519.

Thick wall stress formulas: Table 32, case 1a-1d, pgs. 638-639.

Thin wall buckling formulas: Table 35, case 20, pg. 690.

SPHERE:

Thin wall stress formulas: Table 28, case 3a, pg. 523.

Thick wall stress formulas: Table 32, case 2a-2b, pg. 640.

Thin wall buckling formulas: Table 35, case 22, pg. 691.

FLAT ANNULAR ENDCAP:

Stress Formulas: Table 24, case 1a-1l, 2a-1, pgs. 398-408.

CONICAL ENDCAP:

Thin wall stress formulas: Table 28, case 2a, pg. 520.

Thin wall buckling formulas: see Table 23, case 23, pg. 691 for approximate formula for truncated cones with closed ends (not part of Under Pressure)

HEMISPHERICAL ENDCAP:

Thin wall stress formulas: Table 28, case 3a, pg. 523.

Thick wall stress formulas: Table 32, case 2a-2b, pg. 640.

Thin wall buckling formulas: Table 35, case 22, pg. 691.

FLAT CIRCULAR ENDCAP:

Stress Formulas: Table 24, case 10a-10b, pg. 429.

FAILURE CRITERIA:

Failure criteria formulas used are dependent on Material Main category/Analysis Type (pressure vessel geometry)

| <u>Material Main Category/Analysis Type</u> | <u>Criteria:</u> |
|---|------------------|
| 1. Ceramics/Tube..... | A |
| 2. Glass/Tube..... | A |
| 3. Metals/Tube..... | B |
| 4. Plastics/Tube..... | C |
| 5. Ceramics/Sphere..... | A |
| 6. Glass/Sphere..... | A |
| 7. Metals/Sphere..... | D |
| 8. Plastics/Sphere..... | E |

| | |
|---------------------------------|---|
| 9. Ceramics/Flat Annular..... | F |
| 10. Glass/Flat Annular..... | F |
| 11. Metals/Flat Annular..... | G |
| 12. Plastics/Flat Annular..... | H |
| 13. Ceramics/Conical..... | I |
| 14. Glass/Conical..... | I |
| 15. Metals/Conical..... | J |
| 16. Plastics/Conical..... | K |
| 17. Ceramics/Hemispherical..... | I |
| 18. Glass/Hemispherical..... | I |
| 19. Metals/Hemispherical..... | L |
| 20. Plastics/Hemispherical..... | M |
| 21. Ceramics/Flat Circular..... | F |
| 22. Glass/Flat Circular..... | F |
| 23. Metals/Flat Circular..... | N |
| 24. Plastics/Flat Circular..... | O |

A - Maximum hoop stress compared to Ultimate Strength (tensile) for internal pressure loading, or compared to Ultimate Strength (compressive) for external pressure loading.

B - Maximum equivalent membrane stress (von Mises stress, constant energy of distortion stress) compared to Yield Strength, see theory (4), pg. 26 of Roark's.

C - Maximum hoop stress compared to Ultimate Strength or Working Strength, whichever is active.

D - Maximum equivalent membrane stress (von Mises stress, constant energy of distortion stress) compared to Yield Strength, see theory (4), pg. 26 of Roark's. Maximum shear stress compared to ½ of Yield Strength (when internal pressure and thick walled formulas are used), see theory (2), pg. 26 of Roark's.

E - Maximum hoop stress compared to Ultimate Strength or Working Strength, whichever is active. Maximum shear stress compared to ½ of Ultimate Strength or ½ of Working Strength, whichever is active.

F - Maximum membrane stresses (radial and tangential) compared to Ultimate Strength (tensile) for both external and internal pressure loading. Average seat stress compared to Ultimate Strength (compressive) for external pressure loading.

G - Maximum membrane stresses (radial and tangential) and average seat stress compared to Yield Strength, see theory (1), pg. 26 of Roark's. Maximum shear stress compared to ½ of Yield Strength, see theory (3), pg. 26 of Roark's.

H - Maximum membrane stresses (radial and tangential) and average seat stress compared to Ultimate Strength or Working Strength, whichever is active. Maximum shear stress compared to ½ of Ultimate Strength or ½ of Working Strength, whichever is active.

I - Maximum hoop stress and average seat stress compared to Ultimate Strength (tensile) for internal pressure loading, or compared to Ultimate Strength (compressive) for external pressure loading.

J - Maximum equivalent membrane stress (von Mises stress, constant energy of distortion stress) and average seat stress compared to Yield Strength, see theory (4), pg. 26 of Roark's.

K - Maximum hoop stress and average seat stress compared to Ultimate Strength or Working Strength, whichever is active.

L - Maximum equivalent membrane stress (von Mises stress, constant energy of distortion stress) compared to Yield Strength, see theory (4), pg. 26 of Roark's. Average seat stress compared to Yield Strength. Maximum shear stress compared to ½ of Yield Strength (when internal pressure and thick walled formulas are used), see theory (2), pg. 26 of Roark's.

M - Maximum hoop stress and average seat stress compared to Ultimate Strength or Working Strength, whichever is active. Maximum shear stress compared to ½ of Ultimate Strength or ½ of Working Strength, whichever is active.

N - Maximum membrane stresses (radial and tangential) and average seat stress compared to Yield Strength, see theory (1), pg. 26 of Roark's.

O - Maximum membrane stresses (radial and tangential) and average seat stress compared to Ultimate Strength or Working Strength, whichever is active.

NOTE: Formula references can be accessed by clicking on Window on the menu bar of the Application Window and clicking on References or by entering Alt+W+R from the keyboard.

Conversions:

$$\text{FtSea} = \frac{-0.444 + \text{Sqr}(0.444^2 - 4 * (0.3 / 1000^2) * \text{-PsiVal})}{2 * (0.3 / 1000^2)}$$

Case "Ft (sea)" Conversion = 1

Case "Ft (fresh)" Conversion = 1 / 0.984

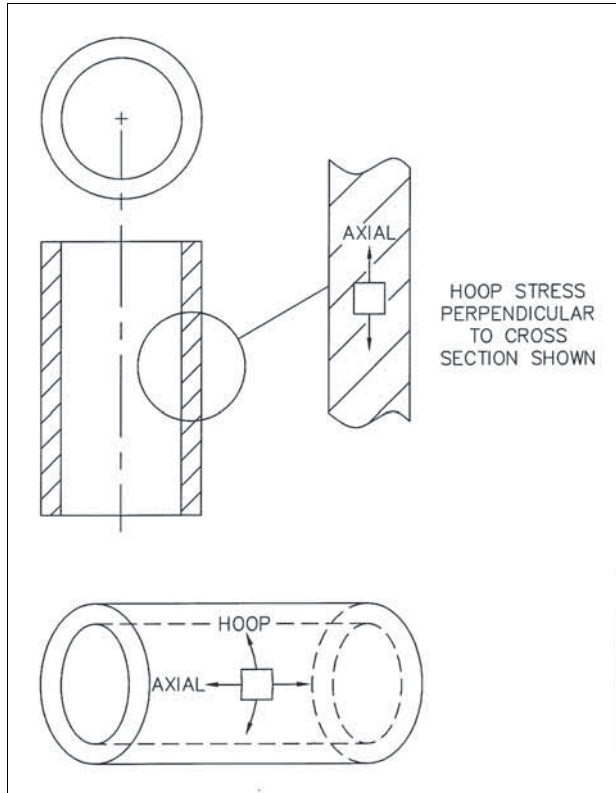
Case "m (sea)" Conversion = 0.3048

Case "m (fresh)" Conversion = 0.3048 / 0.984

APPENDIX D: PRESSURE VESSEL STRESSES

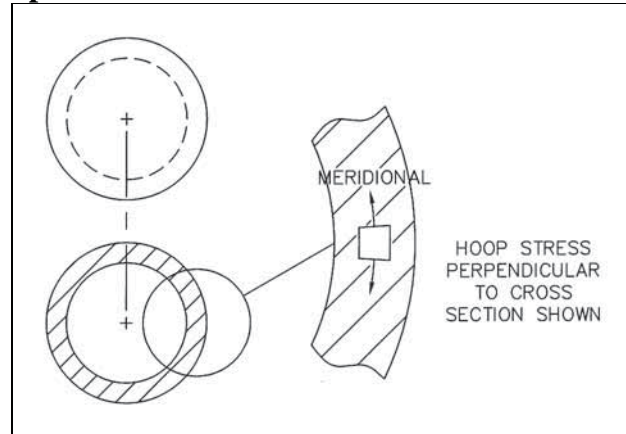
STRESSES

Tube:



- Axial stress = $\frac{1}{2}$ Hoop stress for thin wall equations for both external and internal pressure loading.
- Axial stress = $\frac{1}{2}$ Hoop stress at Tube I.D. for thick wall equations for external pressure loading.

Sphere:

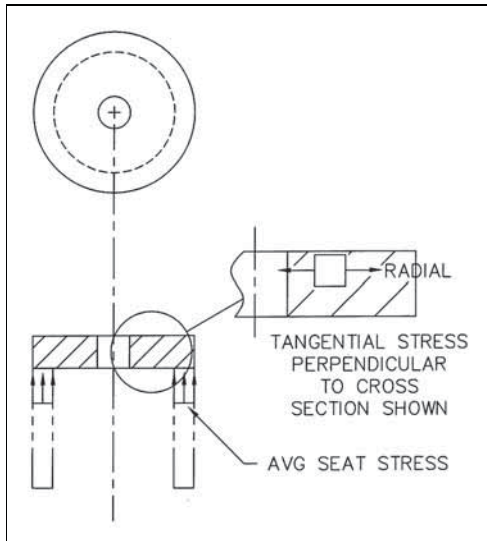


- Meridional stress = Hoop stress for both thin and thick equations and for both external and internal pressure loading.

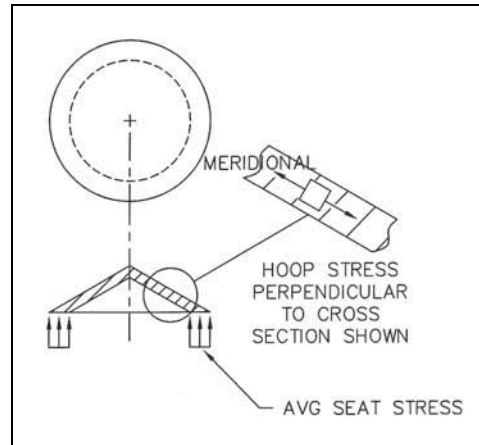
STRESSES (Continued)

ENDCAP

Flat Annular:

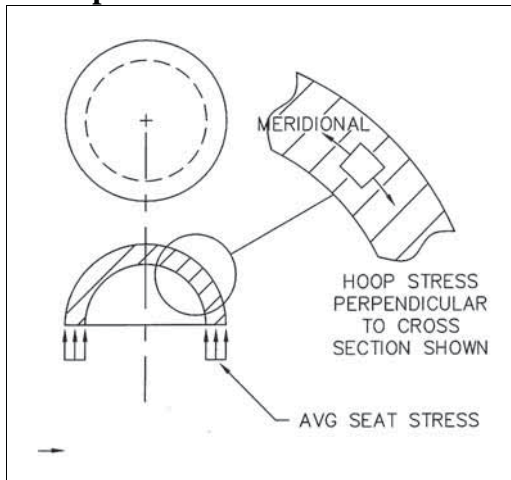


Conical:



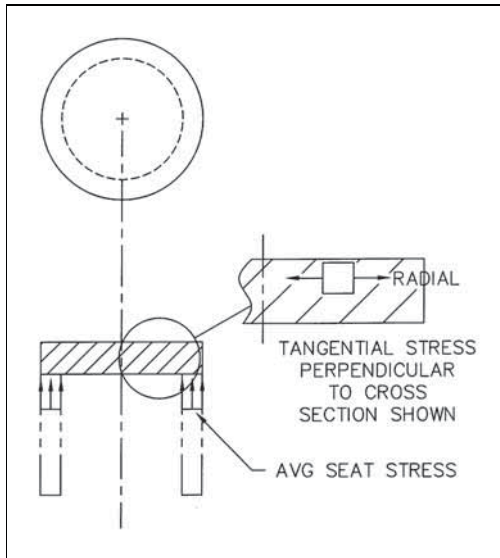
- Meridional stress = $\frac{1}{2}$ Hoop stress

Hemispherical:



- Meridional stress = Hoop stress for both thin and thick wall equations and for both external and internal pressure loading.

STRESSES (Continued)

ENDCAP**Flat Circular:**

- **Simply Supported Edge Restraint:** Maximum radial stress = maximum tangential stress and occurs at the inner and outer surfaces of the plate at the plate center line.
- **Fixed Edge Restraint:** Maximum radial stress occurs at the inner and outer surfaces of the plate at the plate free diameter. Maximum Tangential Stress: Occurs at the inner and outer surfaces of the plate at the plate center line. Peak maximum stress: Radial stress at the inner and outer surfaces of the plate at the plate free diameter.

APPENDIX E: PRESSURE VESSEL STRESS DISTRIBUTION

Once a pressure vessel design has been analyzed using Under Pressure, stress results are displayed numerically in the Analysis Dialog Box. Depending on the pressure vessel geometry that has been analyzed, different significant stress components are displayed in table format in the Analysis Dialog Box. In order to help the pressure vessel designer visualize the distribution of stress in a particular pressure vessel geometry (and thereby better understand the significance of the numerical stress results generated by a Under Pressure analysis), this appendix compares numerical stress results generated by Under Pressure with graphically displayed stress distribution results generated by a computer aided structural analysis using the finite element method.

Surface pressures on an enclosed vessel are resisted by internal forces that develop in the walls of the pressure vessel. The orientation and distribution of internal forces, known as stresses, that develop within the material of the pressure vessel wall are a function of the vessel geometry, material, and type of loading. Typically the material of the pressure vessel wall resists the stresses that develop without failure until some critical stress level is achieved. The critical stress level that causes failure is generally related to the strength of the material. Under Pressure provides the designer with the magnitude of the peak stresses as a function of the pressure loading and also provides the critical pressure at which failure of the vessel material is initiated. Generally, the peak stresses displayed in the Analysis Dialog Box of Under Pressure are localized to a specific region of the pressure vessel wall. This appendix aids in visualizing the location of the peak stresses provided by Under Pressure as well as understanding the complete distribution of stresses that develop for pressure vessel geometry's analyzed by the program.

The intent of this appendix is to aid the pressure vessel designer in visualizing stress distributions in common pressure vessel geometries as a supplement to Under Pressure. In some of the cases presented in this appendix, there is significant difference between the magnitude of stresses calculated by Under Pressure and the finite element method. The accuracy of the finite element method's results is depended upon the methods used to model the pressure vessel geometry, applied loads and boundary conditions. Factors such as the number of elements used (mesh density), type of elements employed and techniques used to represent applied loads and boundary conditions will effect the accuracy of the model.

For the purposes of this appendix, the following parameters shall be fixed:

- **Pressure Vessel Material is a Metal (approx. 6061-T6 Aluminum) with:**
 Young's Modulus = 10 Mpsi
 Poisson's Ratio = .3
 density = .1 lb/cu in
 yield strength = 35,000 psi

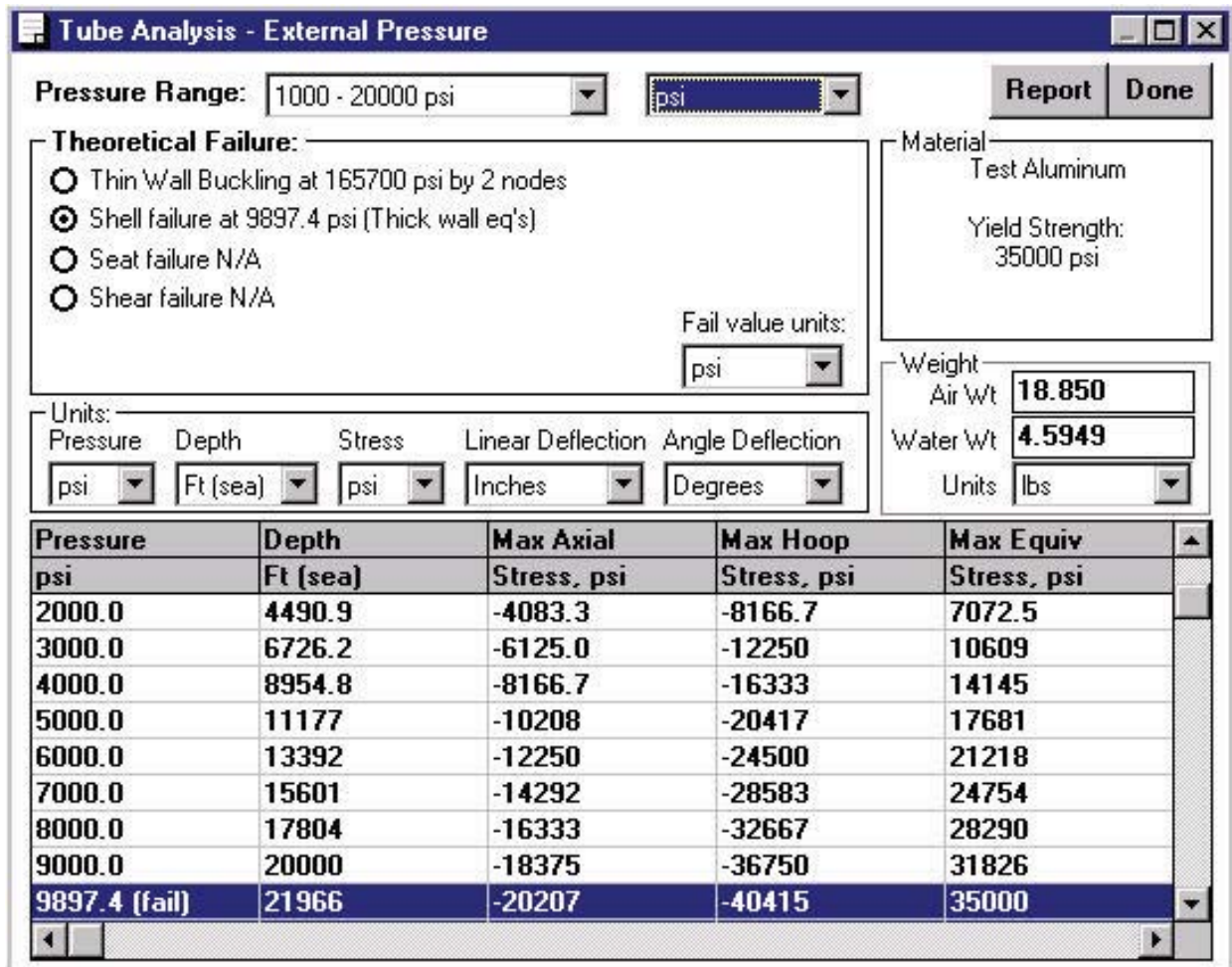
The screenshot shows a software window titled "Material Database". It contains several input fields and buttons. The "Main Category" is set to "Metals" and the "Sub-Category" is "Aluminum". The "Name" field contains "Test Aluminum". The "Yield Strength" is 35000 psi, "Young's Modulus" is 10 Mpsi, "Density" is 0.1 lb/cu in, and "Poisson's Ratio" is 0.3. A "Comments" text area contains the text "Test materials for user manual examples, similar to 60601-T6 AL.". On the right side, there are buttons for "Update Search", "Edit Record", "Add", "Delete", "Done", and "Cancel".

- Under Pressure stress results for Tube and Sphere Analysis are based on Thick Wall Equations
- FEA Stress results presented are based on an applied external pressure of 1000 psi

TUBE ANALYSIS:

Tube length = 10.00", Tube I.D. = 5.00", Wall thickness = 1.00", Tube O.D.=7.00"

Tube cross section with applied pressure and B.C.'s (etube1.bmp):

Under Pressure numerical stress results:

1. Maximum Axial Stress = -2041.7 psi
2. Maximum Hoop Stress = -4083.3 psi

Finite Element Method graphical stress results:

FEA model (element mesh, pressure loading, B.C.'s from etube1mod.bmp):

Axial stress contour (etube1axial.bmp):

Hoop stress contour (etube1hoop.bmp):

1. Maximum Axial Stress = -2042 psi
2. Maximum Hoop Stress = -4074 psi

Discussion:

Under Pressure (based on the formulas given in Appendix C) and the finite element model provide nearly identical results for both axial and hoop stresses. Appendix D defines the orientation of the axial and hoop stresses in the walls of the cylindrical tube. Both of these methods are based on the assumption the tube is capped by some type of end cap which provides a uniform axial pressure to the ends of the tube that is derived from the external pressure applied to the end cap. The axial pressure applied to the tube ends equals:

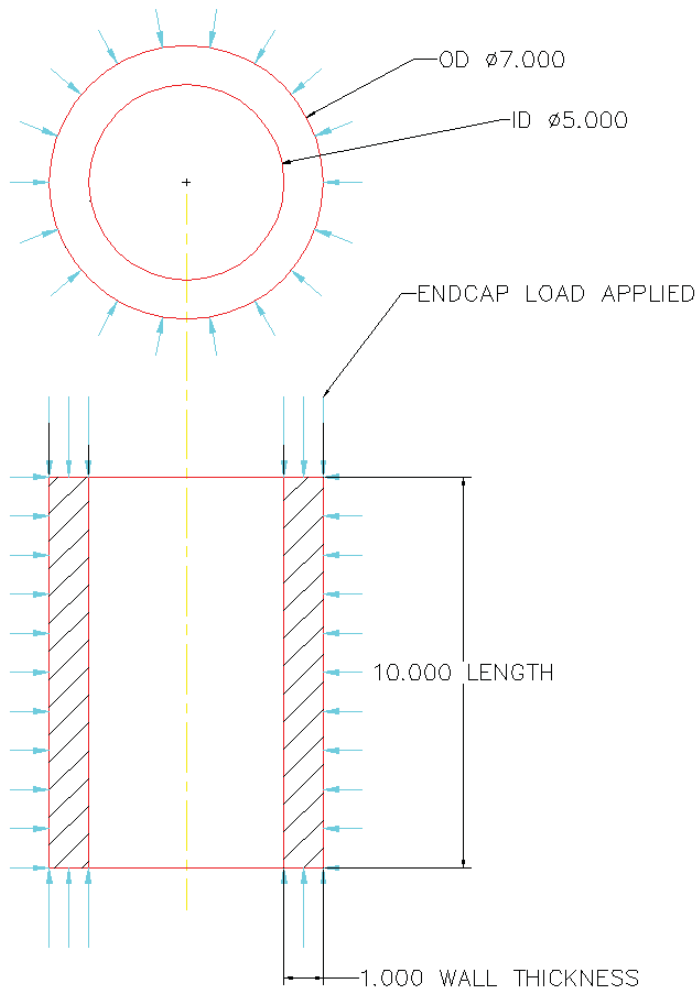
$$\frac{(\text{Total axial force on end cap})}{(\text{annular area of tube end})} = \frac{(P)(\pi)(\text{Tube O.D.})^2}{(\pi)((\text{Tube O.D.})^2 - (\text{Tube I.D.})^2)}$$

where P is the applied external pressure. For the specific case above the axial pressure = $(1000 \text{ psi})(\pi)(7 \text{ in.})^2 / (\pi)((7 \text{ in.})^2 - (5 \text{ in.})^2) = 2042 \text{ psi}$

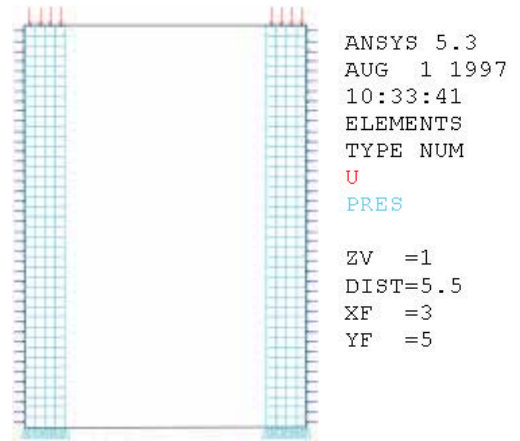
A two dimensional cross-section of the tube is modeled with the finite element method rather than constructing a three dimensional model of the tube. Since the geometry of the tube and the applied loads and boundary conditions are symmetric about the tube centerline, the resulting stresses in the tube are also symmetric about the tube centerline. All pressure vessel geometry's and applied pressures that are analyzed by Under Pressure are axis symmetric. For this type of symmetry, the stress distribution in the pressure vessel wall is the same for any cross-section taken through the centerline axis.

The finite element method involves subdividing the cross section of the tube into a "mesh" of finite elements. For this exercise, a mesh of four quadrilateral elements across the tube wall thickness by 40 elements along the tube length was selected. A uniform pressure of 1000 psi was applied along the outside diameter and an axial pressure of 2042 psi was applied to one end of the tube and resisted at the opposite end by an axial constraint (boundary condition).

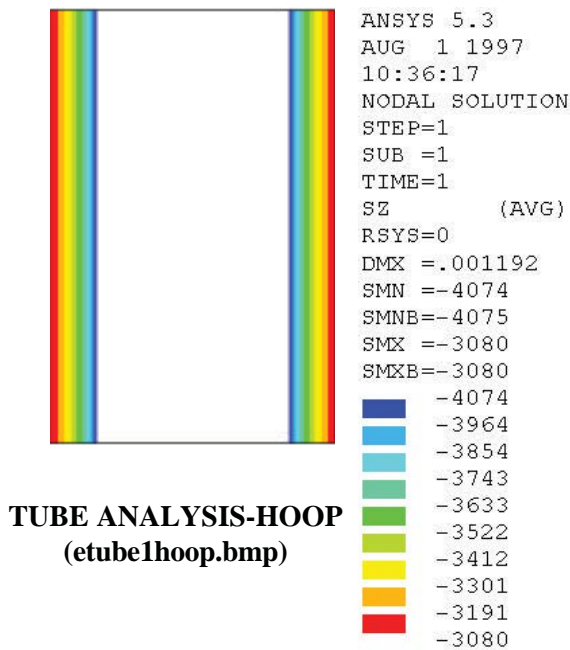
The finite element stress contours indicates that the compressive axial stress in the tube wall is uniform throughout the wall thickness and is equal in magnitude to the applied axial pressure of 2042 psi. The hoop stress contour indicates that the hoop stress varies from -3080 psi compressive stress at the tube outer diameter to a peak -4074 psi compressive stress at the tube inner diameter. Color contours are used to graphically display stress distribution in the tube. For this example, the entire range of stress has been subdivided into nine bands of color ranging from red (stresses from -3080 to -3191 psi) to blue (stresses from -3964 to -4074 psi). For this case, the peak hoop stress (greatest in magnitude) is represented by the blue contour along the tube inner diameter. While Under Pressure only provides a numerical value for the maximum stress in the tube, the finite element color contour provides the distribution of stress throughout the entire tube.



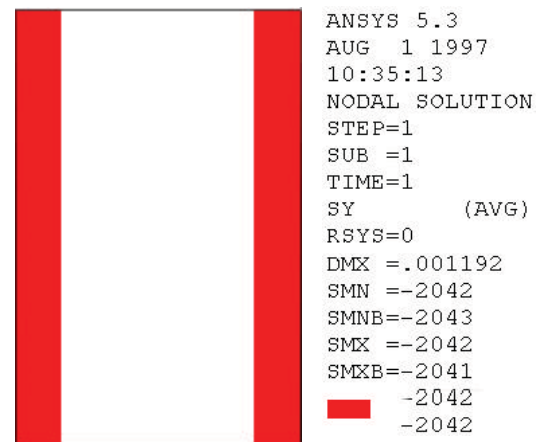
TUBE ANALYSIS-DWG
 (etube1.bmp)



TUBE ANALYSIS-MOD
 (etube1mod.bmp)



TUBE ANALYSIS-HOOP
 (etube1hoop.bmp)

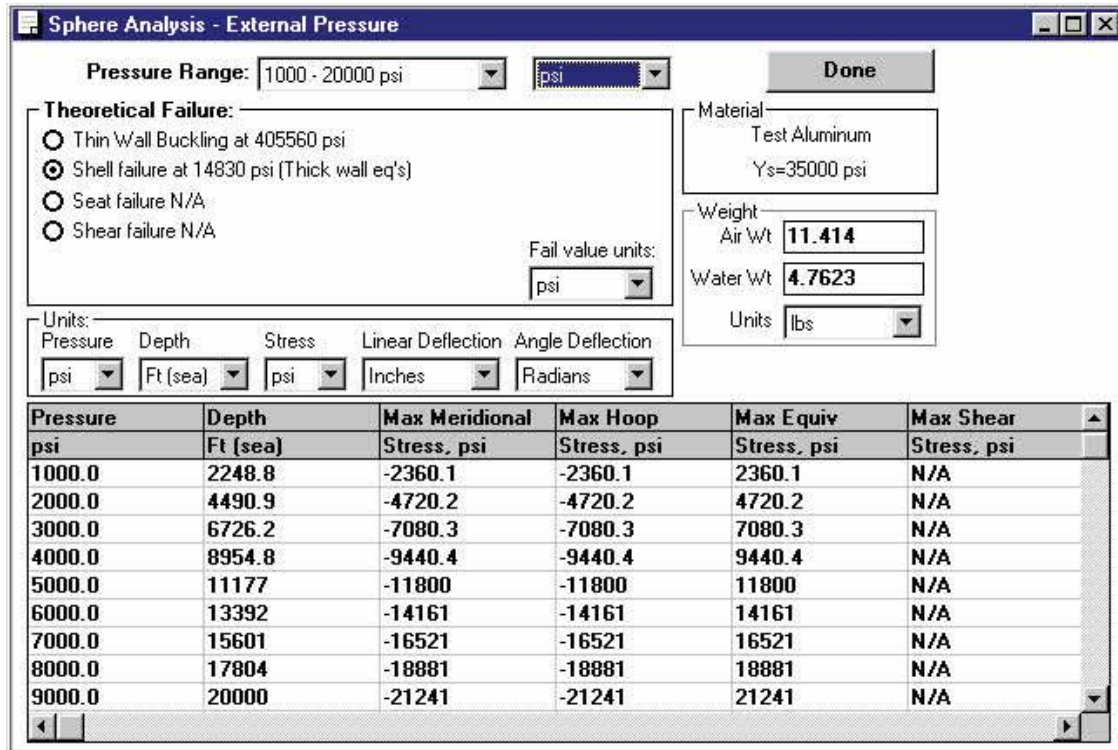


TUBE ANALYSIS-AXIAL
 (etube1axial.bmp)

SPHERE ANALYSIS:

Sphere I.D. = 5.00", Wall thickness = 1.00"

Sphere cross section with applied pressure (esphere1.bmp):

Under Pressure numerical stress results:

1. Maximum Meridional Stress = -2360.1 psi
2. Maximum Hoop Stress = -2360.1 psi

Finite Element Method graphical stress results:

FEA model (element mesh, pressure loading from esphere1mod.bmp):

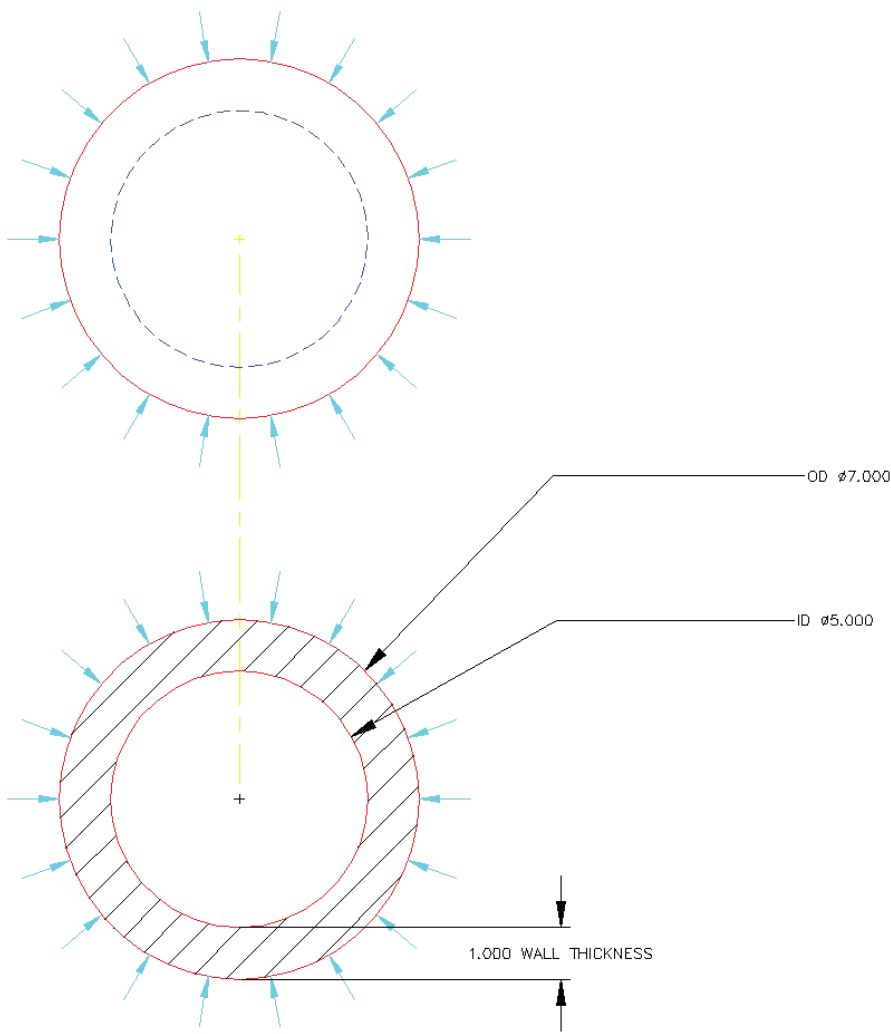
Meridional stress contour (esphere1mer.bmp):

Maximum Meridional Stress = -2354 psi

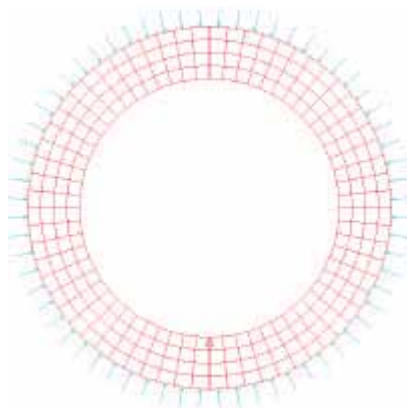
Discussion:

Under Pressure (based on the formulas given in Appendix C) and the sphere finite element model provide nearly identical results for peak meridional stress in the spherical housing. Appendix D defines the orientation of the meridional stresses in the walls of the spherical shell.

For this exercise, a mesh of four quadrilateral elements across the spherical shell wall thickness was selected for the finite element model. A uniform pressure of 1000 psi was applied along the outside radius. The finite element stress contours indicates that the compressive meridional stress in the spherical shell varies from 1858 psi compressive stress at the sphere outer surface to a peak -2354 psi compressive stress at the sphere inner surface.



SPHERE ANALYSIS-DWG
(esphere1.bmp)



```

ANSYS 5.3
AUG 1 1997
10:39:38
ELEMENTS
TYPE NUM
U
PRES
ZV =1
DIST=3.85
XF =1.75
    
```

SPHERE ANALYSIS-MOD
(esphere1mod.bmp)



```

ANSYS 5.3
AUG 1 1997
10:38:12
NODAL SOLUTION
STEP=1
SUB =1
TIME=1
SZ          (AVG)
RSYS=0
DMX =.826E-03
SMN =-2354
SMNB=-2356
SMX =-1858
SMXB=-1857
    
```

| | |
|--------------|-------|
| Blue | -2354 |
| Light Blue | -2299 |
| Medium Blue | -2244 |
| Green-Blue | -2189 |
| Green | -2134 |
| Yellow-Green | -2078 |
| Yellow | -2023 |
| Orange | -1968 |
| Red-Orange | -1913 |
| Red | -1858 |

SPHERE ANALYSIS-MER
(esphere1mer.bmp)

FLAT ANNULAR ENDCAP ANALYSIS:

For the purposes of the flat annular endcap cases presented below the following assumptions are used:

- All flat annular endcap case results from Under Pressure are based on enabling **Uniform Load** and **Line Load** Options in the Flat Annular Endcap Geometry Dialog Box
- All flat annular endcap case results from Under Pressure are based on using **50** radial increments for calculating radial and tangential stresses

Case 1: Plate Outside Diameter = 6.00", Plate Free Diameter = 5.00", Plate thickness = .625", Center Hole Diameter = 1.00", Fixed Outer Edge Restraint, Free Inner (Center Hole) Edge Restraint

NOTE: The elastic formulas used in UP use only the free diameter. The outside diameter is used for weight and seat stress calculations only. As the plate diameter is increased relative to the free diameter the results should be more conservative.

Plate cross section with applied pressure, B.C.'s (eann1.bmp):

Under Pressure numerical stress results:

Flat Annular Endcap Analysis - External Pressure

Pressure Range: 1000 - 20000 psi psi Graph Report Done

Theoretical Failure:

- Radial moment failure at 2.7461 Ksi (Dia. = 5.0000 inches)
- Tangential moment failure at 2.6153 Ksi (Dia. = 1.0000 inches)
- Shear stress failure at 8.7500 Ksi (Dia. = 5.0000 inches)
- Seat failure at 10.694 Ksi

Fail value units: Ksi

Table eval dia., De (inches): 1.0000 ReCalc Ksi

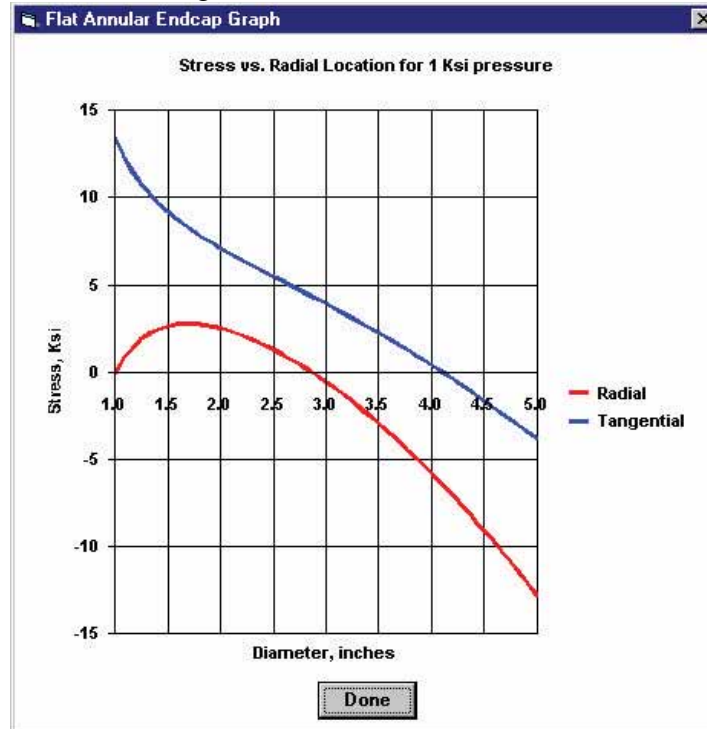
Material: Test Aluminum
Yield Strength: 35000 psi

Weight: Air Wt 1.7181
Water Wt 1.0817
Units lbs

Units:
Pressure: Ksi Depth: Ft (sea) Stress: Ksi Linear Deflection: Inches Angle Deflection: Radians

| Pressure Ksi | Depth Ft (sea) | De Radial Stress, Ksi | De Tangential Stress, Ksi | De Shear Stress, Ksi |
|-----------------|-------------------|--------------------------|------------------------------|-------------------------|
| 1.0000 | 2248.8 | 0.0000 | 13.383 | 0.0000 |
| 2.0000 | 4490.9 | 0.0000 | 26.766 | 0.0000 |
| 2.6153 (fail) | 5867.0 | 0.0000 | 35.000 | 0.0000 |
| 3.0000 (fail) | 6726.2 | 0.0000 | 40.148 | 0.0000 |
| 4.0000 (fail) | 8954.8 | 0.0000 | 53.531 | 0.0000 |
| 5.0000 (fail) | 11177 | 0.0000 | 66.914 | 0.0000 |
| 6.0000 (fail) | 13392 | 0.0000 | 80.297 | 0.0000 |
| 7.0000 (fail) | 15601 | 0.0000 | 93.680 | 0.0000 |
| 8.0000 (fail) | 17804 | 0.0000 | 107.06 | 0.0000 |

Clicking on **Graph** gives the following:



1. Maximum Radial Stress = 12,746 psi , Location = 5.000" diameter (plate free diameter)
2. Maximum Tangential Stress = 13,383 psi, Location = 1.000" diameter (hole diameter)

Finite Element Method graphical stress results:

FEA model (element mesh, pressure loading, B.C.'s from eann1mod.bmp):

Radial stress contour (eann1rad.bmp):

Tangential stress contour (eann1tan.bmp):

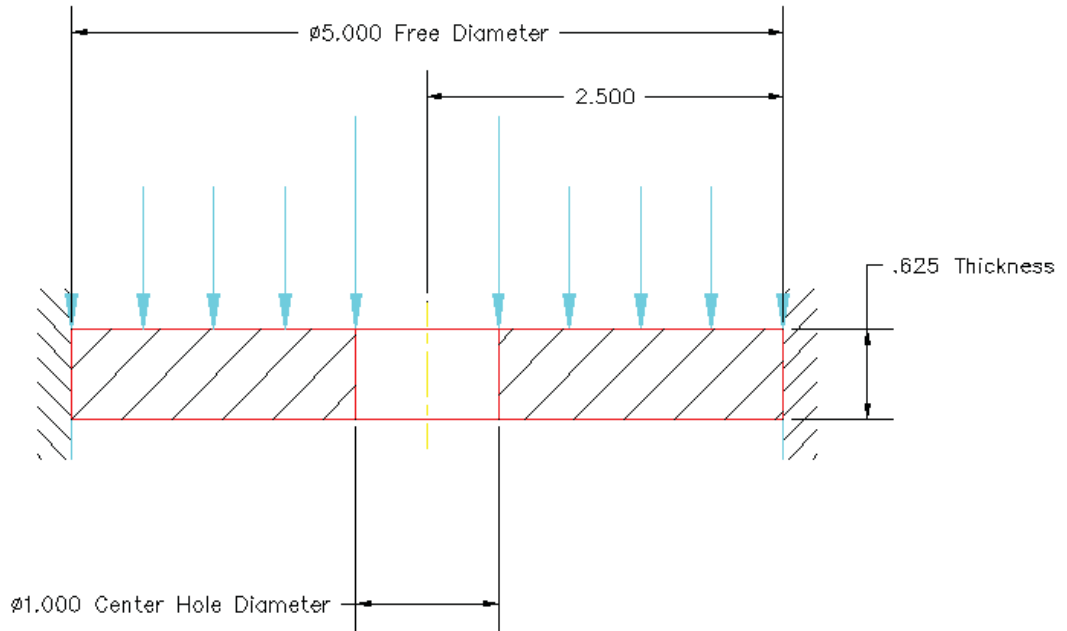
1. Maximum Radial Stress = 12,013 psi, Location = 5.000" diameter
2. Maximum Tangential Stress = 13,398 psi, Location = 1.000" diameter

Discussion:

Under Pressure (based on the formulas given in Appendix C) and the flat annular endcap finite element model provide similar results for peak radial and tangential stresses as well as radial and tangential stress distribution in the endcap. Appendix D defines the orientation of the radial and tangential stresses in the endcap.

For this exercise, a mesh of four quadrilateral elements across the endcap thickness was selected for the finite element model.

The finite element stress contours indicates that the peak tangential stress in the endcap occurs at the inner and outer flat surfaces of the plate at the hole diameter. The stress contours indicate that the mid-thickness of the plate has zero stress (neutral axis). The stress distribution on the outer (upper) half thickness of the plate is compressive due to its concave deflection during pressure loading. The stress distribution on the inner (lower) half thickness of the plate is tensile due to its convex displacement during pressure loading. The distribution of the compressive and tensile stresses in the plate are symmetric and the plate's neutral axis. The peak radial stress occurs at the inner and outer flat surfaces of the plate at the plate free diameter.



FLAT ANNULAR ENDCAP ANALYSIS-DWG
(eann1.bmp)



FLAT ANNULAR ENDCAP ANALYSIS-MOD
(eann1mod.bmp)

```
ANSYS 5.3
AUG 1 1997
10:45:28
ELEMENTS
TYPE NUM
U
F
CP
PRES
```

```
EV -1
*DIST=1.509
*XF =1.047
*YF =-.3125
```

```
ANSYS 5.3
AUG 1 1997
10:43:01
NODAL SOLUTION
STEP=1
SUB =1
TIME=1
SX (AVG)
RSYS=0
DMX =.004447
SMN --12013
SMNB=-22675
SMX =12013
SMXB=22675
```



FLAT ANNULAR ENDCAP ANALYSIS-RAD
(eann1rad.bmp)

| | |
|--------------|--------|
| Blue | -12013 |
| Light Blue | -9343 |
| Light Green | -6674 |
| Green | -4004 |
| Yellow-Green | -1335 |
| Yellow | 1335 |
| Orange | 4004 |
| Red-Orange | 6674 |
| Red | 9343 |
| Dark Red | 12013 |



FLAT ANNULAR ENDCAP ANALYSIS-TAN
(eann1tan.bmp)

```
ANSYS 5.3
AUG 1 1997
10:43:56
NODAL SOLUTION
STEP=1
SUB =1
TIME=1
SZ (AVG)
RSYS=0
DMX =.004447
SMN --13398
SMNB=-13956
SMX =13398
SMXB=13956
```

| | |
|--------------|--------|
| Blue | -13398 |
| Light Blue | -10421 |
| Light Green | -7443 |
| Green | -4466 |
| Yellow-Green | -1489 |
| Yellow | 1489 |
| Orange | 4466 |
| Red-Orange | 7443 |
| Red | 10421 |
| Dark Red | 13398 |

Case 2: Plate Outside Diameter = 6.00”, Plate Free Diameter = 5.00”, Plate thickness = .625”, Center Hole Diameter = 1.00”, Fixed Outer Edge Restraint, Guided Inner (Center Hole) Edge Restraint

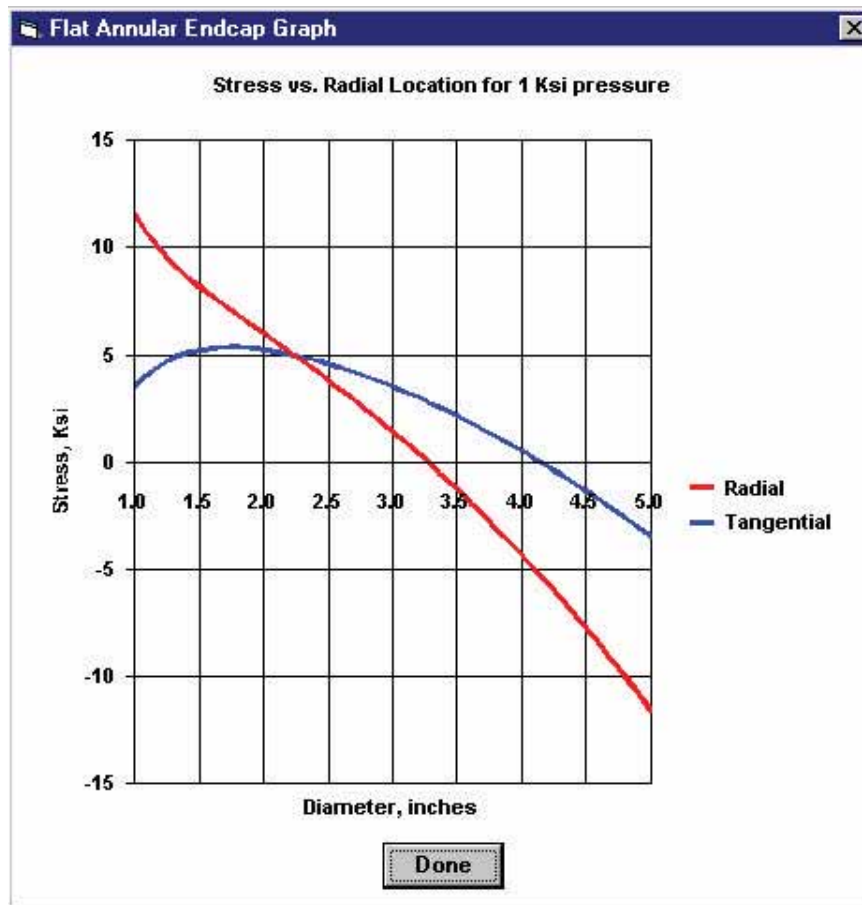
Plate cross section with applied pressure, B.C.’s (eann2.bmp):

Under Pressure numerical stress results:

The screenshot shows the 'Flat Annular Endcap Analysis - External Pressure' software interface. It includes a 'Pressure Range' of 1000 - 20000 psi, a 'Theoretical Failure' section with four radio buttons (Radial moment failure at 3.0382 Ksi is selected), and a table of numerical stress results. The table has columns for Pressure (Ksi), Depth (Ft (sea)), De Radial Stress (Ksi), De Tangential Stress (Ksi), and De Shear Stress (Ksi). The results show that for pressures from 3.0382 Ksi to 8.0000 Ksi, the radial stress reaches a failure value of -35.000 Ksi.

| Pressure Ksi | Depth Ft (sea) | De Radial Stress, Ksi | De Tangential Stress, Ksi | De Shear Stress, Ksi |
|---------------|----------------|-----------------------|---------------------------|----------------------|
| 1.0000 | 2248.8 | -11.520 | -3.4560 | -2.0000 |
| 2.0000 | 4490.9 | -23.040 | -6.9120 | -4.0000 |
| 3.0000 | 6726.2 | -34.560 | -10.368 | -6.0000 |
| 3.0382 (fail) | 6811.4 | -35.000 | -10.500 | -6.0764 |
| 4.0000 (fail) | 8954.8 | -46.080 | -13.824 | -8.0000 |
| 5.0000 (fail) | 11177 | -57.600 | -17.280 | -10.000 |
| 6.0000 (fail) | 13392 | -69.120 | -20.736 | -12.000 |
| 7.0000 (fail) | 15601 | -80.640 | -24.192 | -14.000 |
| 8.0000 (fail) | 17804 | -92.160 | -27.648 | -16.000 |

Clicking on **Graph** gives the following:



1. Maximum Radial Stress = 11,520 psi, Location = 1.000" diameter (hole diameter)
2. Maximum Tangential Stress = 5344 psi , Location = 1.735" diameter

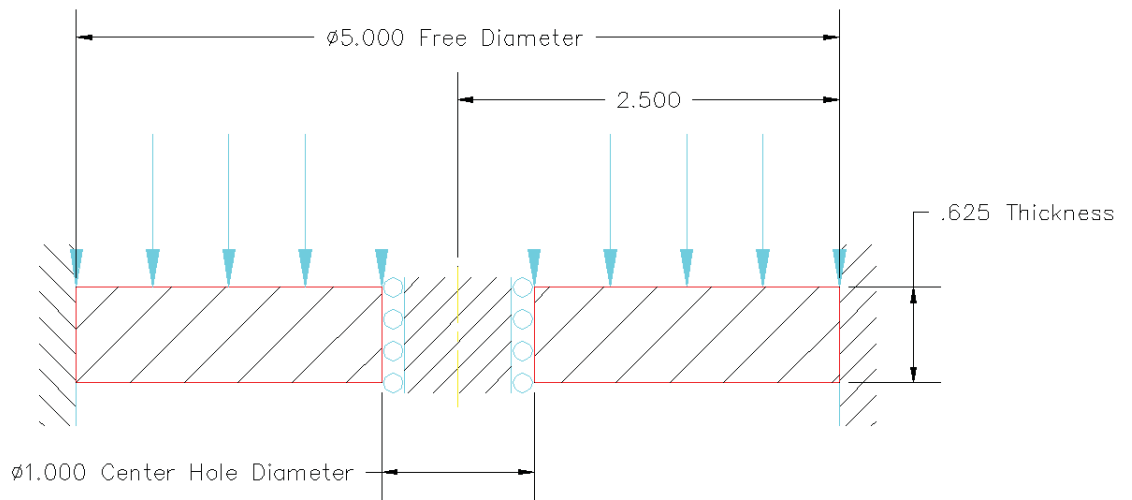
Finite Element Method graphical stress results:

FEA model (element mesh, pressure loading, B.C.'s from eann2mod.bmp)

Radial stress contour (eann2rad.bmp):

Tangential stress contour (eann2tan.bmp)

1. Maximum Radial Stress = 10,965 psi , Location = 1.000" diameter
2. Maximum Tangential Stress = 7256 psi, Location (see stress contour)



**FLAT ANNULAR ENDCAP ANALYSIS 2-DWG
(eann2.bmp)**

```
ANSYS 5.3
AUG 1 1997
10:49:05
ELEMENTS
TYPE NUM
U
F
CP
PRES
```



**FLAT ANNULAR ENDCAP ANALYSIS 2-MOD
(eann2mod.bmp)**

```
ZV =1
*DIST=1.509
*XF =1.047
*YF =.3125
Z-BUFFER
```



**FLAT ANNULAR ENDCAP ANALYSIS 2-RAD
(eann2rad.bmp)**

```
ANSYS 5.3
AUG 1 1997
10:50:58
NODAL SOLUTION
STEP=1
SUB =1
TIME=1
SX (AVG)
RSYS=0
DMX =.003337
SMN =-10965
SMNB=-20700
SMX =10965
SMXB=20700
```

| | |
|--------------|--------|
| Blue | -10965 |
| Light Blue | -8529 |
| Light Green | -6092 |
| Green | -3655 |
| Yellow-Green | -1218 |
| Yellow | 1218 |
| Orange | 3655 |
| Red-Orange | 6092 |
| Red | 8529 |
| Dark Red | 10965 |

```
ANSYS 5.3
AUG 1 1997
10:52:32
NODAL SOLUTION
STEP=1
SUB =1
TIME=1
SZ (AVG)
RSYS=0
DMX =.003337
SMN =-7256
SMNB=-13930
SMX =7256
SMXB=13930
```

| | |
|--------------|----------|
| Dark Blue | -7256 |
| Blue | -5643 |
| Light Blue | -4031 |
| Light Green | -2419 |
| Green | -806.178 |
| Yellow-Green | 806.178 |
| Yellow | 2419 |
| Orange | 4031 |
| Red-Orange | 5643 |
| Red | 7256 |

**FLAT ANNULAR ENDCAP ANALYSIS 2-TAN
(eann2tan.bmp)**

Case 3: Plate Outside Diameter = 6.00", Plate Free Diameter = 5.00", Plate thickness = .625", Center Hole Diameter = 1.00", Fixed Outer Edge Restraint, Simply Supported Inner (Center Hole) Edge Restraint

Plate cross section with applied pressure, B.C.'s (eann3.bmp):

Under Pressure numerical stress results:

Flat Annular Endcap Analysis - External Pressure

Pressure Range: 1000 - 20000 psi psi **Graph** **Report** **Done**

Theoretical Failure:

- Radial moment failure at 6.5012 Ksi (Dia. = 5.0000 inches)
- Tangential moment failure at 7.5336 Ksi (Dia. = 1.0000 inches)
- Shear stress failure at 6.3680 Ksi (Dia. = 1.0000 inches)
- Seat failure at 10.694 Ksi

Fail value units: Ksi

Table eval dia., De (inches): 1.0000 ReCalc Ksi

Units:

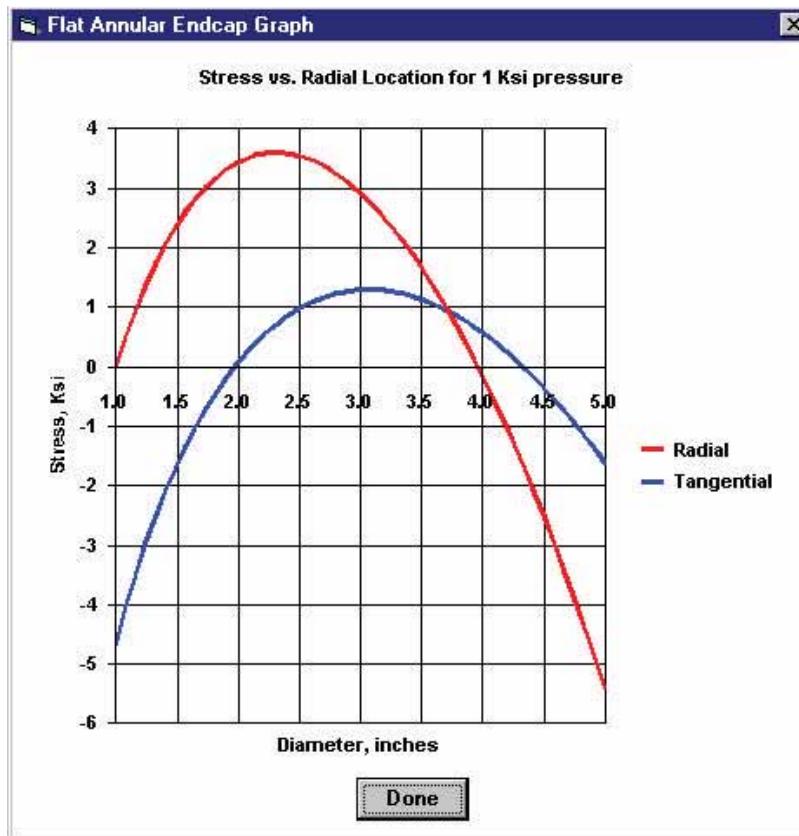
| | | | | |
|----------|----------|--------|-------------------|------------------|
| Pressure | Depth | Stress | Linear Deflection | Angle Deflection |
| Ksi | Ft (sea) | Ksi | Inches | Radians |

Material: Test Aluminum
Yield Strength: 35000 psi

Weight:
Air Wt: 1.7181
Water Wt: 1.0817
Units: lbs

| Pressure | Depth | De Radial | De Tangential | De Shear |
|---------------|----------|-------------|---------------|-------------|
| Ksi | Ft (sea) | Stress, Ksi | Stress, Ksi | Stress, Ksi |
| 1.0000 | 2248.8 | 0.0000 | -4.6459 | 2.7481 |
| 2.0000 | 4490.9 | 0.0000 | -9.2918 | 5.4963 |
| 3.0000 | 6726.2 | 0.0000 | -13.938 | 8.2444 |
| 4.0000 | 8954.8 | 0.0000 | -18.584 | 10.993 |
| 5.0000 | 11177 | 0.0000 | -23.229 | 13.741 |
| 6.0000 | 13392 | 0.0000 | -27.875 | 16.489 |
| 6.3680 (fail) | 14206 | 0.0000 | -29.585 | 17.500 |
| 7.0000 (fail) | 15601 | 0.0000 | -32.521 | 19.237 |
| 8.0000 (fail) | 17804 | 0.0000 | -37.167 | 21.985 |

Clicking on **Graph** gives the following:



1. Maximum Radial Stress = -5384 psi, Location = 5.000" diameter (plate free diameter)
2. Maximum Tangential Stress = -4646 psi , Location = 1.000" diameter (hole diameter)

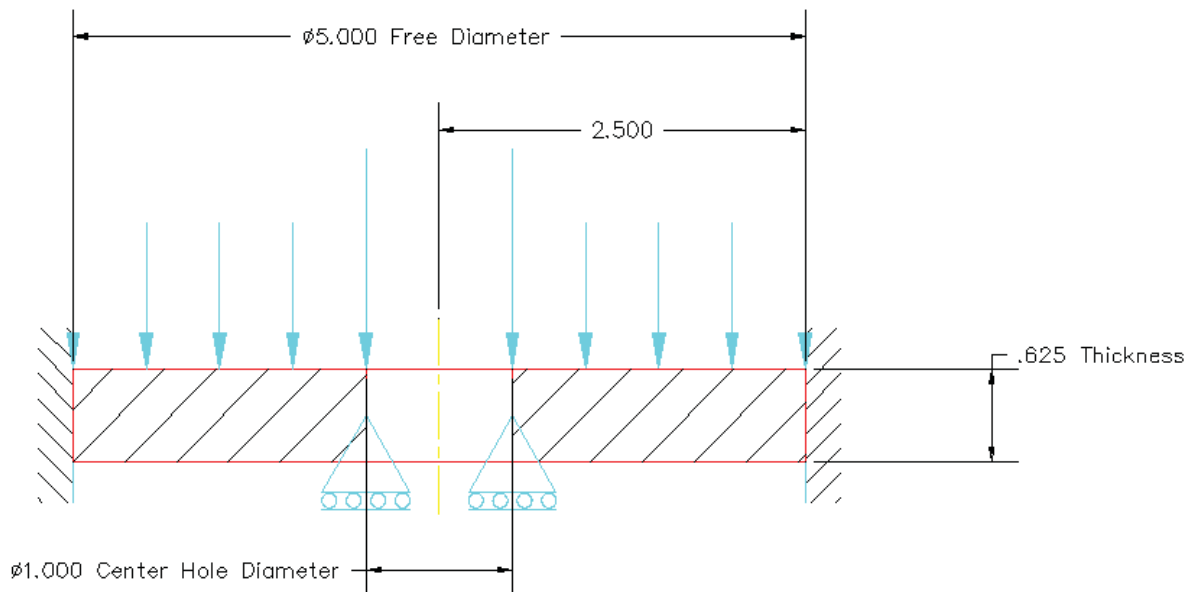
Finite Element Method graphical stress results:

FEA model (element mesh, pressure loading, B.C.'s from eann3mod.bmp):

Radial stress contour (eann3rad.bmp):

Tangential stress contour (eann3tan.bmp):

1. Maximum Radial Stress = -4644 psi, Location = 5.000" diameter
2. Maximum Tangential Stress = -4809 psi, Location = 1.000" diameter



**FLAT ANNULAR ENDCAP ANALYSIS 3-DWG
(eann3.bmp)**



ANSYS 5.3
AUG 1 1997
10:55:21
ELEMENTS
TYPE NUM
U
CP
PRES

**FLAT ANNULAR ENDCAP ANALYSIS 3-MOD
(eann3mod.bmp)**



ZV =1
*DIST=1.509
*XF =1.047
*YF =.3125

ANSYS 5.3
AUG 1 1997
10:57:30
NODAL SOLUTION
STEP=1
SUB =1
TIME=1
SX (AVG)
RSYS=0

**FLAT ANNULAR ENDCAP ANALYSIS 3-RAD
(eann3rad.bmp)**



ANSYS 5.3
AUG 1 1997
11:00:32
NODAL SOLUTION
STEP=1
SUB =1
TIME=1
SZ (AVG)
RSYS=0
DMX =.771E-03
SMN =-4809
SMNB=-16307
SMX =4809
SMXB=16307

DMX =.771E-03
SMN =-4644
SMNB=-16307
SMX =4644
SMXB=16307



**FLAT ANNULAR ENDCAP ANALYSIS 3-TAN
(eann3tan.bmp)**



Case 4: Plate Outside Diameter = 6.00”, Plate Free Diameter = 5.00”, Plate thickness = .625”, Center Hole Diameter = 1.00”, Fixed Outer Edge Restraint, Fixed Inner (Center Hole) Edge Restraint

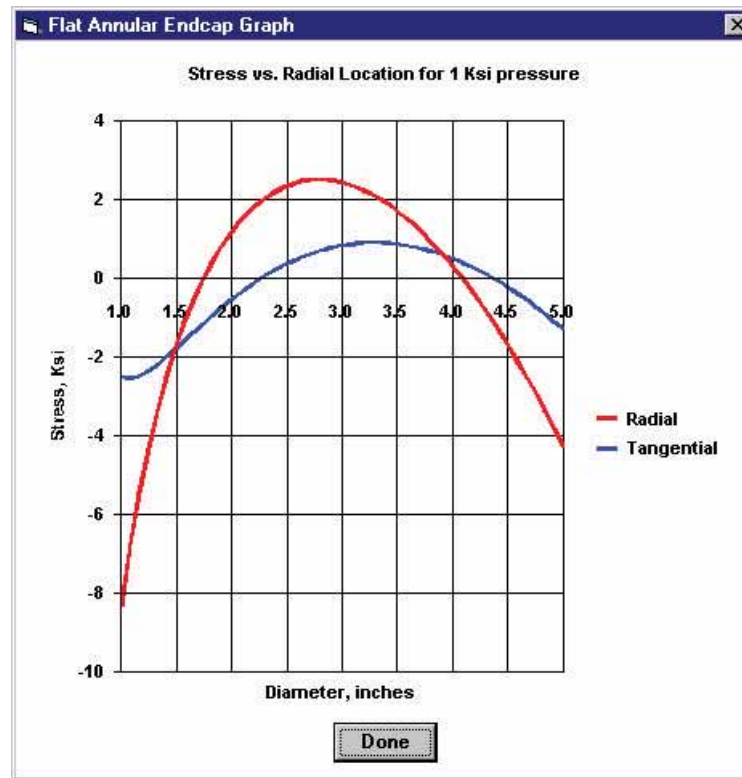
Plate cross section with applied pressure, B.C.’s (eann4.bmp):

Under Pressure numerical stress results:

The screenshot shows the 'Flat Annular Endcap Analysis - External Pressure' software window. It includes a 'Pressure Range' of 1000 - 20000 psi, 'Theoretical Failure' options (Radial moment failure at 4.2272 Ksi is selected), material properties for Test Aluminum (Yield Strength: 35000 psi), and a table of numerical stress results.

| Pressure Ksi | Depth Ft (sea) | De Radial Stress, Ksi | De Tangential Stress, Ksi | De Shear Stress, Ksi |
|---------------|----------------|-----------------------|---------------------------|----------------------|
| 1.0000 | 2248.8 | -8.2797 | -2.4839 | 3.5061 |
| 2.0000 | 4490.9 | -16.559 | -4.9678 | 7.0122 |
| 3.0000 | 6726.2 | -24.839 | -7.4517 | 10.518 |
| 4.0000 | 8954.8 | -33.119 | -9.9356 | 14.024 |
| 4.2272 (fail) | 9460.3 | -35.000 | -10.500 | 14.821 |
| 5.0000 (fail) | 11177 | -41.398 | -12.419 | 17.531 |
| 6.0000 (fail) | 13392 | -49.678 | -14.903 | 21.037 |
| 7.0000 (fail) | 15601 | -57.958 | -17.387 | 24.543 |
| 8.0000 (fail) | 17804 | -66.237 | -19.871 | 28.049 |

Clicking on **Graph** gives the following:



1. Maximum Radial Stress = -8280 psi, Location = 1.000" diameter (hole diameter)
2. Maximum Tangential Stress = -2527 psi , Location = 1.0816" diameter

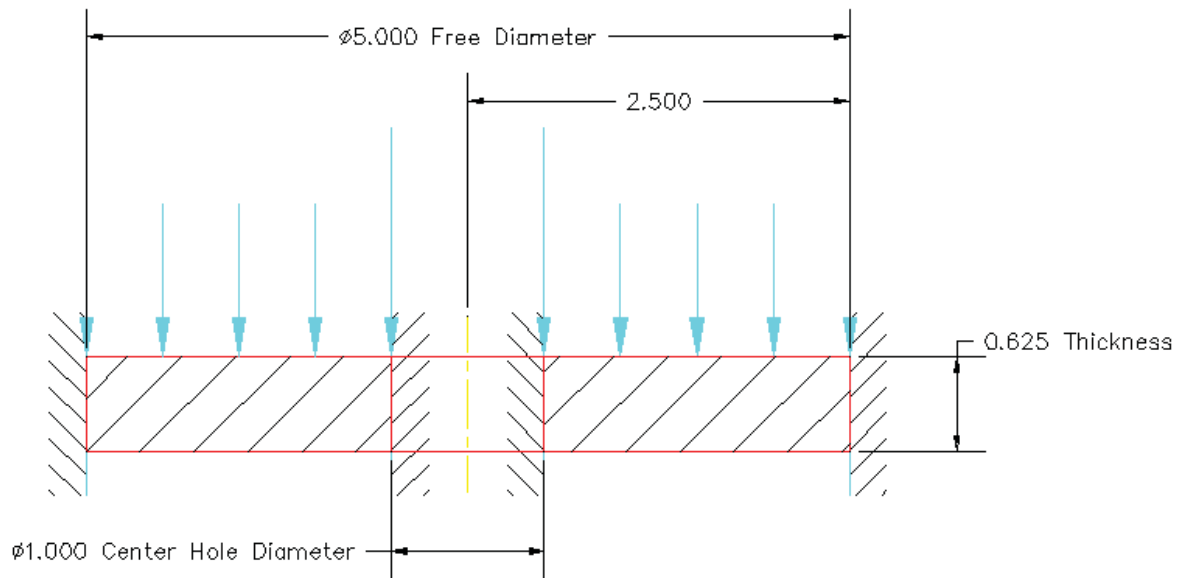
Finite Element Method graphical stress results:

FEA model (element mesh, pressure loading, B.C.'s from eann4mod.bmp)

Radial stress contour (eann4rad.bmp):

Tangential stress contour (eann4tan.bmp):

1. Maximum Radial Stress = -5575 psi, Location = 1.000" diameter
2. Maximum Tangential Stress = -3040 psi, Location (see stress contour)



**FLAT ANNULAR ENDCAP ANALYSIS 4-DWG
(eann4.bmp)**



**FLAT ANNULAR ENDCAP ANALYSIS 4-MOD
(eann4mod.bmp)**

ANSYS 5.3
AUG 1 1997
11:02:17
ELEMENTS
TYPE NUM
U
CP
PRES

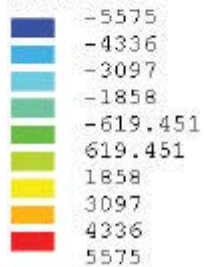
ZV =1
*DIST=1.509
*XF =1.047
*YF =.3125

ANSYS 5.3
AUG 1 1997
11:03:38
NODAL SOLUTION
STEP=1
SUB =1
TIME=1

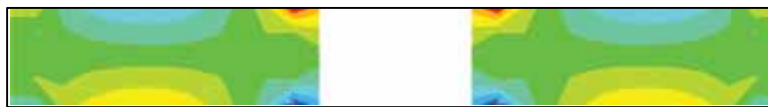


**FLAT ANNULAR ENDCAP ANALYSIS RAD
(eann4rad.bmp)**

SX (AVG)
RSYS=0
DMX =.682E-03
SMN =-5575
SMNB=-18861
SMX =5575
SMXB=18861

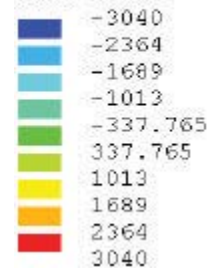


ANSYS 5.3
AUG 1 1997
11:07:11
NODAL SOLUTION
STEP=1
SUB =1
TIME=1



**FLAT ANNULAR ENDCAP ANALYSIS TAN
(eann4tan.bmp)**

SZ (AVG)
RSYS=0
DMX =.682E-03
SMN =-3040
SMNB=-18861
SMX =3040
SMXB=18861



Case 5: Plate Outside Diameter = 6.00”, Plate Free Diameter = 5.00”, Plate thickness = .625”, Center Hole Diameter = 1.00”, Simply Supported Outer Edge Restraint, Free Inner (Center Hole) Edge Restraint

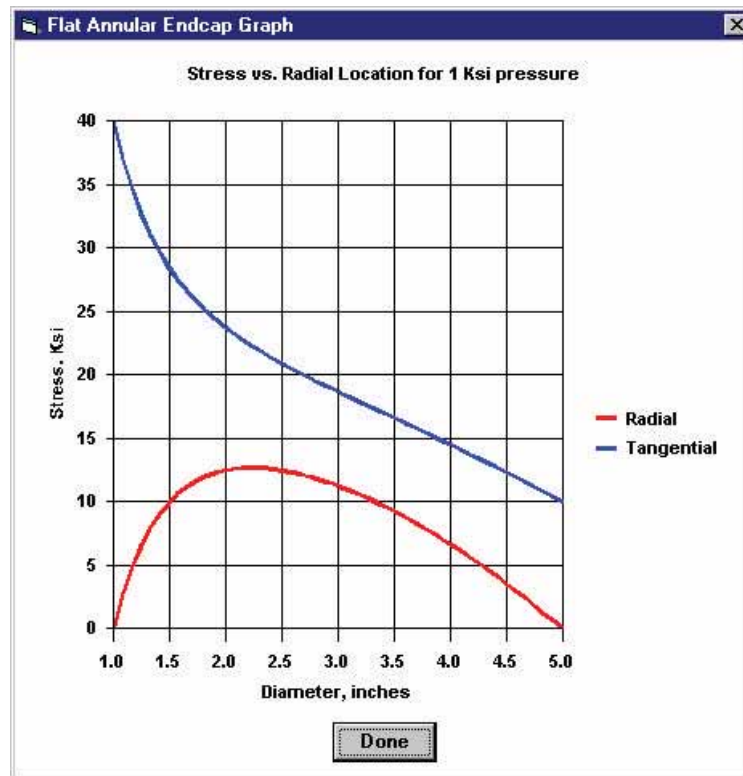
Plate cross section with applied pressure, B.C.’s: (eann5.bmp):

Under Pressure numerical stress results:

The screenshot shows the 'Flat Annular Endcap Analysis - External Pressure' window. The 'Pressure Range' is set to 10000 - 200000 psi. Under 'Theoretical Failure', 'Tangential moment failure at 876.40 psi (Dia. = 1.0000 inches)' is selected. Material is 'Test Aluminum' with a 'Yield Strength: 35000 psi'. Weight values are 'Air Wt 1.7181' and 'Water Wt 1.0817' in 'lbs' units. The table below shows results for pressures from 10000 to 90000 psi, all marked as failures.

| Pressure | Depth | De Radial | De Tangential | De Shear |
|--------------|----------|-------------|---------------|-------------|
| psi | Ft (sea) | Stress, psi | Stress, psi | Stress, psi |
| 10000 (fail) | 22190 | 0.0000 | 399360 | 0.0000 |
| 20000 (fail) | 43752 | 0.0000 | 798720 | 0.0000 |
| 30000 (fail) | 64736 | 0.0000 | 1198100 | 0.0000 |
| 40000 (fail) | 85187 | 0.0000 | 1597400 | 0.0000 |
| 50000 (fail) | 105140 | 0.0000 | 1996800 | 0.0000 |
| 60000 (fail) | 124640 | 0.0000 | 2396200 | 0.0000 |
| 70000 (fail) | 143700 | 0.0000 | 2795500 | 0.0000 |
| 80000 (fail) | 162370 | 0.0000 | 3194900 | 0.0000 |
| 90000 (fail) | 180650 | 0.0000 | 3594200 | 0.0000 |

Clicking on **Graph** gives the following:



1. Maximum Radial Stress = 12,672 psi , Location = 2.225" diameter
2. Maximum Tangential Stress = 39,936 psi , Location = 1.000" diameter

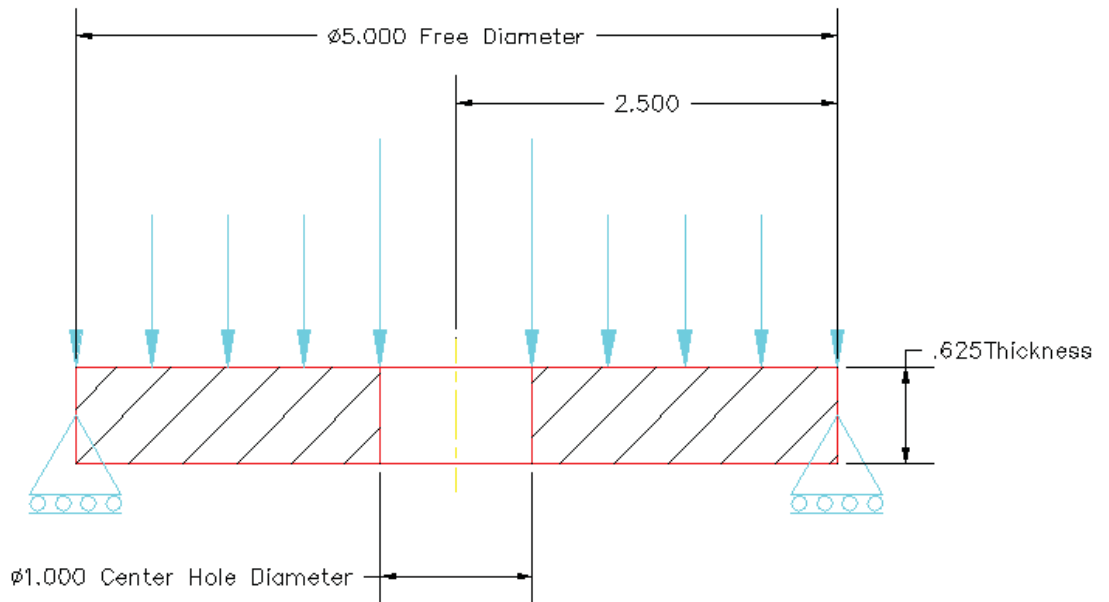
Finite Element Method graphical stress results:

FEA model (element mesh, pressure loading, B.C.'s from eann5mod.bmp)

Radial stress contour (eann5rad.bmp):

Tangential stress contour (eann5tan.bmp):

1. Maximum Radial Stress = 12,328 psi, Location (see stress contour)
2. Maximum Tangential Stress = 35,963 psi, Location = 1.000" diameter



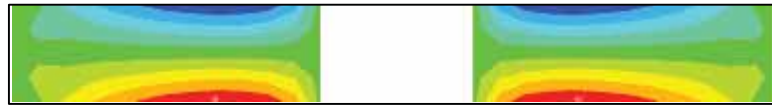
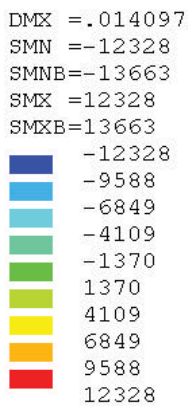
FLAT ANNULAR ENDCAP ANALYSIS 5-DWG
(eann5.bmp)



FLAT ANNULAR ENDCAP ANALYSIS 5-MOD
(eann5mod.bmp)

ANSYS 5.3
AUG 1 1997
11:36:11
ELEMENTS
TYPE NUM
U
E
PRES

ANSYS 5.3
AUG 1 1997
11:37:29
NODAL SOLUTION
STEP=1
SUB =1
TIME=1
SX (AVG)
RSYS=0



FLAT ANNULAR ENDCAP ANALYSIS 5-RAD
(eann5rad.bmp)

ZV =1
*DIST=1.509
*XF =1.047
*YF =.3125



FLAT ANNULAR ENDCAP ANALYSIS 5-TAN
(eann5tan.bmp)

ANSYS 5.3
AUG 1 1997
11:38:38
NODAL SOLUTION
STEP=1
SUB =1
TIME=1
SZ (AVG)
RSYS=0
DMX =.014097
SMN =-35963
SMNB=-36142
SMX =35963
SMXB=36142
-35963
-27971
-19979
-11988
-3996
3996
11988
19979
27971
35963

Case 6: Plate Outside Diameter = 6.00”, Plate Free Diameter = 5.00”, Plate thickness = .625”, Center Hole Diameter = 1.00”, Simply Supported Outer Edge Restraint, Guided Inner (Center Hole) Edge Restraint

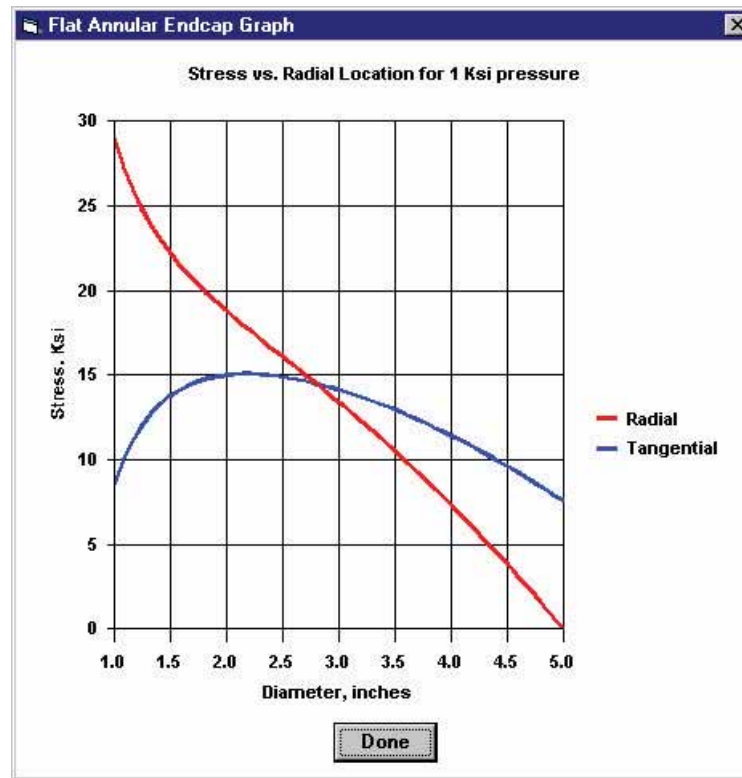
Plate cross section with applied pressure, B.C.’s (eann6.bmp):

Under Pressure numerical stress results:

The screenshot shows the 'Flat Annular Endcap Analysis - External Pressure' software interface. At the top, the 'Pressure Range' is set to '1000 - 20000 psi' with a unit dropdown set to 'psi'. There are buttons for 'Graph', 'Report', and 'Done'. The 'Theoretical Failure' section lists four options: 'Radial moment failure at 1212.4 psi (Dia. = 1.0000 inches)' (selected), 'Tangential moment failure at 2324.3 psi (Dia. = 2.1429 inches)', 'Shear stress failure at 8750.0 psi (Dia. = 5.0000 inches)', and 'Seat failure at 10694 psi'. A 'Fail value units' dropdown is set to 'psi'. Below this, 'Table eval dia., De (inches)' is set to '1.0000' with a 'ReCalc' button. The 'Units' section has dropdowns for Pressure (psi), Depth (Ft (sea)), Stress (psi), Linear Deflection (Inches), and Angle Deflection (Degrees). On the right, 'Material' is 'Test Aluminum' with a 'Yield Strength' of '35000 psi'. Below that, 'Weight' is shown as 'Air Wt 1.7181' and 'Water Wt 1.0817' with 'Units' set to 'lbs'. At the bottom is a table with the following data:

| Pressure | Depth | De Radial | De Tangential | De Shear |
|---------------|----------|-------------|---------------|-------------|
| psi | Ft (sea) | Stress, psi | Stress, psi | Stress, psi |
| 1000.0 | 2248.8 | 28869 | 8660.8 | 0.0000 |
| 1212.4 (fail) | 2725.5 | 35000 | 10500 | 0.0000 |
| 2000.0 (fail) | 4490.9 | 57739 | 17322 | 0.0000 |
| 3000.0 (fail) | 6726.2 | 86608 | 25982 | 0.0000 |
| 4000.0 (fail) | 8954.8 | 115480 | 34643 | 0.0000 |
| 5000.0 (fail) | 11177 | 144350 | 43304 | 0.0000 |
| 6000.0 (fail) | 13392 | 173220 | 51965 | 0.0000 |
| 7000.0 (fail) | 15601 | 202090 | 60626 | 0.0000 |
| 8000.0 (fail) | 17804 | 230960 | 69287 | 0.0000 |

Clicking on **Graph** gives the following:



1. Maximum Radial Stress = 28,869 psi, Location = 1.000" diameter (hole diameter)
2. Maximum Tangential Stress = 15,058 psi , Location = 2.143" diameter

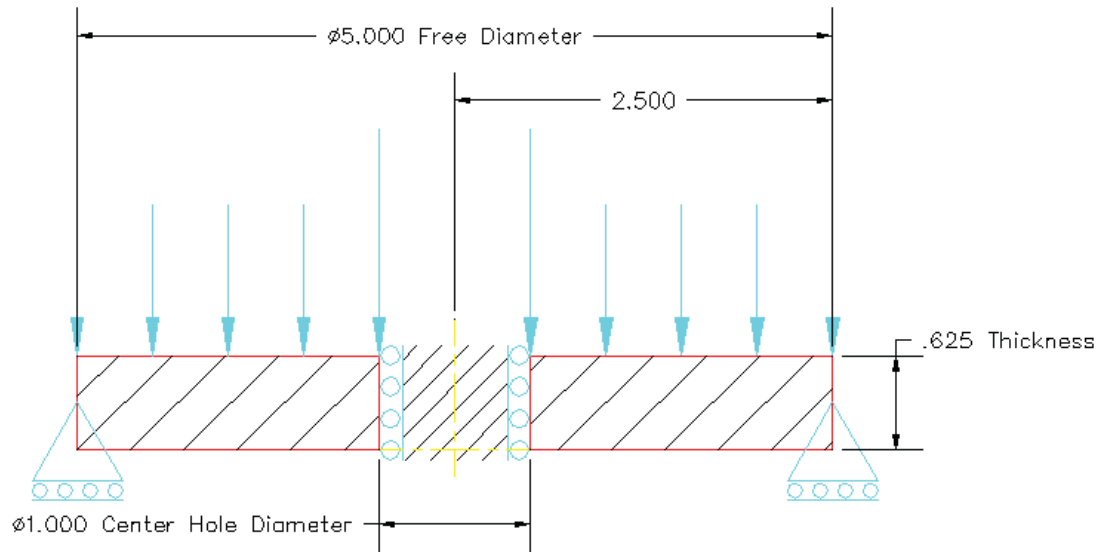
Finite Element Method graphical stress results:

FEA model (element mesh, pressure loading, B.C.'s from eann6mod.bmp)

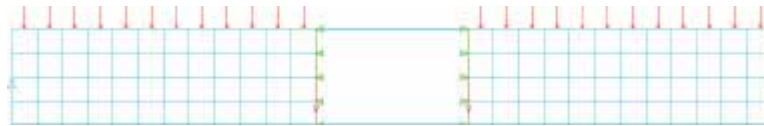
Radial stress contour (eann6rad.bmp):

Tangential stress contour (eann6tan.bmp):

1. Maximum Radial Stress = 24,257 psi, Location = 1.000" diameter
2. Maximum Tangential Stress = 17,005 psi, Location (see stress contour)



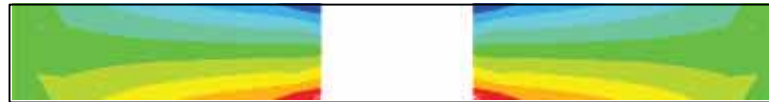
FLAT ANNULAR ENDCAP ANALYSIS 6-DWG
(eann6.bmp)



FLAT ANNULAR ENDCAP ANALYSIS 6-MOD
(eann6mod.bmp)

ANSYS 5.3
AUG 1 1997
11:40:12
ELEMENTS
TYPE NUM
U
F
CP
PRES

ANSYS 5.3
AUG 1 1997
11:43:47
NODAL SOLUTION
STEP=1
SUB =1
TIME=1
SX (AVG)
RSYS=0



FLAT ANNULAR ENDCAP ANALYSIS 6-RAD
(eann6rad.bmp)

ZV =1
*DIST=1.509
*XF =1.047
*YF =.3125

DMX =.009515
SMN =-24257
SMNB=-43617
SMX =24257
SMXB=43617
-24257
-18866
-13476
-8086
-2695
2695
8086
13476
18866
24257



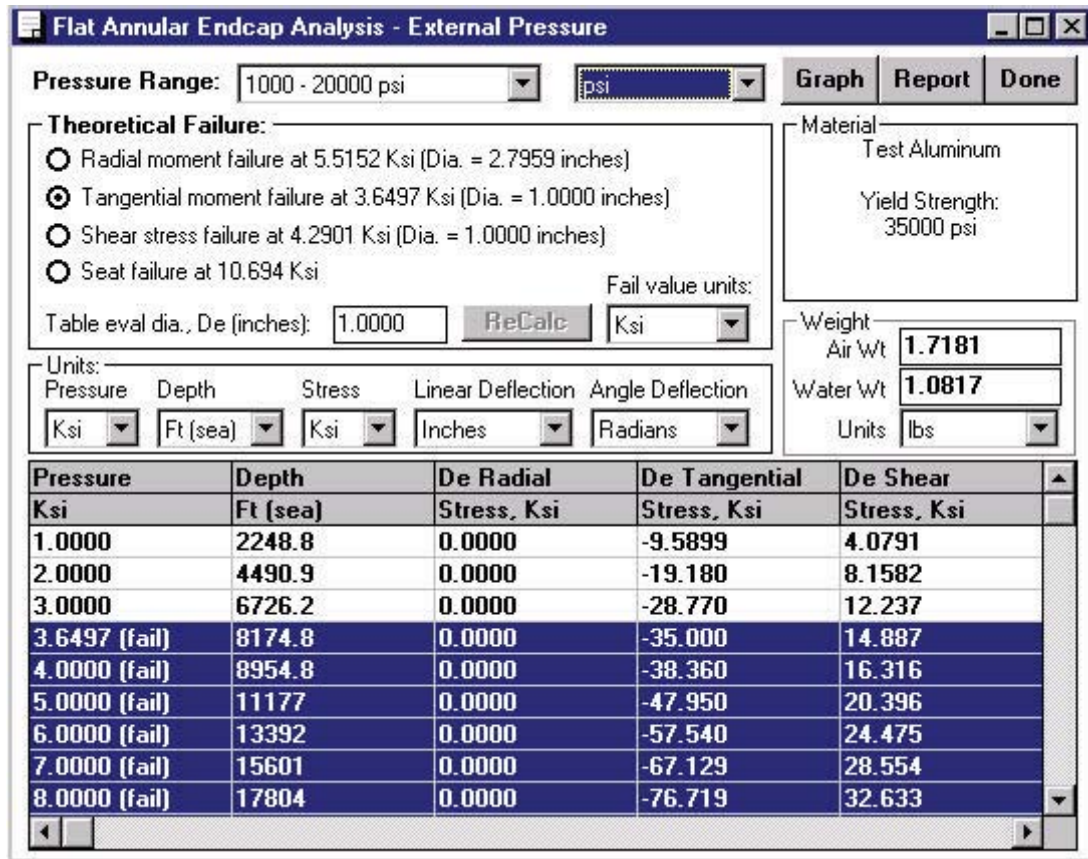
FLAT ANNULAR ENDCAP ANALYSIS 6-TAN
(eann6tan.bmp)

ANSYS 5.3
AUG 1 1997
10:43:01
NODAL SOLUTION
STEP=1
SUB =1
TIME=1
SX (AVG)
RSYS=0
DMX =.004447
SMN =-12013
SMNB=-22675
SMX =12013
SMXB=22675
-12013
-9343
-6674
-4004
-1335
1335
4004
6674
9343
12013

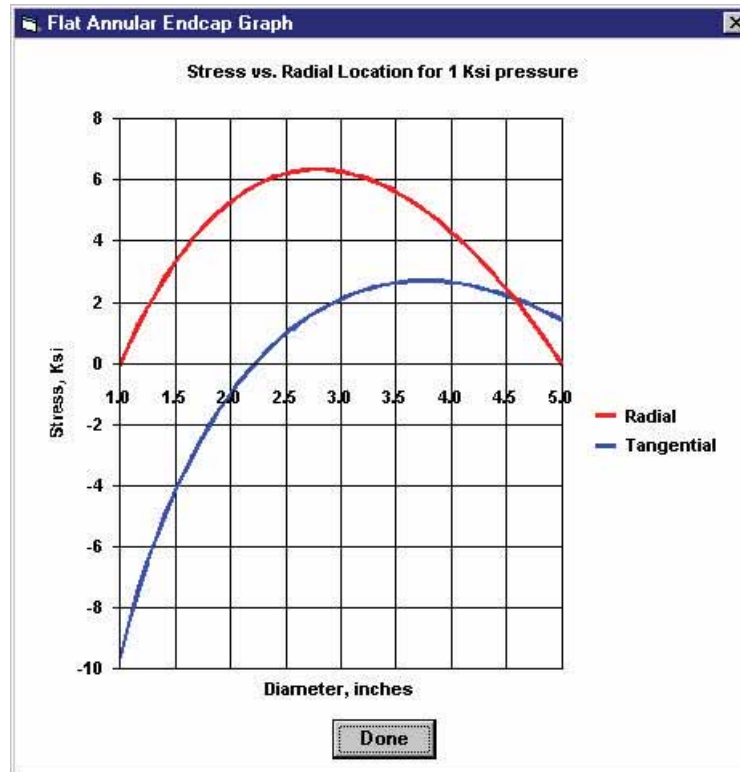
Case 7: Plate Outside Diameter = 6.00”, Plate Free Diameter = 5.00”, Plate thickness = .625”, Center Hole Diameter = 1.00”, Simply Supported Outer Edge Restraint, Simply Supported Inner (Center Hole) Edge Restraint

Plate cross section with applied pressure, B.C.’s (eann7.bmp):

Under Pressure numerical stress results:



Clicking on **Graph** gives the following:



1. Maximum Radial Stress = 6346 psi, Location = 2.796" diameter
2. Maximum Tangential Stress = -9590 psi, Location = 1.000" diameter (hole diameter)

Finite Element Method graphical stress results:

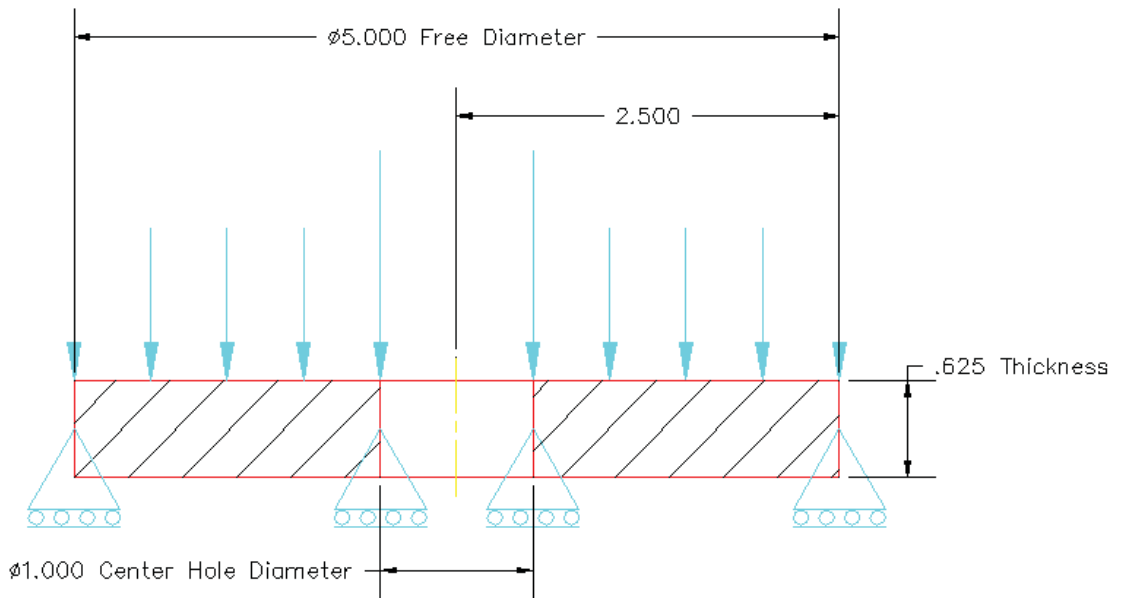
FEA model (element mesh, pressure loading, B.C.'s from eann7mod.bmp):

Radial stress contour (eann7rad.bmp):

Tangential stress contour (eann7tan.bmp):

1. Maximum Radial Stress = 6777 psi, Location (see stress contour)
2. Maximum Tangential Stress = 7694 psi, Location = 1.000" diameter

Under Pressure, Ve



FLAT ANNULAR ENDCAP ANALYSIS 7-DWG
(eann7.bmp)

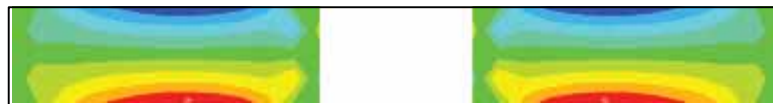


FLAT ANNULAR ENDCAP ANALYSIS 7-MOD
(eann7mod.bmp)

```
ANSYS 5.3
AUG 1 1997
11:46:24
ELEMENTS
TYPE NUM
U
PRES
ZV =1
*DIST=1.509
*KF =1.047
*YF =.3125
```

```
ANSYS 5.3
AUG 1 1997
11:48:07
NODAL SOLUTION
STEP=1
SUB =1
TIME=1
SX (AVG)
RSYS=0
DMX =.001218
SMN =-6777
SMNB=-22612
SMX =6777
SMXB=22612
```

| | |
|--------------|----------|
| Dark Blue | -6777 |
| Blue | -5271 |
| Light Blue | -3765 |
| Green | -2259 |
| Light Green | -752.965 |
| Yellow-Green | 752.965 |
| Yellow | 2259 |
| Orange | 3765 |
| Red-Orange | 5271 |
| Red | 6777 |



FLAT ANNULAR ENDCAP ANALYSIS 7-RAD
(eann7rad.bmp)



FLAT ANNULAR ENDCAP ANALYSIS 7-TAN
(eann7tan.bmp)

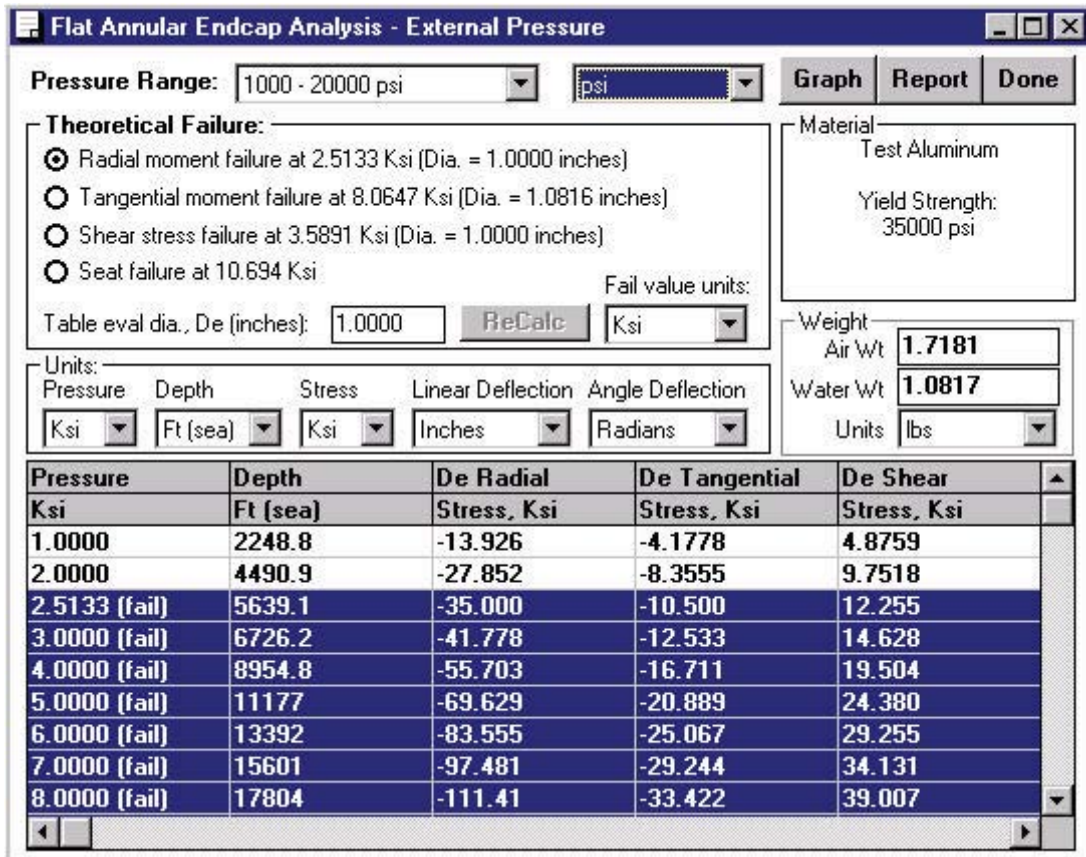
```
ANSYS 5.3
AUG 1 1997
11:49:05
NODAL SOLUTION
STEP=1
SUB =1
TIME=1
SZ (AVG)
RSYS=0
DMX =.001218
SMN =-7694
SMNB=-22612
SMX =7694
SMXB=22612
```

| | |
|--------------|----------|
| Dark Blue | -7694 |
| Blue | -5984 |
| Light Blue | -4275 |
| Green | -2565 |
| Light Green | -854.911 |
| Yellow-Green | 854.911 |
| Yellow | 2565 |
| Orange | 4275 |
| Red-Orange | 5984 |
| Red | 7694 |

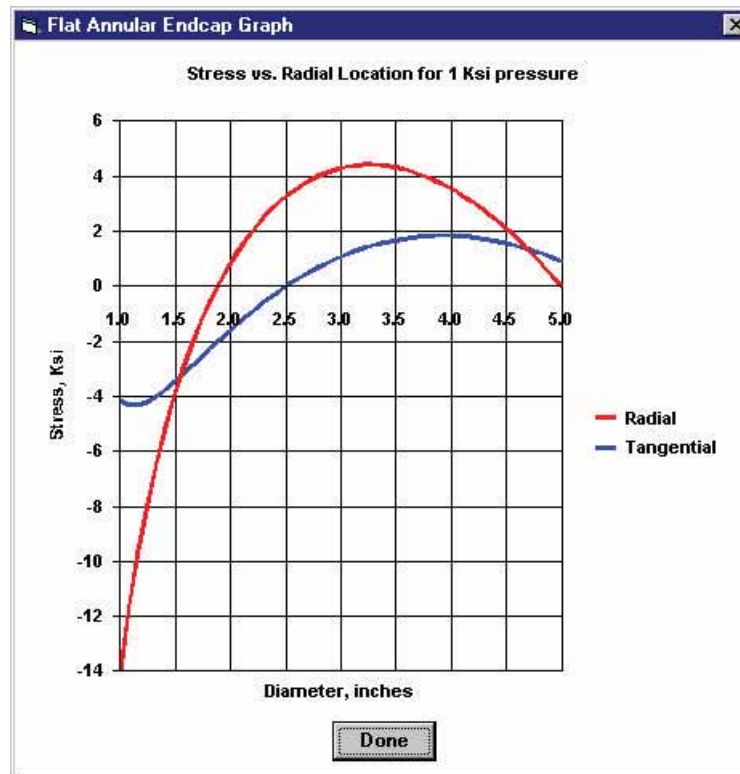
Case 8: Plate Outside Diameter = 6.00”, Plate Free Diameter = 5.00”, Plate thickness = .625”, Center Hole Diameter = 1.00”, Simply Supported Outer Edge Restraint, Fixed Inner (Center Hole) Edge Restraint

Plate cross section with applied pressure, B.C.’s (eann8.bmp):

Under Pressure numerical stress results:



Clicking on **Graph** gives the following:



1. Maximum Radial Stress = -13,926 psi, Location = 1.000" diameter (hole diameter)
2. Maximum Tangential Stress = -4340 psi , Location = 1.082" diameter

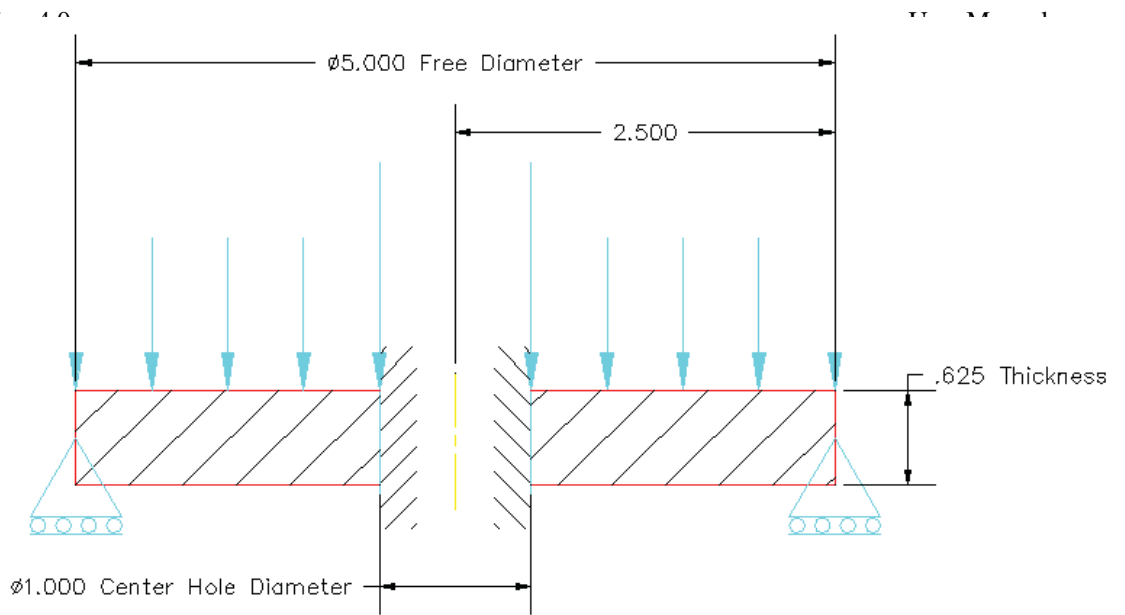
Finite Element Method graphical stress results:

FEA model (element mesh, pressure loading, B.C.'s from eann8mod.bmp):

Radial stress contour (eann8rad.bmp):

Tangential stress contour (eann8tan.bmp):

1. Maximum Radial Stress = -8442 psi, Location = 1.000" diameter
2. Maximum Tangential Stress = -4745 psi, Location (see stress contour)



FLAT ANNULAR ENDCAP ANALYSIS 8-DWG
(eann8.bmp)



FLAT ANNULAR ENDCAP ANALYSIS 8-MOD
(eann8mod.bmp)

ANSYS 5.3
AUG 1 1997
11:53:26
ELEMENTS
TYPE NUM
U
CP
PRES

ANSYS 5.3
AUG 1 1997
11:55:17
NODAL SOLUTION
STEP=1
SUB =1
TIME=1
SX (AVG)
RSYS=0
DMX =.001027
SMN =-8442
SMNB=-25539
SMX =8442
SMXR=25539

| |
|----------|
| 8442 |
| 6566 |
| 4690 |
| 2814 |
| -937.999 |
| -25539 |
| -4690 |
| -6566 |
| -8442 |



FLAT ANNULAR ENDCAP ANALYSIS 8-RAD
(eann8rad.bmp)

ZV =1
*DIST=1.509
*XF =1.047
*YF =.3125



FLAT ANNULAR ENDCAP ANALYSIS 8-TAN
(eann8tan.bmp)

ANSYS 5.3
AUG 1 1997
11:57:10
NODAL SOLUTION
STEP=1
SUB =1
TIME=1
SZ (AVG)
RSYS=0
DMX =.001027
SMN =-4745
SMNB=-25539
SMX =4745
SMXB=25539

| |
|---------|
| 4745 |
| 3691 |
| 2636 |
| 1582 |
| -527.25 |
| -1582 |
| -2636 |
| -3691 |
| -4745 |

Case 9: Plate Outside Diameter = 6.00”, Plate Free Diameter = 5.00”, Plate thickness = .625”, Center Hole Diameter = 1.00”, Guided Outer Edge Restraint, Simply Supported Inner (Center Hole) Edge Restraint

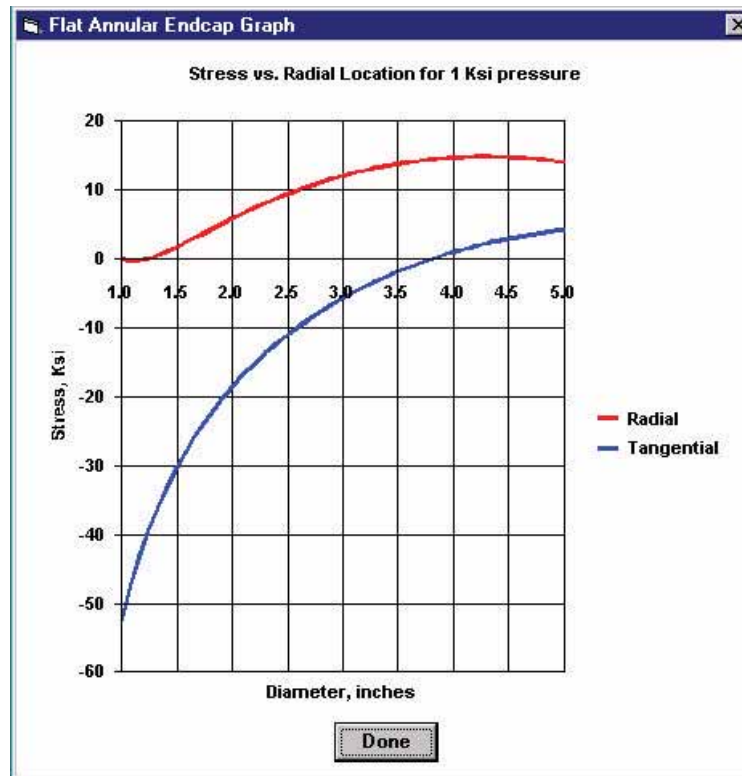
Plate cross section with applied pressure, B.C.’s (eann9.bmp):

Under Pressure numerical stress results:

The screenshot shows the 'Flat Annular Endcap Analysis - External Pressure' software interface. At the top, the 'Pressure Range' is set to '1000 - 20000 psi'. The 'Theoretical Failure' section lists four failure modes: Radial moment failure at 2.3739 Ksi (Dia. = 4.2653 inches), Tangential moment failure at 0.67023 Ksi (Dia. = 1.0000 inches) (selected), Shear stress failure at 1.7500 Ksi (Dia. = 1.0000 inches), and Seat failure at 10.694 Ksi. The material is 'Test Aluminum' with a 'Yield Strength: 35000 psi'. The weight is 'Air Wt 1.7181' and 'Water Wt 1.0817' in 'lbs' units. The table below shows stress results for pressures from 1.0000 to 9.0000 Ksi at depths from 2248.8 to 20000 Ft (sea). All results are marked as 'fail'.

| Pressure Ksi | Depth Ft (sea) | De Radial Stress, Ksi | De Tangential Stress, Ksi | De Shear Stress, Ksi |
|---------------|----------------|-----------------------|---------------------------|----------------------|
| 1.0000 (fail) | 2248.8 | 0.0000 | -52.221 | 10.000 |
| 2.0000 (fail) | 4490.9 | 0.0000 | -104.44 | 20.000 |
| 3.0000 (fail) | 6726.2 | 0.0000 | -156.66 | 30.000 |
| 4.0000 (fail) | 8954.8 | 0.0000 | -208.88 | 40.000 |
| 5.0000 (fail) | 11177 | 0.0000 | -261.10 | 50.000 |
| 6.0000 (fail) | 13392 | 0.0000 | -313.32 | 60.000 |
| 7.0000 (fail) | 15601 | 0.0000 | -365.55 | 70.000 |
| 8.0000 (fail) | 17804 | 0.0000 | -417.77 | 80.000 |
| 9.0000 (fail) | 20000 | 0.0000 | -469.99 | 90.000 |

Clicking on **Graph** gives the following:



1. Maximum Radial Stress = 14,744 psi , Location = 4.265" diameter
2. Maximum Tangential Stress = -52,221 psi , Location = 1.000" diameter (hole diameter)

Finite Element Method graphical stress results:

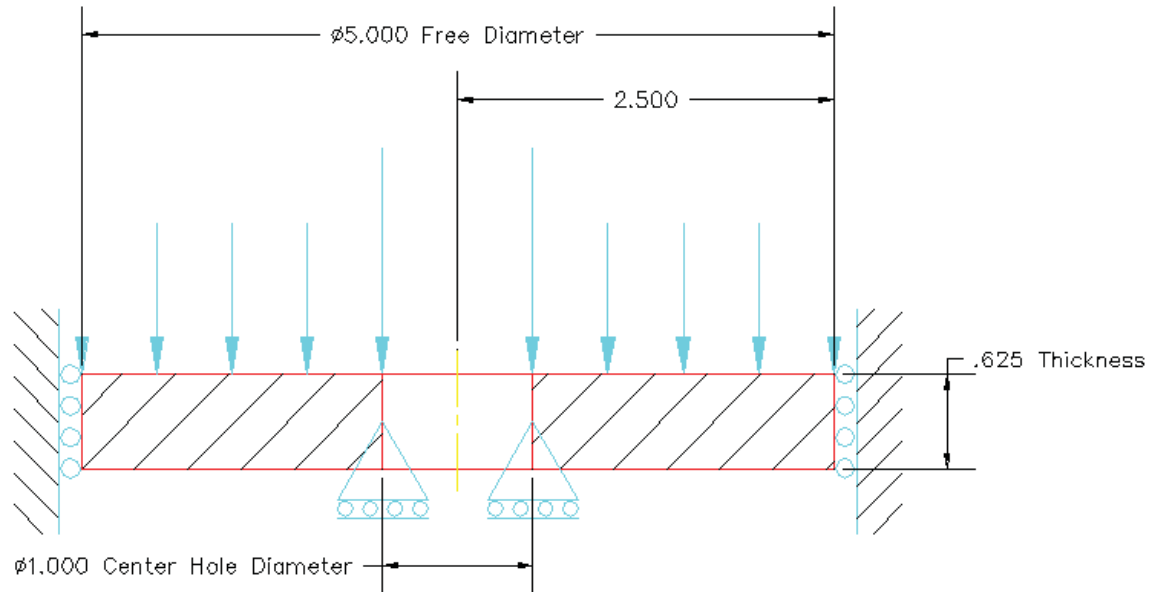
FEA model (element mesh, pressure loading, B.C.'s from eann9mod.bmp):

Radial stress contour (eann9rad.bmp):

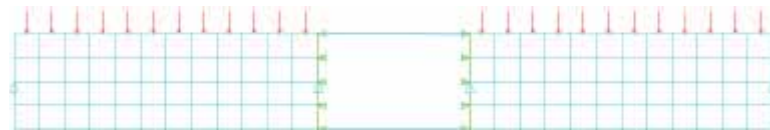
Tangential stress contour (eann9tan.bmp):

1. Maximum Radial Stress = 15,887 psi, Location (see stress contour)
2. Maximum Tangential Stress = -53,181 psi, Location = 1.000" diameter

Under Pressure, Ve

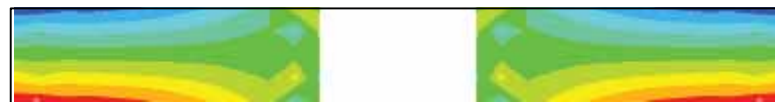


FLAT ANNULAR ENDCAP ANALYSIS 9-DWG
(eann9.bmp)



FLAT ANNULAR ENDCAP ANALYSIS 9-MOD
(eann9mod.bmp)

ANSYS 5.3
AUG 1 1997
11:53:26
ELEMENTS
TYPE NUM
U
CP
PRES



FLAT ANNULAR ENDCAP ANALYSIS 9-RAD
(eann9rad.bmp)

ZV =1
*DIST=1.509
*XF =1.047
*YF =.3125

ANSYS 5.3
AUG 1 1997
12:00:28
NODAL SOLUTION
STEP=1
SUB =1
TIME=1
SX (AVG)
RSYS=0



FLAT ANNULAR ENDCAP ANALYSIS 9-TAN
(eann9tan.bmp)

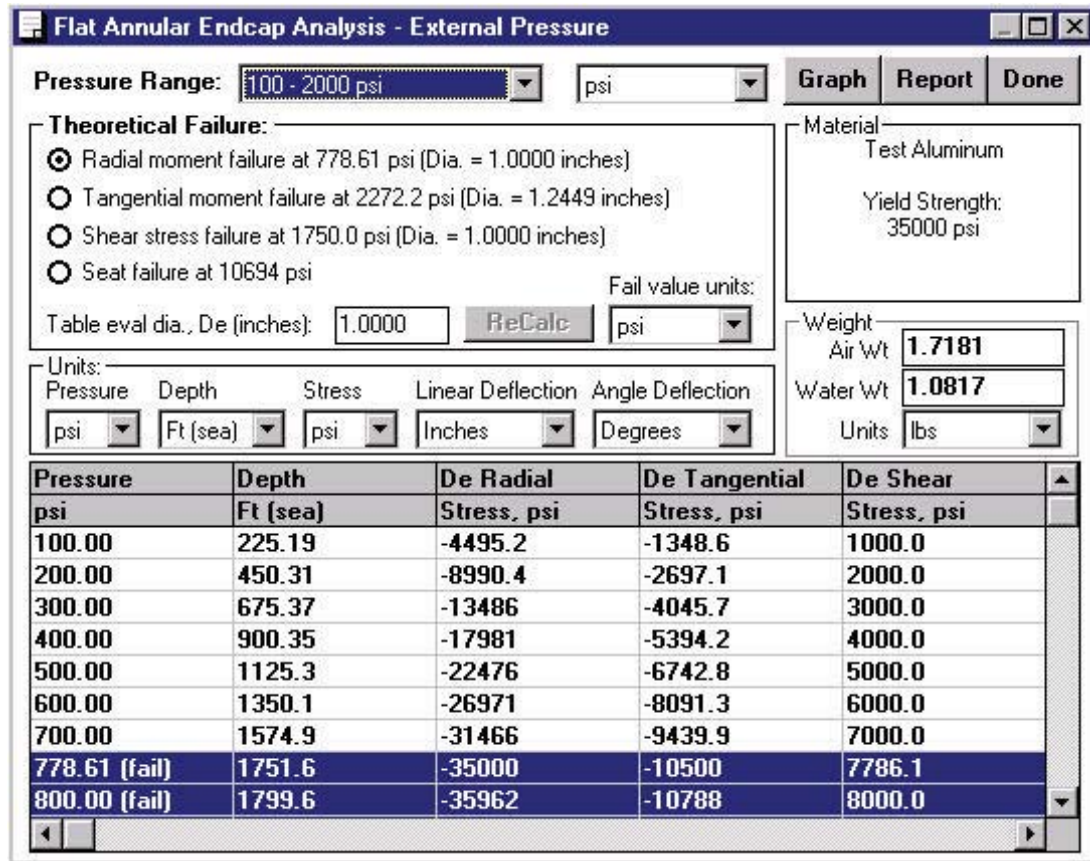
ANSYS 5.3
AUG 1 1997
12:01:18
NODAL SOLUTION
STEP=1
SUB =1
TIME=1
SZ (AVG)
RSYS=0
DMX =.011819
SMN =-53181
SMNB=-68341
SMX =53181
SMXB=68341

| | |
|--------------|--------|
| Blue | -53181 |
| Light Blue | -41363 |
| Light Green | -29545 |
| Green | -17727 |
| Yellow-Green | -5909 |
| Yellow | 5909 |
| Light Orange | 17727 |
| Orange | 29545 |
| Red-Orange | 41363 |
| Red | 53181 |

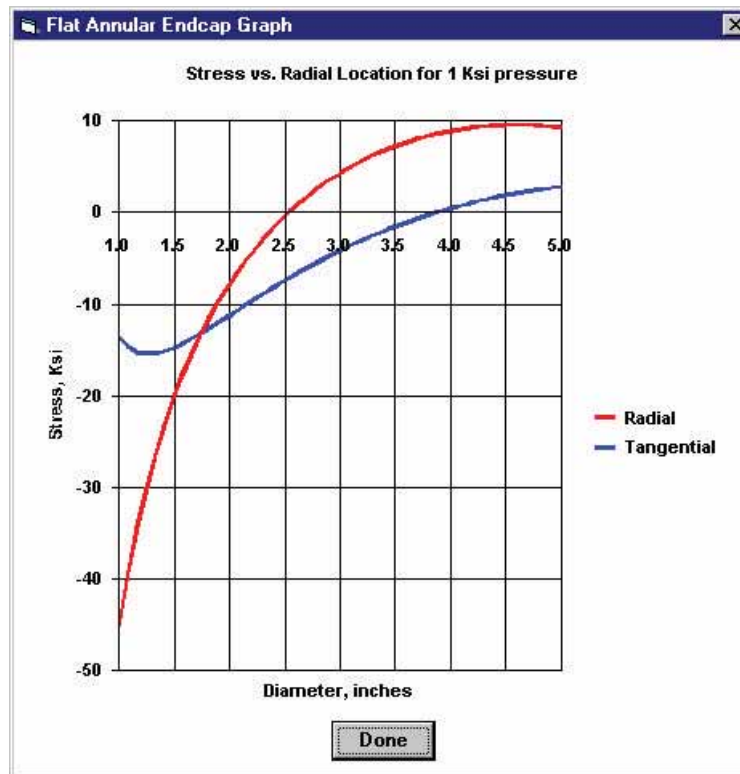
Case 10: Plate Outside Diameter = 6.00”, Plate Free Diameter = 5.00”, Plate thickness = .625”, Center Hole Diameter = 1.00”, Guided Outer Edge Restraint, Fixed Inner (Center Hole) Edge Restraint

Plate cross section with applied pressure, B.C.’s (eann10.bmp):

Under Pressure numerical stress results:



Clicking on **Graph** gives the following:



1. Maximum Radial Stress = -44,952 psi, Location = 1.000" diameter (hole diameter)
2. Maximum Tangential Stress = -15,403 psi, Location = 1.245" diameter

Finite Element Method graphical stress results:

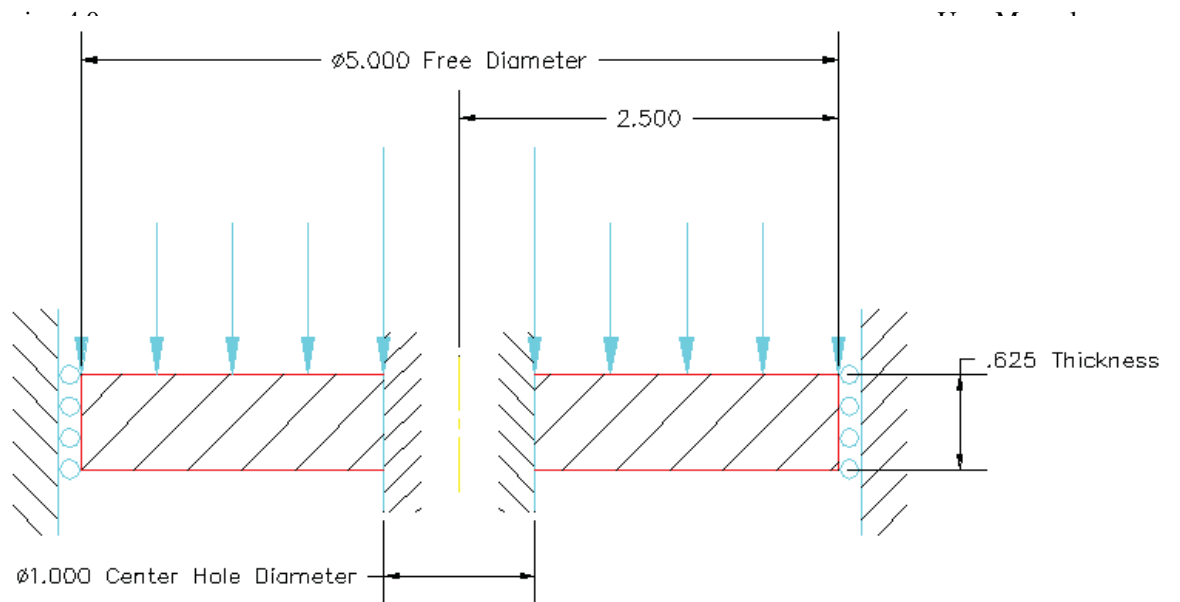
FEA model (element mesh, pressure loading, B.C.'s from eann10mod.bmp):

Radial stress contour (eann10rad.bmp):

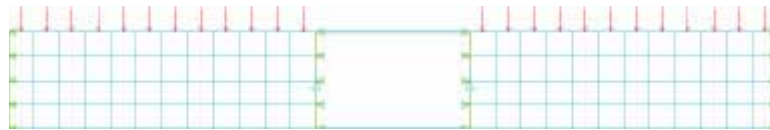
Tangential stress contour (eann10tan.bmp):

1. Maximum Radial Stress = -40,655 psi, Location = 1.000" diameter
2. Maximum Tangential Stress = -25,680 psi, Location (see stress contour)

Under Pressure, V



FLAT ANNULAR ENDCAP ANALYSIS 10-DWG
(eann10.bmp)

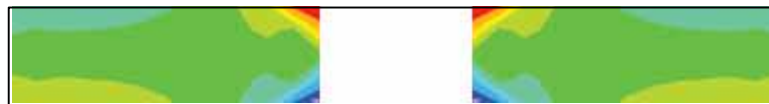


FLAT ANNULAR ENDCAP ANALYSIS 10-MOD
(eann10mod.bmp)

```
ANSYS 5.3
AUG 1 1997
12:02:41
ELEMENTS
TYPE NUM
U
CP
PRES
```

```
EV =1
*DIST=1.509
*XF =1.047
*YF =.3125
```

```
ANSYS 5.3
AUG 1 1997
12:03:56
NODAL SOLUTION
STEP=1
SUB =1
TIME=1
SX (AVG)
RSYS=0
DMX =.007346
SMN =-40655
SMNB=-74796
SMX =40655
SMXB=74796
```



FLAT ANNULAR ENDCAP ANALYSIS 10-RAD
(eann10rad.bmp)

| | |
|--------------|--------|
| Blue | -40655 |
| Light Blue | -31621 |
| Light Green | -22586 |
| Green | -13552 |
| Yellow-Green | -4517 |
| Yellow | 4517 |
| Orange | 13552 |
| Red-Orange | 22586 |
| Red | 31621 |
| Dark Red | 40655 |



FLAT ANNULAR ENDCAP ANALYSIS 10-TAN
(eann10tan.bmp)

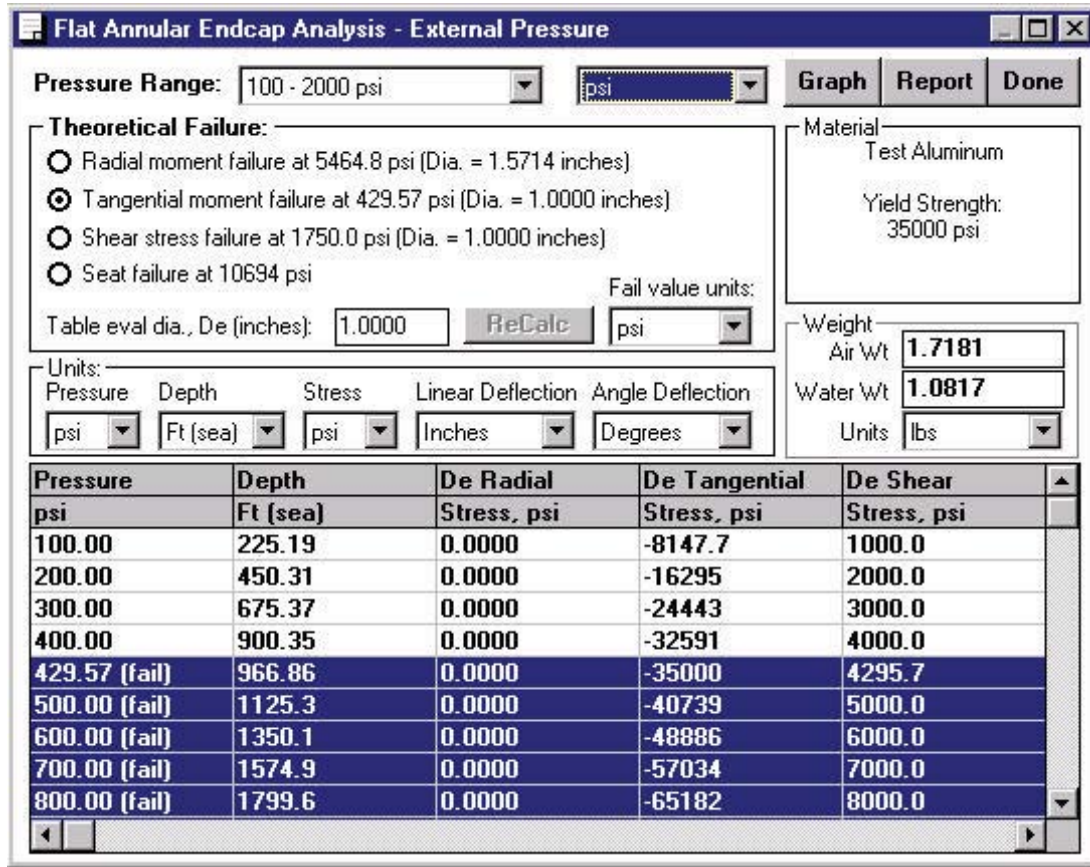
```
ANSYS 5.3
AUG 1 1997
12:04:59
NODAL SOLUTION
STEP=1
SUB =1
TIME=1
SZ (AVG)
RSYS=0
DMX =.007346
SMN =-25680
SMNB=-62496
SMX =25680
SMXB=62496
```

| | |
|--------------|--------|
| Blue | -25680 |
| Light Blue | -19974 |
| Light Green | -14267 |
| Green | -8560 |
| Yellow-Green | -2853 |
| Yellow | 2853 |
| Orange | 8560 |
| Red-Orange | 14267 |
| Red | 19974 |
| Dark Red | 25680 |

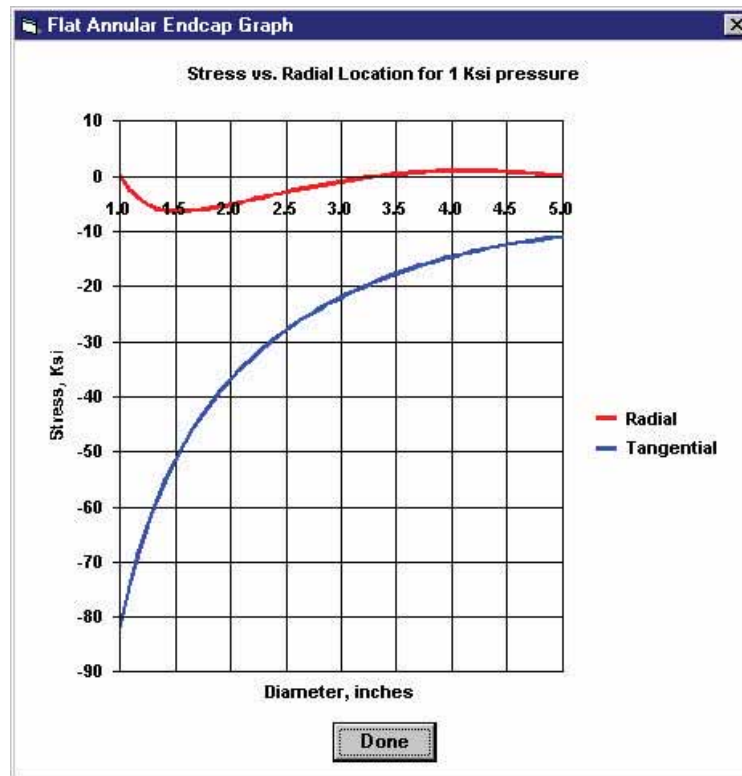
Case 11: Plate Outside Diameter = 6.00”, Plate Free Diameter = 5.00”, Plate thickness = .625”, Center Hole Diameter = 1.00”, Free Outer Edge Restraint, Simply Supported (Center Hole) Edge Restraint

Plate cross section with applied pressure, B.C.’s (eann11.bmp):

Under Pressure numerical stress results:



Clicking on **Graph** gives the following:



1. Maximum Radial Stress = -6405 psi, Location = 1.571" diameter
2. Maximum Tangential Stress = -81,477 psi, Location = 1.000" diameter (hole diameter)

Finite Element Method graphical stress results:

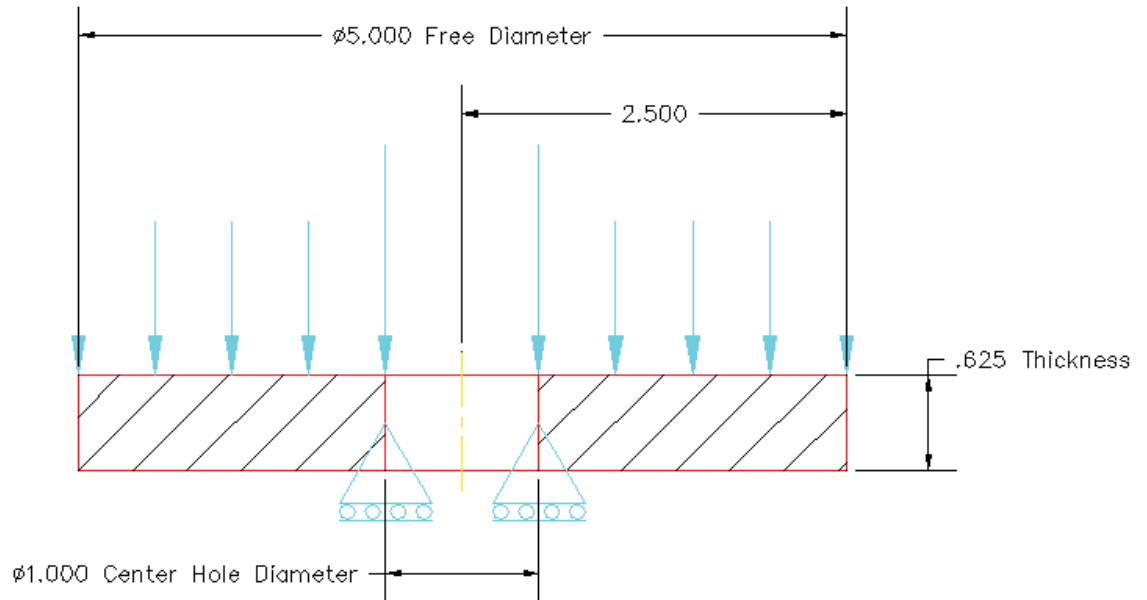
FEA model (element mesh, pressure loading, B.C.'s from eann11mod.bmp):

Radial stress contour (eann11rad.bmp):

Tangential stress contour (eann11tan.bmp):

1. Maximum Radial Stress = -12,444 psi, Location (see stress contour)
2. Maximum Tangential Stress = -80,678 psi, Location = 1.000" diameter

Under Pressure,



FLAT ANNULAR ENDCAP ANALYSIS 11-DWG
(eann11.bmp)

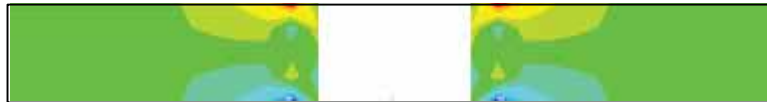


FLAT ANNULAR ENDCAP ANALYSIS 11-MOD
(eann11mod.bmp)

```
ANSYS 5.3
AUG 1 1997
12:07:44
ELEMENTS
TYPE NUM
U
PRES
```

```
EV =1
*DIST=1.509
*XF =1.047
*YF =.3125
```

```
ANSYS 5.3
AUG 1 1997
12:09:17
NODAL SOLUTION
STEP=1
SUB =1
TIME=1
SX (AVG)
RSYS=0
```



FLAT ANNULAR ENDCAP ANALYSIS 11-RAD
(eann11rad.bmp)

```
DMX =.023817
SMN =-12444
SMNB=-61741
SMX =12444
SMXB=61741
-12444
-9678
-6913
-4148
-1383
1383
4148
6913
9678
12444
```



FLAT ANNULAR ENDCAP ANALYSIS 11-TAN
(eann11tan.bmp)

```
ANSYS 5.3
AUG 1 1997
12:10:22
NODAL SOLUTION
STEP=1
SUB =1
TIME=1
SZ (AVG)
RSYS=0
```

```
DMX =.023817
SMN =-80678
SMNB=-87099
SMX =80678
SMXB=87099
-80678
-62749
-44821
-26893
-8964
8964
26893
44821
62749
80678
```


Case 12: Plate Outside Diameter = 6.00”, Plate Free Diameter = 5.00”, Plate thickness = .625”, Center Hole Diameter = 1.00”, Free Outer Edge Restraint, Fixed Inner (Center Hole) Edge Restraint

Plate cross section with applied pressure, B.C.’s (eann12.bmp):

Under Pressure numerical stress results:

Flat Annular Endcap Analysis - External Pressure

Pressure Range: 100 - 2000 psi psi Graph Report Done

Theoretical Failure:

- Radial moment failure at 594.23 psi (Dia. = 1.0000 inches)
- Tangential moment failure at 1610.1 psi (Dia. = 1.4082 inches)
- Shear stress failure at 1750.0 psi (Dia. = 1.0000 inches)
- Seat failure at 10694 psi

Table eval dia., De (inches): 1.0000 ReCalc psi Fail value units:

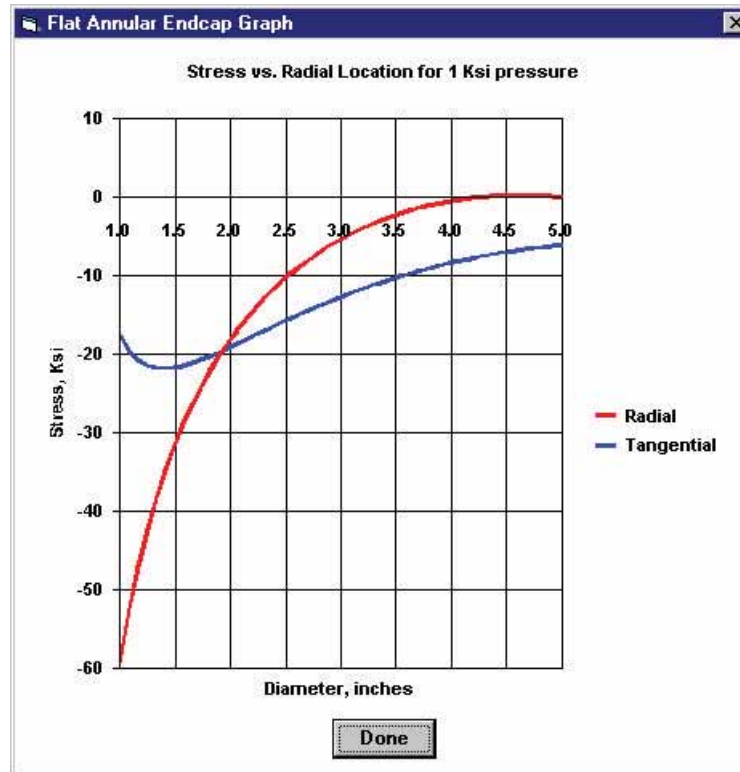
Material: Test Aluminum
Yield Strength: 35000 psi

Weight: Air Wt 1.7181 Water Wt 1.0817 Units lbs

Units: Pressure: psi Depth: Ft (sea) Stress: psi Linear Deflection: Inches Angle Deflection: Degrees

| Pressure | Depth | De Radial | De Tangential | De Shear |
|---------------|----------|-------------|---------------|-------------|
| psi | Ft (sea) | Stress, psi | Stress, psi | Stress, psi |
| 100.00 | 225.19 | -5889.9 | -1767.0 | 1000.0 |
| 200.00 | 450.31 | -11780 | -3534.0 | 2000.0 |
| 300.00 | 675.37 | -17670 | -5300.9 | 3000.0 |
| 400.00 | 900.35 | -23560 | -7067.9 | 4000.0 |
| 500.00 | 1125.3 | -29450 | -8834.9 | 5000.0 |
| 594.23 (fail) | 1337.2 | -35000 | -10500 | 5942.3 |
| 600.00 (fail) | 1350.1 | -35340 | -10602 | 6000.0 |
| 700.00 (fail) | 1574.9 | -41230 | -12369 | 7000.0 |
| 800.00 (fail) | 1799.6 | -47119 | -14136 | 8000.0 |

Clicking on **Graph** gives the following:



1. Maximum Radial Stress = -58,889 psi, Location = 1.000" diameter (hole diameter)
2. Maximum Tangential Stress = -21,738 psi, Location = 1.408" diameter

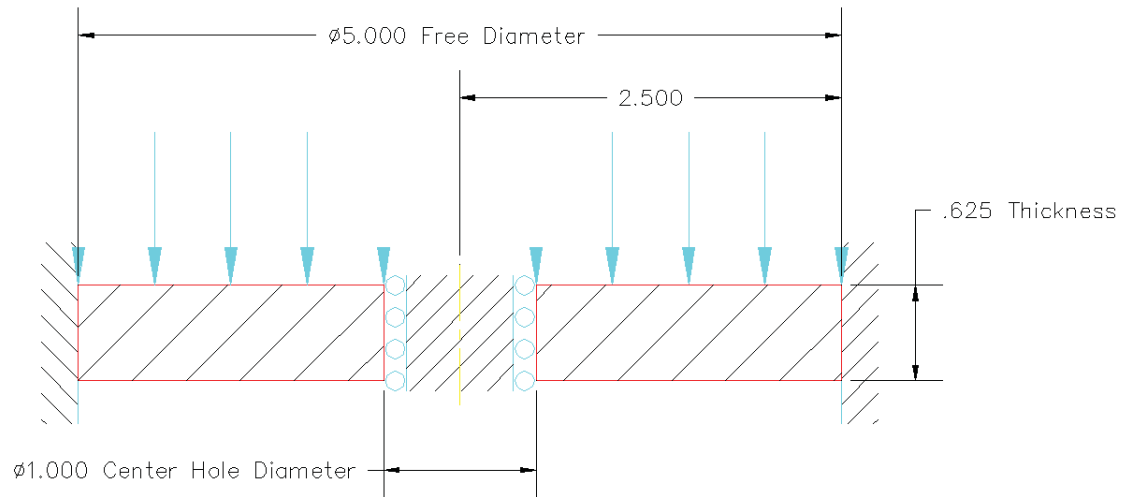
Finite Element Method graphical stress results:

FEA model (element mesh, pressure loading, B.C.'s from eann12mod.bmp):

Radial stress contour (eann12rad.bmp):

Tangential stress contour (eann12tan.bmp):

1. Maximum Radial Stress = -53,886 psi, Location = 1.000" diameter
2. Maximum Tangential Stress = -34,960 psi, Location (see stress contour)



FLAT ANNULAR ENDCAP ANALYSIS 12-DWG
(eann12.bmp)



FLAT ANNULAR ENDCAP ANALYSIS 12-MOD
(eann12mod.bmp)

```
ANSYS 5.3
AUG 1 1997
12:11:46
ELEMENTS
TYPE NUM
U
CP
```

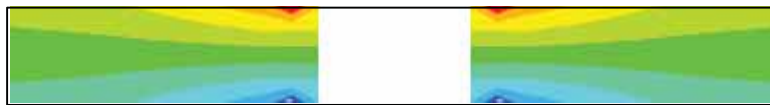
```
ZV =1
*DIST=1.509
*XF =1.047
*YF =.3125
```

```
ANSYS 5.3
AUG 1 1997
12:13:16
NODAL SOLUTION
STEP=1
SUB =1
TIME=1
SX (AVG)
RSYS=0
DMX =.013385
SMN =-53866
SMNB=-98064
SMX =53866
SMXB=98064
```



FLAT ANNULAR ENDCAP ANALYSIS 12-RAD
(eann12rad.bmp)

| | |
|--------------|--------|
| Blue | -53866 |
| Light Blue | -41896 |
| Medium Blue | -29926 |
| Green | -17955 |
| Yellow-Green | -5985 |
| Yellow | 5985 |
| Orange | 17955 |
| Red-Orange | 29926 |
| Red | 41896 |
| Dark Red | 53866 |



FLAT ANNULAR ENDCAP ANALYSIS 12-TAN
(eann12tan.bmp)

```
ANSYS 5.3
AUG 1 1997
12:14:10
NODAL SOLUTION
STEP=1
SUB =1
TIME=1
SZ (AVG)
RSYS=0
DMX =.013385
SMN =-34960
SMNB=-67171
SMX =34960
SMXB=67171
```

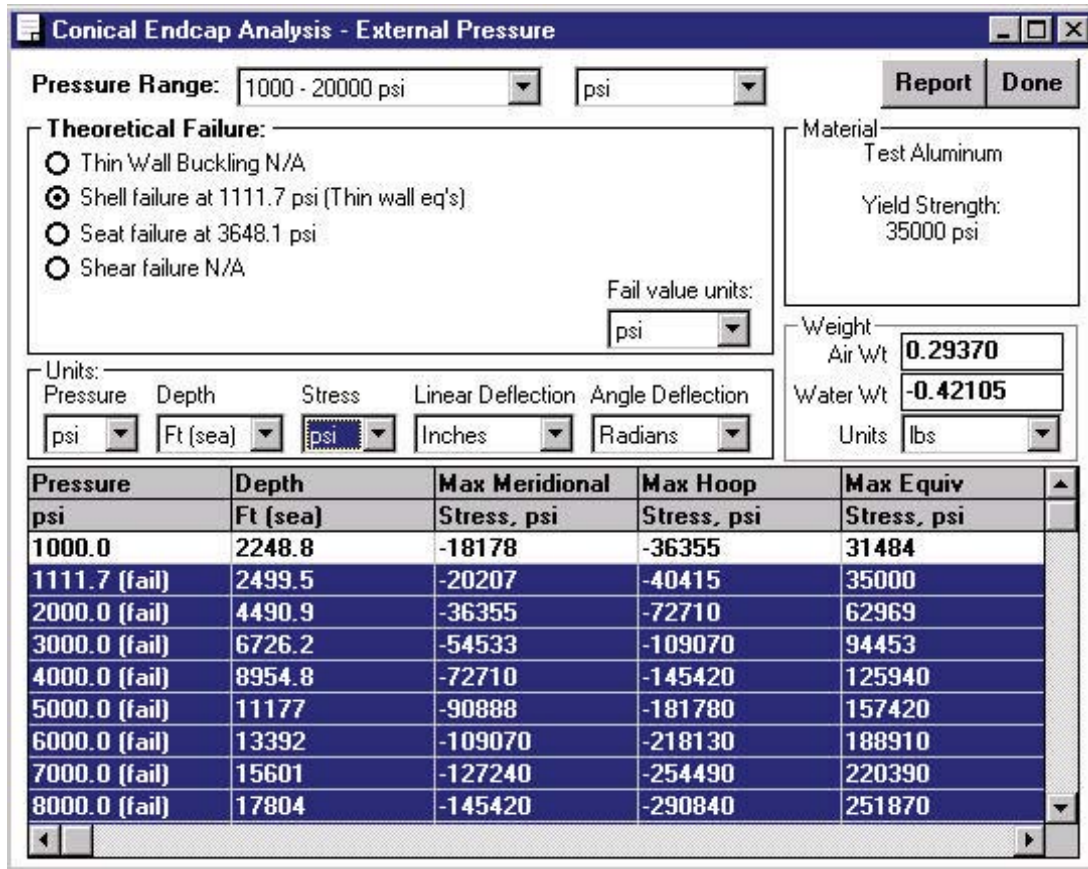
| | |
|--------------|--------|
| Blue | -34960 |
| Light Blue | -27191 |
| Medium Blue | -19422 |
| Green | -11653 |
| Yellow-Green | -3884 |
| Yellow | 3884 |
| Orange | 11653 |
| Red-Orange | 19422 |
| Red | 27191 |
| Dark Red | 34960 |

CONICAL ENDCAP ANALYSIS:

Base Inside Diameter = 5.00", Conical Outer Height = 2.641", Wall thickness = .10"

Conical endcap cross section with applied pressure, B.C.'s (econ1.bmp):

Under Pressure numerical stress results:



1. Maximum Hoop Stress = -36,355 psi

Finite Element Method graphical stress results:

FEA model (element mesh, pressure loading from econ1mod.bmp):

Hoop stress contour (econ1hoop.bmp):

1. Maximum Hoop Stress = -34,243 psi

Discussion:

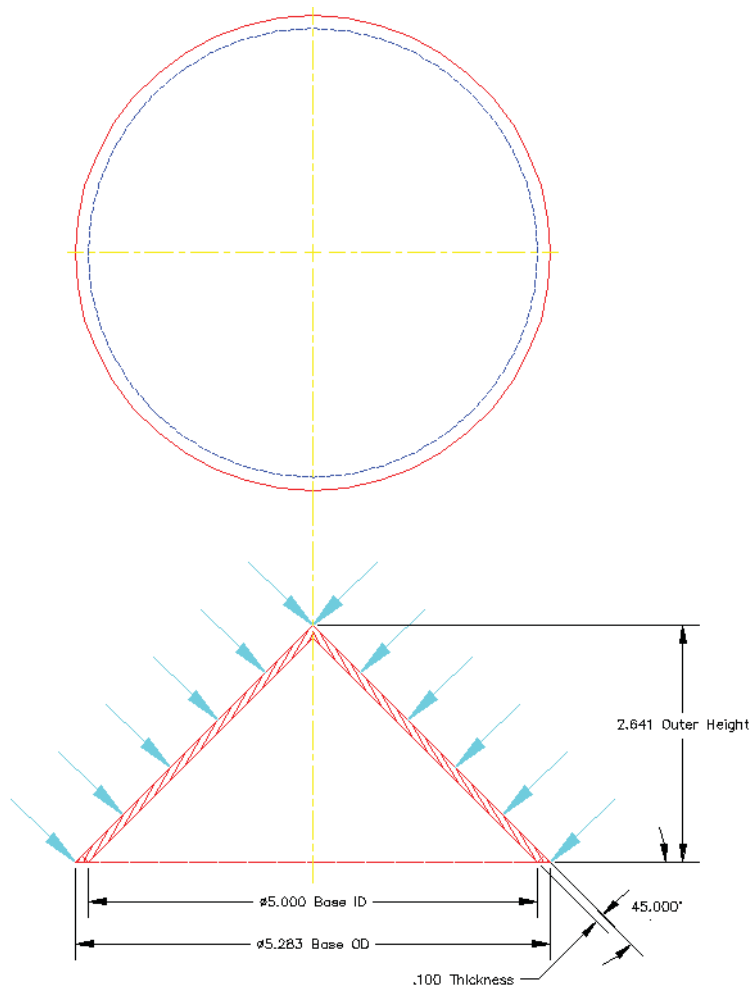
Under Pressure (based on the formulas given in Appendix C) and the conical endcap finite element model provide similar results for peak hoop stress. Appendix D defines the orientation of the hoop stresses in the conical endcap.

For this exercise, a mesh of two quadrilateral elements across the conical shell thickness was selected for the finite element model.

The finite element stress contour indicates that the peak hoop stress occurs at the base of the endcap.

HEMISPHERICAL ENDCAP ANALYSIS:Discussion:

See discussion of the sphere analysis above. Under Pressure used the same formulas to perform stress analysis for spheres and hemispherical endcaps.



CONICAL ENDCAP ANALYSIS – DWG
(econe1.bmp)



CONICAL ENDCAP ANALYSIS – HOOP
(econe1hoop.bmp)

```

ANSYS 5.3
AUG 1 1997
12:17:37
ELEMENTS
TYPE NUM
U
PRES
ZV =1
DIST=1.453
XF =1.285
YF =1.321
    
```



CONICAL ENDCAP ANALYSIS – MOD
(econe1mod.bmp)

```

ANSYS 5.3
AUG 1 1997
12:19:34
NODAL SOLUTION
STEP=1
SUB =1
TIME=1
SZ      (AVG)
RSYS=0
DMX =.009786
SMN =-34243
SMNB=-35955
SMX =-159.254
SMXB=258.883
    
```

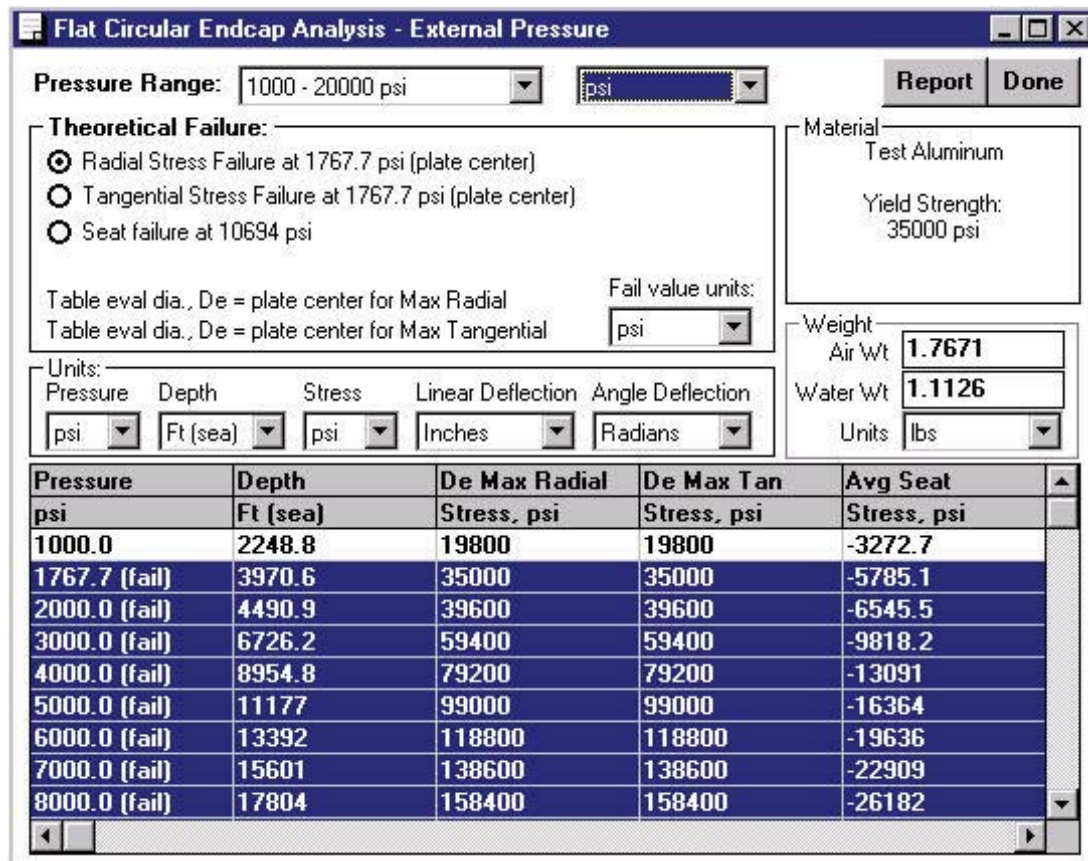
| | |
|--------------|----------|
| Blue | -34243 |
| Light Blue | -30456 |
| Medium Blue | -26669 |
| Green | -22882 |
| Light Green | -19095 |
| Yellow-Green | -15308 |
| Yellow | -11521 |
| Orange | -7733 |
| Red-Orange | -3946 |
| Red | -159.254 |

FLAT CIRCULAR ENDCAP ANALYSIS:

Case 1: Plate Outside Diameter = 6.00", Plate Free Diameter = 5.00", Plate thickness = .625", Simply Supported Edge Restraint

Plate cross section with applied pressure, B.C.'s (ecirc1.bmp):

Under Pressure numerical stress results:



1. Maximum Radial Stress = 19,800 psi, Location = plate center
2. Maximum Tangential Stress = 19,800 psi, Location = plate center

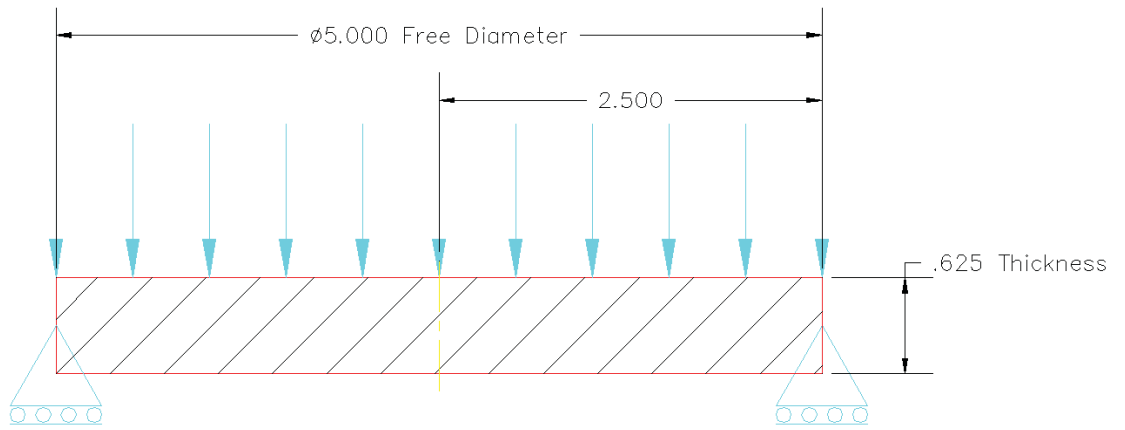
Finite Element Method graphical stress results:

FEA model (element mesh, pressure loading, B.C.'s from ecirc1mod.bmp):

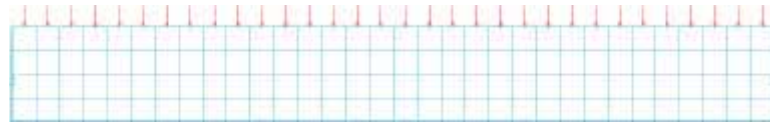
Radial stress contour (ecirc1rad.bmp):

Tangential stress contour (ecirc1tan.bmp):

1. Maximum Radial Stress = 19,904 psi, Location = plate center
2. Maximum Tangential Stress = 19,871 psi, Location = plate center



FLAT CIRCULAR ENDCAP ANALYSIS 1 – DWG
(ecirc1.bmp)



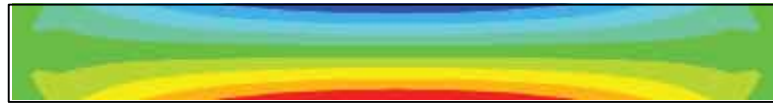
FLAT CIRCULAR ENDCAP ANALYSIS 1 – MOD
(ecirc1mod.bmp)

ANSYS 5.3
AUG 1 1997
12:33:03
ELEMENTS
TYPE NUM
U
PRES

ZV =1
DIST=1.375
XF =1.25
YF =.3125

ANSYS 5.3
AUG 1 1997
12:35:07
NODAL SOLUTION
STEP=1
SUB =1
TIME=1
SX (AVG)
RSYS=0
DMX =.01186
SMN =-19904
SMNB=-19918
SMX =19904
SMXB=19918

| | |
|-------------|--------|
| Blue | -19904 |
| Dark Blue | -15481 |
| Light Blue | -11058 |
| Green | -6635 |
| Light Green | -2212 |
| Yellow | 2212 |
| Orange | 6635 |
| Red | 11058 |
| Dark Red | 15481 |
| Red | 19904 |



FLAT CIRCULAR ENDCAP ANALYSIS 1 – RAD
(ecirc1mod.bmp)



FLAT CIRCULAR ENDCAP ANALYSIS 1 – TAN
(ecirc1tan.bmp)

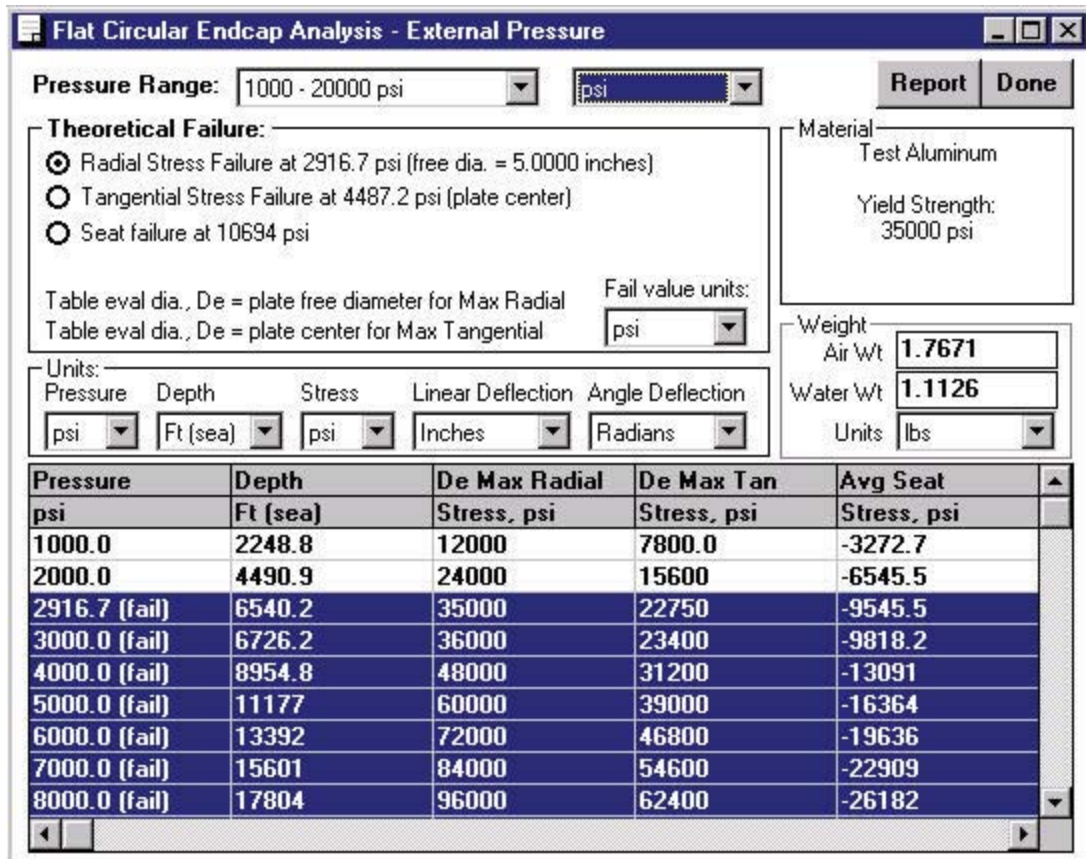
ANSYS 5.3
AUG 1 1997
12:35:50
NODAL SOLUTION
STEP=1
SUB =1
TIME=1
SZ (AVG)
RSYS=0
DMX =.01186
SMN =-19871
SMNB=-19885
SMX =19871
SMXB=19885

| | |
|-------------|--------|
| Blue | -19871 |
| Dark Blue | -15455 |
| Light Blue | -11039 |
| Green | -6624 |
| Light Green | -2208 |
| Yellow | 2208 |
| Orange | 6624 |
| Red | 11039 |
| Dark Red | 15455 |
| Red | 19871 |

Case 2: Plate Outside Diameter = 6.00", Plate Free Diameter = 5.00", Plate thickness = .625", Fixed Edge Restraint

Plate cross section with applied pressure and B.C.'s (ecirc2.bmp):

Under Pressure numerical stress results:



1. Maximum Radial Stress = 12,000 psi, Location = plate free diameter
2. Maximum Tangential Stress = 7800 psi, Location = plate center

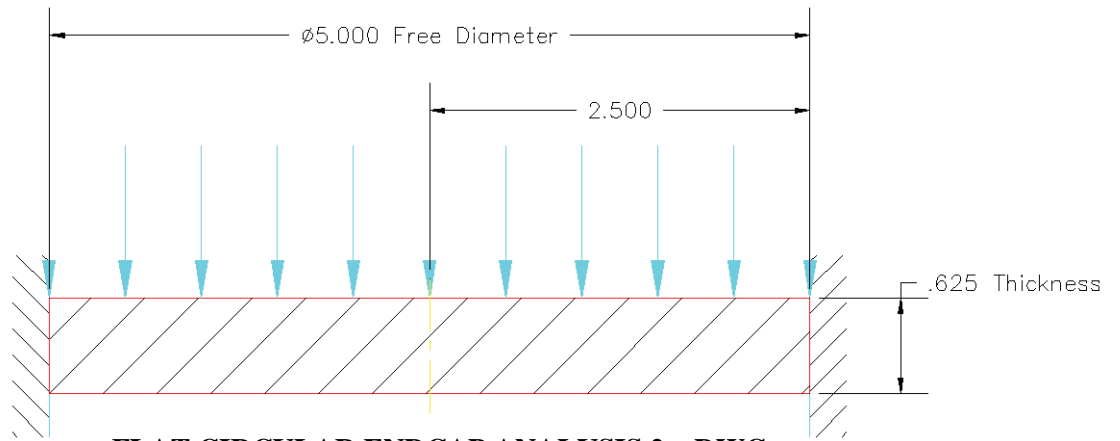
Finite Element Method graphical stress results:

FEA model (element mesh, pressure loading, B.C.'s from ecirc2mod.bmp):

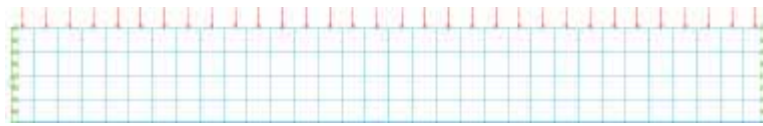
Radial stress contour (ecirc2rad.bmp):

Tangential stress contour (ecirc2tan.bmp):

1. Maximum Radial Stress = 11,612 psi, Location = plate free diameter
2. Maximum Tangential Stress = 8192 psi, Location = plate center



FLAT CIRCULAR ENDCAP ANALYSIS 2 – DWG
(ecirc2.bmp)



FLAT CIRCULAR ENDCAP ANALYSIS 2 – MOD
(ecirc2.bmp)

```
ANSYS 5.3
AUG 1 1997
12:38:12
ELEMENTS
TYPE NUM
U
CP
PRES
ZV =1
DIST=1.375
XF =1.25
YF =.3125
```

```
ANSYS 5.3
AUG 1 1997
12:39:20
NODAL SOLUTION
STEP=1
SUB =1
TIME=1
SX (AVG)
RSYS=0
DMX =.003678
SMN =-11612
SMNB=-16898
SMX =11612
SMXB=16898
```



FLAT CIRCULAR ENDCAP ANALYSIS 2 – RAD
(ecirc2.bmp)

| | |
|--------------|--------|
| Blue | -11612 |
| Light Blue | -9031 |
| Light Green | -6451 |
| Green | -3871 |
| Yellow-Green | -1290 |
| Yellow | 1290 |
| Orange | 3871 |
| Red-Orange | 6451 |
| Red | 9031 |
| Dark Red | 11612 |



FLAT CIRCULAR ENDCAP ANALYSIS 2 – TAN
(ecirc2.bmp)

```
ANSYS 5.3
AUG 1 1997
12:40:09
NODAL SOLUTION
STEP=1
SUB =1
TIME=1
SZ (AVG)
RSYS=0
DMX =.003678
SMN =-8192
SMNB=-14018
SMX =8192
SMXB=14018
```

| | |
|--------------|----------|
| Blue | -8192 |
| Light Blue | -6372 |
| Light Green | -4551 |
| Green | -2731 |
| Yellow-Green | -910.278 |
| Yellow | 910.278 |
| Orange | 2731 |
| Red-Orange | 4551 |
| Red | 6372 |
| Dark Red | 8192 |

